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A STRUCTURAL WEIGHT ESTIMATION PROGRAM (SWEEP) FOR AIRCRAFT. VOLUME III - AIRLOADS ESTIMATION MODULE. APPENDIX A: MODULE FLOW CHARTS AND FORTRAN LISTS. APPENDIX B: SAMPLE OUTPUT

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Rockwell International Corporation

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20. ABSTRACT (Continue on reverse side if necessary and identify by block number) Three computer programs were written with the objective of predicting the structural weight of aircraft through analytical methods. The first program, the structural weight estimation program (SWEEP), is a completely integrated program including routines for airloads, loads spectra, skin tem- peratures, material properties, flutter stiffness requirements, fatigue life, structural sizing, and for weight estimation of each of the major aircraft structural components. The program produces first-order weight estimates		

and indicates trends when parameters are varied. Fighters, bombers, and cargo aircraft can be analyzed by the program. The program operates within 100,000 octal units on the Control Data Corporation 6600 computer. Two stand-alone programs operating within 100,000 octal units were also developed to provide optional data sources for SWEEP. These include (1) the flexible airloads program to assess the effects of flexibility on lifting surface airloads, and (2) the flutter optimization program to optimize the stiffness distribution required for lifting surface flutter prevention.

The final report is composed of 11 volumes. This volume (Volume III) contains the methodology, program description, and user's information for the airload module of SWEEP.

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JAMES H. HALL, Colonel, USAF
Deputy for Development Planning

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APPENDIX A

MODULE FLOW CHARTS AND FORTRAN LISTS

FLOW CHART USAGE

The automatically generated computer program flow charts (AUTOFLOW) presented in this document include a table of contents, flow charts, and FORTRAN lists of all routines in the module. The 80-column card lists are sequenced and grouped by routine.

Because the AUTOFLOW system used is IBM-oriented, the functions of the BUFFERIN and BUFFEROUT statements are not recognized, but these statements appear in proper order in note boxes. Also, the PROGRAM name does not appear on the main program, and library routines READMS and WRITMS are listed as undefined external references.

CROSS-REFERENCE LIST

The AUTOFLOW table of contents which precedes the flow charts and FORTRAN lists serves to cross reference the latter two. This table lists the following from left to right:

- The card identification from columns 73 through 80 of this card, or card sequence number. When sequence number is used in place of card identification, it is enclosed in parentheses.
- The page and box number where this card is displayed in a flow chart.
- The FORTRAN statement number from columns 1 through 5 of this card.
- The card identification(s) or sequence number(s) of the card(s) referring to this card (repeated as required).
- The pages and box numbers where the cards referring to this card are displayed in a flow chart (repeated as required).

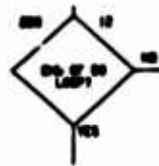
FLOW CHARTS

The flow charts produced by AUTOFLOW use USASI conventional symbols. Since the flow charts are mechanically drawn from the program source deck, there are no omissions or vague generalizations about the processing within the boxes.

Every box on each page is uniquely numbered and may be referred to from elsewhere in the program. The source of a reference to a box will be indicated by showing the page and box number. If the number is followed by an asterisk, there are multiple references to this point, and the others may be found by using the cross-reference list.



The most-often-used symbol is the decision box. Like all boxes, its box number is above and to the right of the box. Its FORTRAN statement number is above and to the left of the box. The decision choices for the paths are printed.



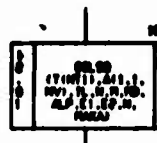
The unconditional transfer connector has its page number destination printed above or to the left of the box number destination within the connector. If there is a FORTRAN statement number at the destination, it is printed below the connector.



The exit box example shows a connector from page 9, box 15.



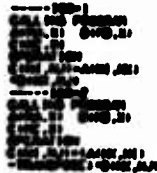
The subroutine call box includes the calling sequence. The page and box numbers of the flow chart of the called subroutine are shown on the left-hand side of the box. The page number is above the box number.



The note box encloses comments of a functional nature,



as differentiated from the 21 column comments, which are left justified without a box, that show the comment cards included in the FORTRAN deck.



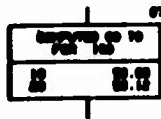
The process box is used to enclose FORTRAN arithmetic statements.



Input and output are shown as communicating with a device. The list used follows, if appropriate:



The computed $G\theta$ $T\theta$ becomes a branch table showing the page and box number of each of the ordered branches.



The column connectors and initial connectors are the only boxes without external box numbers. The function of the initial connector is always clear,

but the label given is the symbol in the next FORTRAN card, which is often blank.



The column connector identifies the page and box number to which it connects.



MODULE TABLE OF CONTENTS AND REFERENCES

PORTMAN HUBBLE AIRLADS OVERLAY

Chart Title - Introductory Comments

CHART TITLE - PROCEDURES

[illegible]

CARD NO	PAGE/FORM	NAME	REFERENCE	SOURCE	SEQUENCE NO.	AND PAGE/FORM
(000000)	0 04 00		(000100)	0 02		
(000001)	0 05 01		(000201)	0 03	(000210)	7 01
(000010)	0 07 02					
(000012)	0 09 03		(000200)	0 05		
(000014)	7 01 04		(000170)	0 13		
(000019)	7 02 05		(000170)	0 13	(000170)	0 13
(000020)	7 03 06		(000170)	0 13	(000170)	0 13
(000025)	7 04 07		(000100)	0 14	(000100)	0 15
			(000211)	7 02	(000100)	0 16
					(000100)	0 17
					(000100)	0 18
(000030)	7 05 08					
(000031)	7 07 09		(000030)	7 05		
(000032)	0 01 00		(000100)	0 10	(000034)	7 03
(000037)	0 02 01		(000030)	7 06	(000030)	7 07
(000040)	0 04 02					
(000041)	0 05 03					
(000045)	0 06 04					
(000046)	0 08 05		(000030)	0 03		
(000046)	0 07 07					
(000047)	0 09 08		(000040)	0 04		
(000048)	0 09 09		(000045)	0 05		
(000049)	0 10 00		(000051)	0 11		
(000050)	0 11 00		(000051)	0 09		
(000050)	0 12 01					
(000053)	0 13 02					
(000055)	0 01 03		(000052)	0 12		
(000055)	0 02 04		(000042)	0 06	(000047)	0 07
(000059)	0 04 05		(000051)	0 05	(000044)	0 08
(000059)	0 05 06		(000050)	0 03	(000050)	0 10
(000059)	0 06 07				(000059)	0 13
(000060)	0 07 08		(000061)	0 07	(000061)	0 08
(000060)	0 07 09		(000060)	0 03	(000060)	0 04
(000070)	0 10 11					
(000070)	0 12 12		(000060)	0 06		
(000070)	0 10 14		(000071)	0 13	(000071)	0 13
			(000071)	0 13	(000071)	0 13
(000077)	0 16 15		(000071)	0 13		
(000077)	0 17 16		(000071)	0 13		
(000081)	0 10 17		(000071)	0 13	(000071)	0 13
(000083)	0 10 18		(000071)	0 13	(000071)	0 13
(000083)	0 20 19		(000071)	0 13	(000071)	0 13
(000087)	0 21 20		(000071)	0 13	(000071)	0 13
(000090)	10 01 21		(000071)	0 13	(000071)	0 13
(000090)	10 02 22		(000070)	0 15	(000070)	0 16
			(000090)	0 20	(000090)	0 21
(000314)	10 03 403		(000312)	10 05		
(000310)	10 07 0		(000314)	10 05	(000314)	10 05
			(000314)	10 05	(000314)	10 05
			(000314)	10 05	(000314)	10 05
(000300)	10 00 7		(000314)	10 05	(000314)	10 05
			(000314)	10 05	(000314)	10 05
(000304)	10 00 8		(000314)	10 05		
(000300)	11 01 9		(000314)	10 05		
(000321)	11 02 10		(000310)	10 07	(000302)	10 08
(000325)	11 04 11		(000324)	11 03	(000305)	10 09
(000342)	11 05 12		(000324)	11 03		
(000344)	11 05 13		(000342)	11 05		
(000340)	11 07 14		(000312)	10 05	(000342)	11 05
(000307)	11 10 015		(000303)	11 09		
(000301)	11 11 123		(000171)	0 12	(000303)	11 09
(000305)	11 14 034		(000304)	11 13		
(000300)	11 15 030		(000304)	11 13		
(000302)	12 01 0033		(000300)	11 15		
(000304)	12 02 0034		(000300)	11 15		
(000300)	12 05		(000304)	12 09		
(000307)	12 06 1133		(000300)	12 05		
(000304)	12 08 133		(000300)	12 05		
(000307)	12 10 400					
(000300)	12 11 400		(000305)	12 09		
(000400)	12 12 407					
(000402)	12 13 410		(000300)	12 11		
(000403)	12 14 412					
(000405)	12 15 400		(000402)	12 13		
(000401)	12 17 0003					
(000404)	12 20 0007		(000402)	12 25		
(000407)	12 22 0000		(000407)	12 16	(000402)	12 25

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE ATYDS(H,ROW,PH,AN)

(000400)	10 01 ATYDS	(000500)	10 01-2	(000500)	10 10-2
(000500)	10 04 20				
(000507)	10 05 40	(000501)	10 03		
(000508)	10 06 40				
(000510)	10 07 00	(000507)	10 05		
(000513)	10 08 00				
(000516)	10 09 00	(000512)	10 07		
(000519)	10 10 00				
(000523)	10 01 70	(000518)	10 09		
(000524)	10 02 70				
(000529)	10 03 00	(000523)	10 01		
(000533)	10 04 100	(000505)	10 04	(000517)	10 08
(000540)	10 05 110	(000511)	10 06	(000520)	10 10
(000546)	10 06 120	(000530)	10 09	(000520)	10 03

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE SHLOS

(000500)	10 01 SHLOS	(000500)	10 02-2		
(000500)	10 04 3	(000500)	10 04		
(000571)	10 05 4	(000570)	10 05		
(000574)	10 07 5	(000570)	10 05		
(000576)	10 08 6	(000575)	10 06		
(000577)	10 09 7				
(000579)	10 10 8	(000570)	10 06		
(000581)	10 01 9	(000570)	10 06		
(000583)	10 02 10	(000570)	10 06		
(000585)	10 03 11	(000570)	10 06	(000500)	10 10
(000586)	10 04 12			(000500)	10 01
(000588)	10 05 13				
(000593)	10 06 00				
(000594)	10 06 01	(000507)	10 07		
(000597)	10 10 14	(000500)	10 08		
(000599)	10 11 15	(000500)	10 08		
(000600)	10 12 16				
(000602)	10 13 17	(000500)	10 11		
(000603)	10 14 18	(000601)	10 12		
(000605)	10 15 19	(000605)	10 15	(000605)	10 15
		(000613)	10 16		
(000606)	10 17 20	(000606)	10 16	(000605)	10 16
(000611)	10 18 21	(000606)	10 16		
(000614)	20 01 22	(000607)	10 16	(000610)	10 17
(000623)	20 02 23				
(000627)	20 03 24	(000602)	20 05		
(000629)	20 10 25				
(000631)	20 11 26	(000629)	20 06		
(000632)	20 12 26	(000630)	20 10		
(000640)	20 13 26				
(000642)	20 17 26	(000630)	20 15		
(000643)	20 18 27	(000611)	20 16		
(000644)	20 19 28				
(000645)	20 20 28	(000643)	20 18		
(000651)	20 23 29				
(000651)	20 23	(000651)	20 24		
(000659)	20 27 30	(000653)	20 26	(000653)	20 26
		(000657)	20 26	(000650)	20 26
(000660)	21 01 27	(000653)	20 26	(000653)	20 26
(000667)	21 02 28				
(000668)	21 03 28	(000660)	21 01		
(000669)	21 04 29	(000660)	21 02		
(000670)	21 05 31				
(000674)	21 07 32	(000661)	21 05		
(000675)	21 08 33	(000663)	21 06		
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(000677)	21 12 2				
(000678)	21 13	(000671)	21 13		
(000679)	21 14 40	(000660)	21 10		

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(000000)	00.03	35		
(000007)	00.04	51	(000000)	00.00
(000000)	00.05	36	(000001)	00.00
(000000)	00.07	36	(000001)	00.00
(000000)	00.08	37	(000000)	00.07
(000000)	00.10	0001		
(000701)	00.12	47		
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(000000) 21.17

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - FUNCTION CODING (H, S, I, V, N, M)

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(000700)	00.05	100	(000700)	00.00		
(000700)	00.05	100	(000701)	00.00		
(000700)	00.07	100				
(000700)	00.08		(000700)	00.10		
(000700)	00.10	170				
(000700)	00.11	170	(000700)	00.00		
(000700)	00.10		(000707)	00.10		
(000707)	00.10	100	(000700)	00.10	(000700)	00.00
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(000701)	00.17	210				
(000700)	00.01	000	(000701)	00.00	(000700)	00.10
(000700)	00.02	000	(000700)	00.10		
(000707)	00.03	000				
(000700)	00.04	000	(000700)	00.00		
(000707)	00.05	000	(000703)	00.17	(000700)	00.00
(000700)	00.05	000				
(000707)	00.07	000	(000701)	00.00		
(000700)	00.10	000				
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(000707)	00.15	310	(000700)	00.10		
(000700)	00.16	300	(000707)	00.10		
(000001)	07.01	330	(000707)	00.10		
(000007)	07.03	300	(000707)	00.10	(000000)	00.10
(000010)	07.05	400				
(000012)	07.05	410	(000000)	07.00		
(000010)	07.07	000	(000707)	00.00	(000703)	00.00
					(000700)	00.01
					(000011)	07.00

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE PATHS

(000000)	00.01	FATH	(000700)	00.17-0
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(000007)	00.12	6	(000000)	00.10
(000000)	00.13	7	(000000)	00.10
(000000)	00.14	8	(000000)	00.10
(000000)	00.15	9	(000000)	00.10
(000000)	00.16	10	(000000)	00.10
(000000)	00.17		(000000)	00.10
(000000)	00.18	11		
(000000)	00.21		(000000)	00.00
(000000)	00.02	12		
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(000000)	01.01	14	(000000)	00.00
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(000000) 00.11 (000000) 00.12 (000000) 00.13 (000000) 00.14

CARD NO PAGE/NO

REFERENCES SOURCE SEQUENCE NO AND PAGE/NO

100000	31 03 10	100001	30 25						
100000	31 04 17	100001	30 25	100001	30 25	100003	30 25	100005	31 01
100070	31 05	100070	31 05					100007	31 02
100073	31 07 10								
100070	31 08 10	100070	31 08						
100070	31 09 20	100070	31 07						
100002	31 12	100070	31 21						
100000	31 15 21								
100001	31 16 22	100007	31 14						
100003	31 17 23	100000	31 15						
100004	31 18 24	100004	31 23	100005	31 24				
100000	31 21 25								
100001	31 22 27	100000	31 26	100000	31 25				
100004	31 23 28	100003	31 17						
100005	31 24 29								
100000	31 25 30								
100007	31 26 31	100003	31 20						
100000	31 28 32	100000	31 27	100000	31 27				
100011	31 30 33	100010	31 28						
100013	32 01 34	100010	31 29						
100015	32 02 35	100010	31 29						
100017	32 03 36	100000	31 27						
100019	32 04 37	100000	31 27						
100021	32 05 38								
100023	32 07 39	100020	32 05						
100025	32 08 40	100000	31 27						
100027	32 10 41								
100029	32 11 42	100000	31 27						
100031	32 13 43								
100032	32 14 44	100010	31 29	100012	31 30	100014	32 01	100016	32 02
		100010	32 05	100020	32 05	100024	32 07	100026	32 08
		100030	32 12					100018	32 03
100033	32 15	100030	32 17					100020	32 10
100035	32 16 45								
100040	32 21	100041	32 01						
100041	32 22 46								
100052	33 07	100000	33 10						
100000	33 08 47								
100000	33 10 48	100004	33 08						
100007	33 11 49								
100000	33 12 50	100000	33 10						
100000	33 15 51								
100007	33 16 52	100004	33 14						
100000	33 17 53	100000	33 15						
100070	33 20 54								
100070	33 23 55								
100073	33 24	100074	33 26						
100074	33 25 56								
100077	33 26 57								
100070	33 30 58	100070	33 28						
100070	33 31	100003	33 33						
100003	33 32 59								
100000	33 35 60								
100000	34 01 61	100000	33 35						
100000	34 02 62	100071	33 36						
100000	34 05	100000	34 00						
100000	34 07 63								
100012	35 02 64								
100014	35 04 65								
100015	35 05 66	100011	35 01	100013	35 03				
100023	35 06 67	100071	33 37						
100000	35 10 68								
100020	35 11 69	100001	35 02						
100030	35 12 70								
100035	35 14 71	100020	35 11						
100040	35 16 72	100034	35 13						
100041	35 17	100002	34 27						
100042	35 18 73								
100044	35 19 74	100041	35 17						
100045	35 21 75								
100046	35 23 101	100045	35 20						
100050	35 25 76	100046	35 22						
100054	35 26 102								

CARD NO	PAGE/END	NAME	REFERENCES	SOURCE	SEQUENCE NO.	AND PAGE/END
(001007)	35 32	105	(001003)	35 30		
(001008)	35 33	76				
(001009)	35 31	75	(001007)	35 32		
(001010)	35 32	77	(001008)	35 33		
(001011)	35 33	78	(001009)	35 31		
(001012)	35 37	9000	(001017)	35 36	(001000)	35 33
(001013)	35 38	115				
(001014)	35 39		(001015)	35 18		
(001015)	35 12	116				
(001016)	35 12		(001017)	35 13		
(001017)	35 14	88				
(001018)	35 15	112	(001017)	35 14		
(001019)	37 01	113	(001017)	35 14		
(001020)	37 02	114	(001000)	35 15		
(001021)	37 03		(001000)	37 07		
(001022)	37 05	110				
(001023)	37 05		(001000)	37 08		
(001024)	37 07	117				
(001025)	37 08	120				
(001026)	37 08		(001002)	37 10		
(001027)	37 12	122				
(001028)	37 12		(001004)	37 13		

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - FUNCTION FORMS (R,V,H1,Z1,H1,H2,H3)

(001125)	40 01	100				
(001126)	40 02	110				
(001127)	40 03	120	(001123)	40 01		
(001128)	40 04	125	(001131)	40 05		
(001129)	40 05	130				
(001130)	40 07	135	(001129)	40 04		
(001131)	40 08	140				
(001132)	40 09	145	(001132)	40 07		
(001133)	40 10	150	(001131)	40 06		
(001134)	41 01	170	(001135)	40 08		
(001135)	41 02	200	(001136)	40 09	(001137)	40 10
(001136)	41 03		(001143)	41 05		
(001137)	41 04	200				

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE FORMS

(001150)	44 01	FUNCTION	(001157)	11 10-2	(001200)	11 14-2
(001151)	44 02	0				
(001152)	44 04		(001155)	44 05		
(001153)	44 05	10				
(001154)	44 06	700	(001152)	44 01		
(001155)	44 07	701				
(001156)	44 07		(001150)	44 02		
(001157)	44 08	710	(001155)	44 04		
(001158)	44 10	815	(001151)	44 05		
(001159)	44 12	821				
(001160)	44 13	825	(001155)	44 11		
(001161)	44 14	824				
(001162)	44 15	826	(001155)	44 13		
(001163)	44 16	828				
(001164)	45 01	829	(001155)	45 03	(001201)	45 05
(001165)	45 02	827	(001155)	44 11		
(001166)	45 03	827	(001155)	44 15		
(001167)	45 04	828	(001155)	44 16		
(001168)	45 05	818	(001155)	45 03		
(001169)	45 05	811				
(001170)	45 07	812	(001157)	44 12	(001154)	44 14
			(001200)	45 04	(001157)	44 16
					(001202)	45 01
					(001155)	45 02

CARD NO PAGE/BOX NAME REFERENCES (SOURCE SEQUENCE NO. AND PAGE/BOX)

(001200) 45 00 020 (001200) 45 07
 (001211) 45 01 021 (001200) 45 07
 (001213) 45 02 022 (001200) 45 30
 (001230) 45 00 3012 (001101) 44 00
 (001247) 45 11 3015 (001235) 46 07

CHART TITLE - NEW-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE PARLODS

(001200) 45 01 PARLODS (000351) 11 00-5 (000370) 12 01-2
 (001204) 45 01 100
 (001205) 45 02 101
 (001206) 45 03 102
 (001200) 45 05 (001200) 45 00
 (001200) 45 00 103
 (001213) 45 07 104 (001204) 45 01
 (001214) 45 00 (001215) 45 00
 (001215) 45 00 105
 (001234) 45 10 102 (001205) 45 02 (001230) 53 11
 (001230) 45 11 (001201) 57 10
 (001244) 45 15 1 (001242) 45 14 (001242) 45 14 (001242) 45 14 (001242) 45 14
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 (001247) 45 25 (001213) 53 02
 (001240) 45 26 001 (001247) 45 25
 (001201) 50 01 021 (001240) 45 00
 (001202) 50 02 (001253) 50 04
 (001253) 50 03 005
 (001207) 50 00 002 (001247) 45 25
 (001200) 50 07 002 (001257) 50 00
 (001200) 50 00 (001201) 50 10
 (001201) 50 00 000
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 (001200) 50 14 (001200) 50 10
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			(001559)	08.01								
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(001637)	08.17	130	(001637)	08.18								
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(001631)	08.02	121	(001638)	08.01								
(001635)	08.03	120	(001634)	08.19								
(001638)	08.04	151	(001635)	08.03								
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(001630)	08.12	191										
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(001650)	08.16	240	(001645)	08.16								
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CHART TITLE - NEW-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE SPAN

(001716) 04.01 SPAN (000000) 10 03-11

CARD NO	PAGE/BOX	NAME	REFERENCES	(SOURCE SEQUENCE NO AND PAGE/BOX)
(001706)	04 02 2		(001731)	04 07
(001707)	04 03		(001708)	04 05
(001708)	04 04 3			
(001730)	04 07 4			
(001732)	04 08 5		(001709)	04 06
(001730)	04 10 1			
(001730)	04 11 6		(001724)	04 08
(001730)	04 13 7			
(001700)	04 14 15		(001730)	04 10
(001704)	04 16 47		(001021)	05 20
(001705)	04 17 10		(001703)	04 15
(001770)	04 19		(001772)	04 21
(001770)	04 20 10			
(001777)	04 23 10		(001704)	04 16
(001700)	04 25 40		(001023)	05 01
(001701)	04 26 20			
(001702)	04 27 21		(001774)	04 24
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(001743)	05 02 9		(001740)	04 13
(001747)	05 04		(001749)	05 06
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(001704)	05 09 22		(001701)	04 20
(001707)	05 10 23		(001701)	04 26
(001708)	05 11 24		(001703)	04 27
(001708)	05 12 25		(001705)	05 09
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(001000)	05 22 32		(001700)	05 10
(001000)	05 23 33		(001001)	05 19
(001000)	05 24		(001020)	05 35
(001012)	05 26 34			
(001013)	05 29 35			
(001015)	05 30 36		(001012)	05 20
(001016)	05 31 37		(001011)	05 27
(001020)	05 34 38		(001014)	05 29
(001022)	05 37 39			
(001023)	05 01 40		(001021)	05 30
(001024)	05 02 41		(001700)	04 25
(001025)	05 04 5501			
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(001029)	05 07 50		(001027)	05 05
(001030)	05 13 5502		(001025)	05 03
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(001044)	05 18 53		(001042)	05 16
(001040)	05 24 5504		(001040)	05 14
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(001051)	05 26 5505			
(001053)	05 28 55			
(001054)	05 29 56		(001052)	05 27
(001050)	05 35 57		(001053)	05 28
(001050)	05 38 5506		(001050)	05 25

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE USPAN

(001005)	00 01 0	(001005)	00 07
(001000)	00 02 USPAN	(000271)	9 11-X (000004) 31.10-X
(001003)	00 03 1		
(001003)	00 03	(001003)	00 04
(001005)	00 06 2	(000200)	70 13
(001005)	00 07	(000202)	01 07
(001007)	00 08 3	(000204)	01 08
		(001005)	70 03

CARD ID PAGE/BOOK NAME

REFERENCES (SOURCE SEQUENCE NO. AND PAGE/BOOK)

(002067)	75.13	04																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																																							
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07/03/74	TABLE OF CONTENTS AND REFERENCES		AUTOFLOW CHART SET - SHEEP	
CARD NO	PAGE/BOOK	NAME	REFERENCES (SOURCE SEQUENCE NO. AND PAGE/BOOK)	
(000000)	01.04	125	(000277)	00.17
(000270)	01.06	127	(000270)	00.10
(000271)	01.07	128	(000270)	00.10
(000273)	01.08	129	(000270)	01.00
(000276)	01.09	130	(000275)	00.15
(000278)	01.10	95.17	(000270)	00.10
(000277)	01.11	131	(000277)	01.05
(000278)	01.12	132	(002330)	02.00
(000278)	01.13	133	(002343)	02.14
(000280)	01.14	134	(000270)	01.10
(000210)	01.16		(000277)	01.11
(000213)	01.18	135	(000213)	01.01
(000216)	01.23	136		
(000217)	01.24	137		
(000220)	02.02		(002290)	02.00
(000220)	02.03	141		
(000230)	02.05	142	(000216)	01.23
(000231)	02.07	143	(000270)	01.12
(000232)	02.10		(002342)	02.13
(000242)	02.11	145		
(000244)	02.15	146	(000270)	01.13
(000250)	02.18		(000270)	02.21
(000250)	02.19	148		
(000250)	02.22	95.18	(000270)	01.00
(000257)	02.23	149	(000270)	01.13
			(000215)	01.02

CHART TITLE - NON-PROCEDURAL STATEMENTS

CHART TITLE - INTRODUCTORY COMMENTS

CHART TITLE - SUBROUTINE SHEET

(000205)	05.01	4000ET	(000374)	12.02-11		
(000400)	05.04	4005	(000200)	05.03		
(000417)	05.05	4006	(000400)	05.04	(000441)	05.22
(000420)	05.06	4007	(000417)	05.05	(000441)	05.22
(000200)	05.01	4001	(000200)	05.03		
(000400)	05.04	4002	(000400)	05.04		
(000412)	05.06	4012				
(000412)	05.06		(000412)	05.07		
(000410)	05.06	4003	(000417)	05.05		
(000423)	05.11	4013				
(000423)	05.11		(000423)	05.12		
(000430)	05.14	4004	(000420)	05.06		
(000434)	05.16	4014				
(000434)	05.16		(000434)	05.17		
(000430)	05.19	4010	(000404)	05.03	(000415)	05.00
(000430)	05.20	4011			(000420)	05.13
(000430)	05.20		(000430)	05.21		
(000443)	05.23	4020	(000420)	05.06	(000441)	05.22
(000454)	07.01	400	(000452)	05.25	(000441)	05.22
(000450)	07.02	401	(000454)	07.01		
(000400)	07.04		(000505)	07.17		
(000500)	07.16	403				
(000511)	07.18		(000513)	07.21		
(000513)	07.20	4030				
(000517)	07.22	402	(000452)	05.25	(000454)	07.01
(000531)	08.01	402	(000520)	07.24		
(000537)	08.03		(000531)	08.13		
(000530)	08.05	207				
(000543)	08.06	208	(000530)	08.04		
(000544)	08.07	214				
(000545)	08.08	211				
(000540)	08.09	210	(000530)	08.05	(000543)	08.06
(000540)	08.10	212	(000545)	08.08		
(000540)	08.11	213	(000547)	08.09	(000542)	08.01
(000501)	08.12	404				
(000505)	08.15		(000557)	08.17		
(000557)	08.16	4040				
(000501)	08.18	405	(000520)	07.24		
(000541)	08.01	209	(000530)	08.05		

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(002563)	09.02	407	(002561)	09.10		
(002569)	09.04		(002577)	09.12		
(002571)	09.06	215				
(002572)	09.07	221				
(002573)	09.08	220	(002570)	09.09	(002571)	09.06
(002575)	09.09	222	(002572)	09.07		
(002576)	09.10	223	(002574)	09.08		
(002577)	09.11	409				
(002581)	09.14		(002582)	09.16		
(002582)	09.15	4000				
(002583)	09.17	410	(002581)	09.16		
(002585)	09.18	412	(002586)	09.17		
(002594)	09.20		(002582)	09.04		
(002595)	09.22	216				
(002597)	09.23	231				
(002598)	09.24	230	(002596)	09.21	(002598)	09.22
(002600)	09.25	232	(002597)	09.23		
(002601)	09.26	233	(002599)	09.24		
(002602)	09.28	414				
(002605)	09.28		(002607)	09.28		
(002607)	09.27	4140				
(002610)	09.29	420	(002605)	09.17		
(002613)	09.31	9501				
(002620)	09.34	452	(002610)	09.13		
(002657)	09.20	456	(002610)	09.13		
(002690)	01.01	462	(002657)	09.20		
(002694)	01.15	464				
(002695)	01.16	465	(002694)	01.15		
(002701)	01.17	465	(002697)	09.20	(002694)	01.15
(002703)	02.01	472	(002701)	01.17		
(002706)	02.15	9562	(002702)	09.10	(002701)	01.17
(002735)	02.17	3001				
(002736)	02.18	3002				
(002737)	02.19	3003				
(002740)	02.20	3004	(002734)	02.18	(002735)	02.17
(002743)	02.22	3005	(002737)	02.19	(002736)	02.18
					(002737)	02.19

CHART TITLE - NON-PROCEDURAL STATEMENTS

LOCATION		DIAGNOSTIC
CARD 10	PAGE/BOOK	
(000030)	2. 02	UNRECOGNIZED SYNTAX
(000005)	2. 10	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000005)	2. 10	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000007)	2. 20	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000008)	2. 21	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000009)	2. 30	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000009)	2. 31	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000009)	2. 32	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000401)	3. 10	UNRECOGNIZED SYNTAX
(000535)	30. 01	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000537)	30. 02	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000538)	30. 03	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000538)	30. 04	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000540)	30. 05	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(001005)	37. 14	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(001245)	40. 10	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(001303)	40. 10	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(001305)	40. 10	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(001705)	01. 10	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(000700)	05. 01	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000701)	05. 02	UNDEFINED - 'READMS' EXTERNAL REFERENCE
(000740)	05. 20	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE
(000741)	05. 21	UNDEFINED - 'WRITMS' EXTERNAL REFERENCE

MODULE FLOW CHARTS

CHART TITLE - INTRODUCTORY COMMENTS

PROGRAM BLCHIL IS LINE 4 OF SHEEP

DETERMINES BASIC AIRLOAD CONDITIONS TO BE COMPUTED ON AND WING
BENDING MOMENT FATIGUE SPECTRA
PROVIDES LOGIC AND CONTROL FOR THE AIRLOAD SUBROUTINES

[illegible]

COURT TITLE - PROCEEDINGS

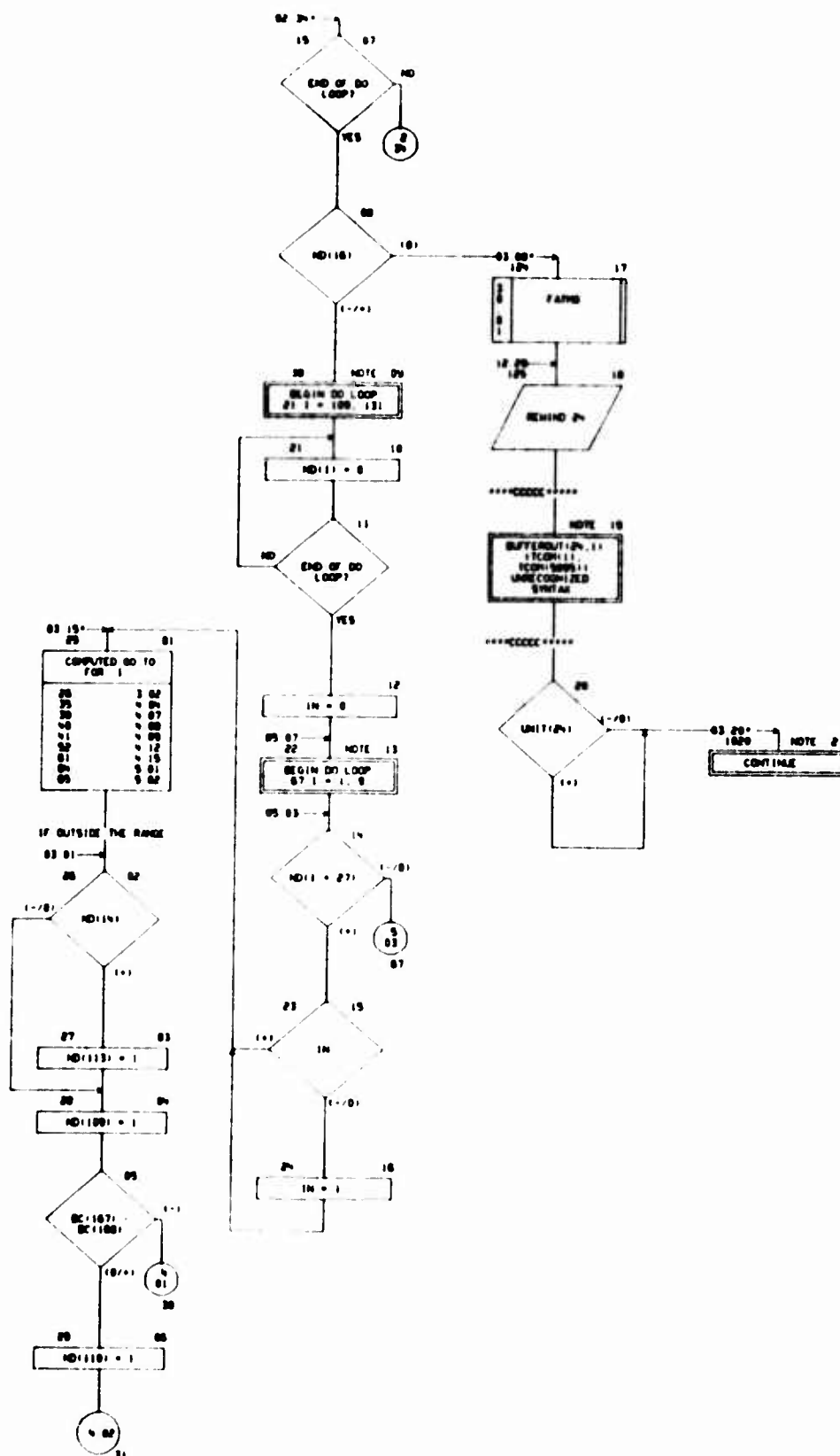


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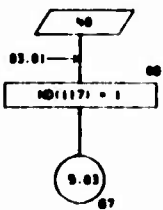
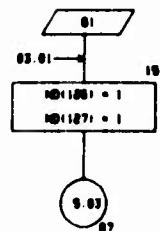
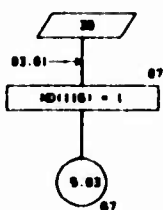
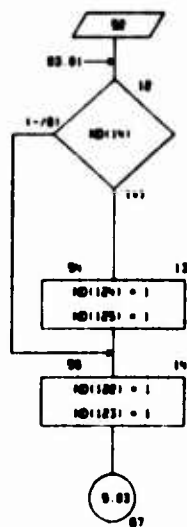
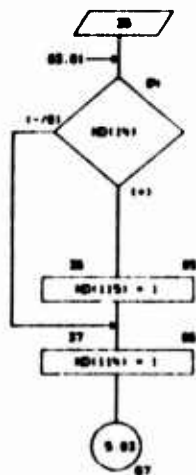
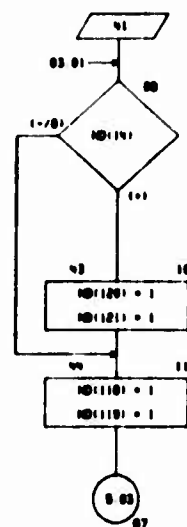
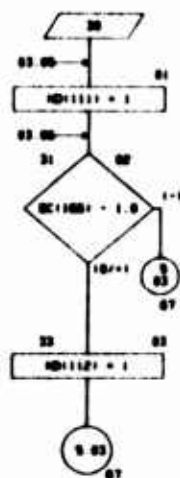


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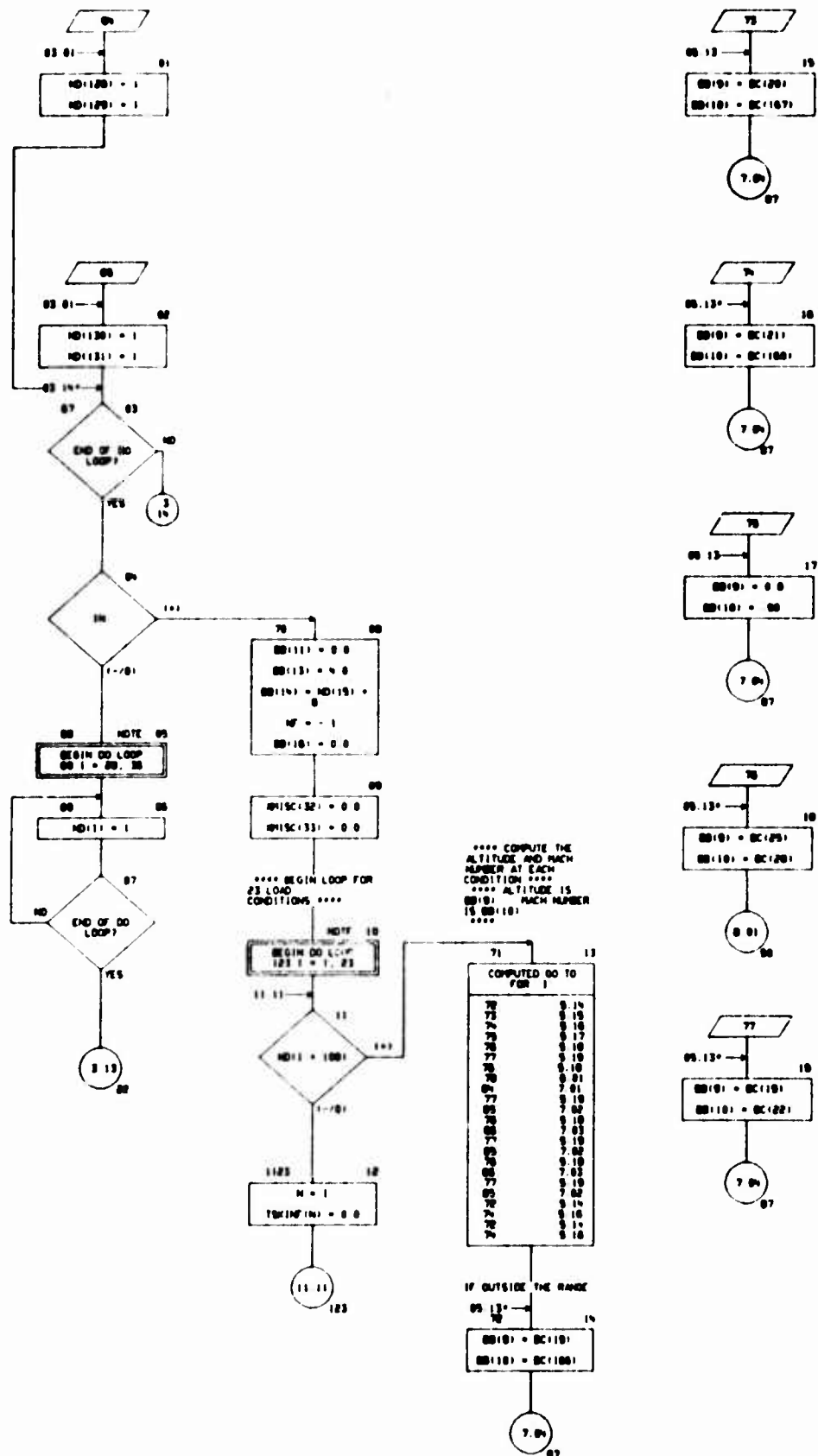


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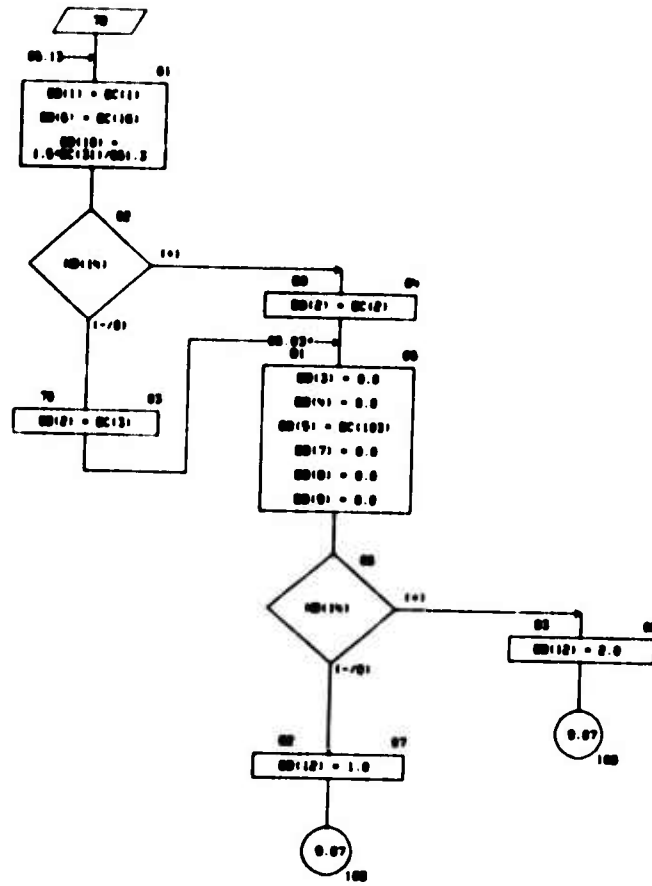


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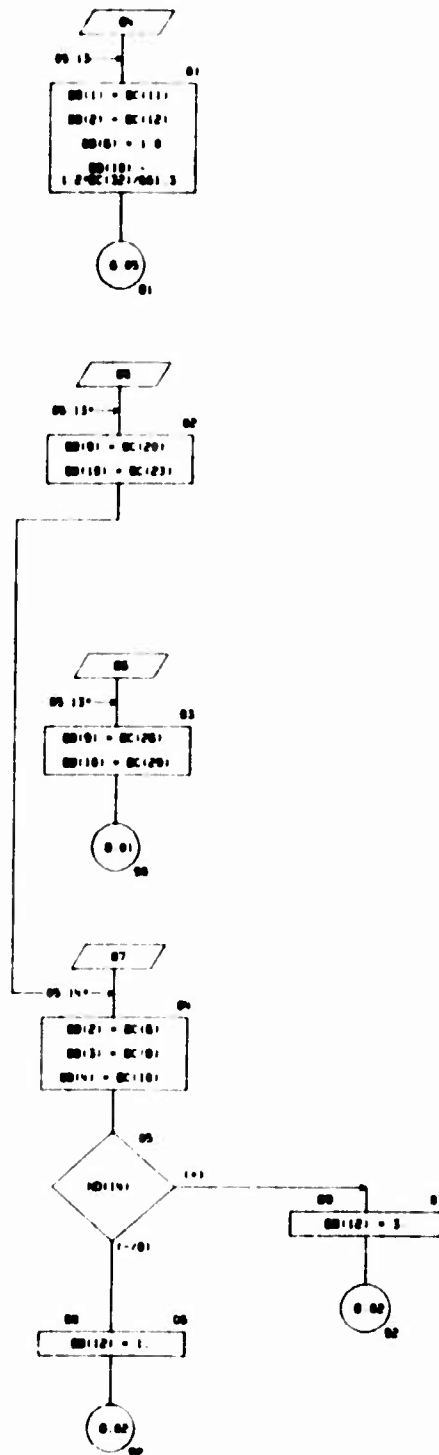


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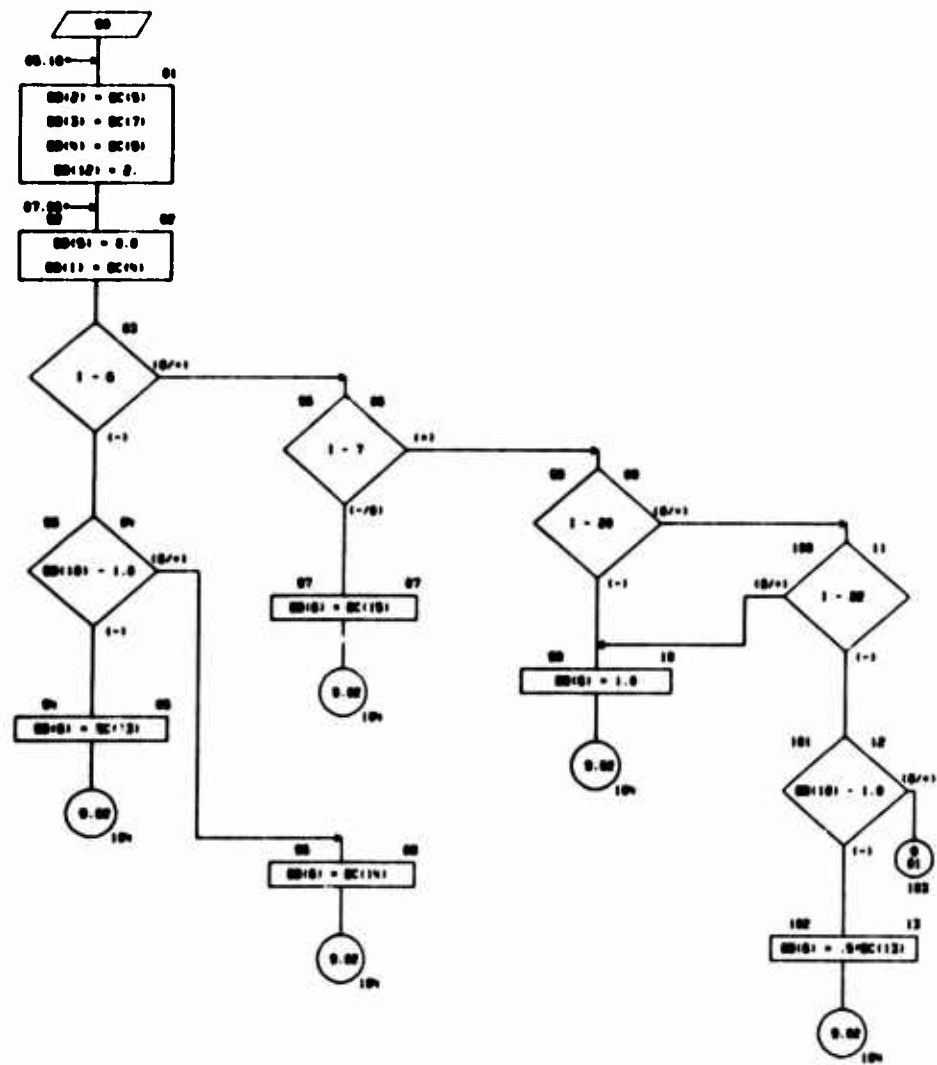


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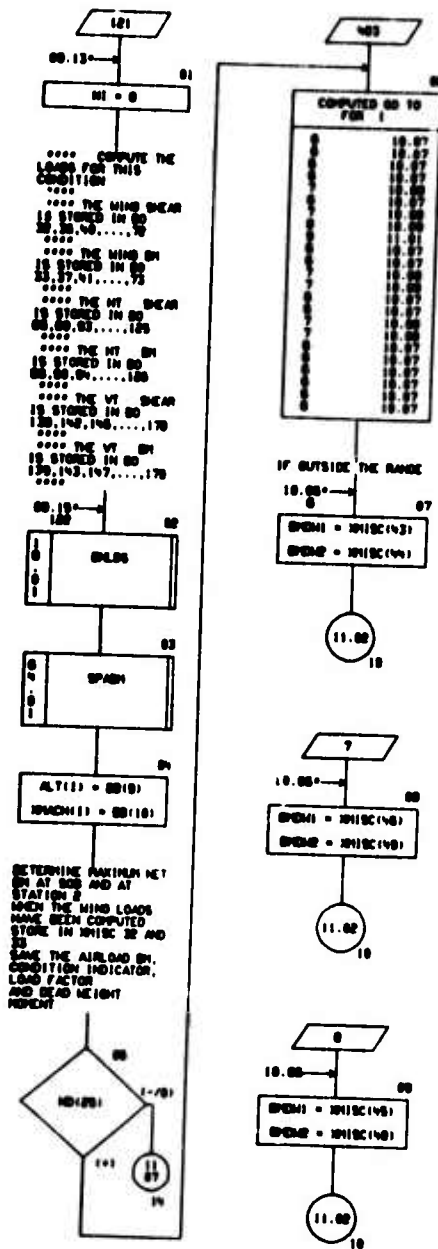


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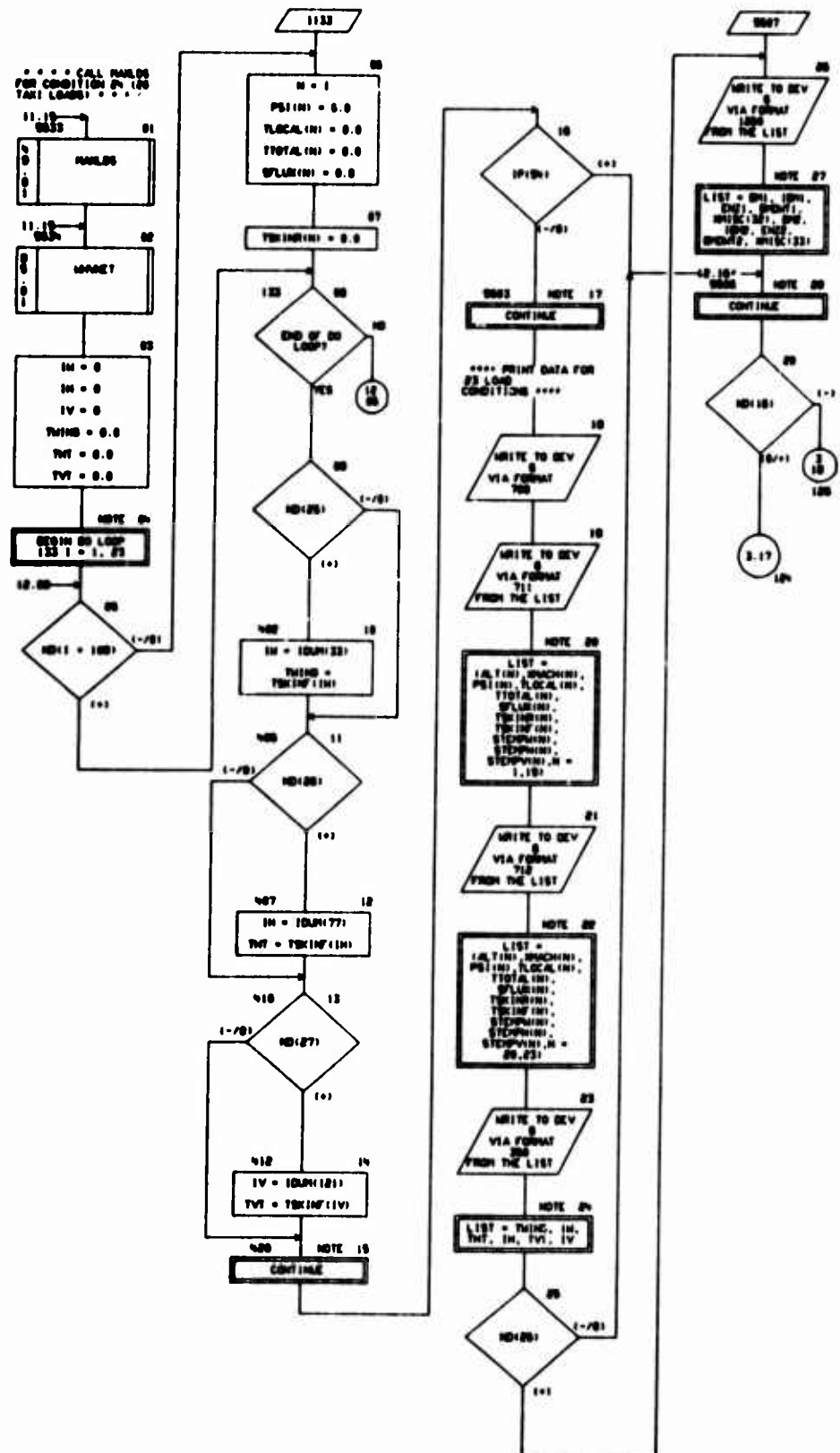
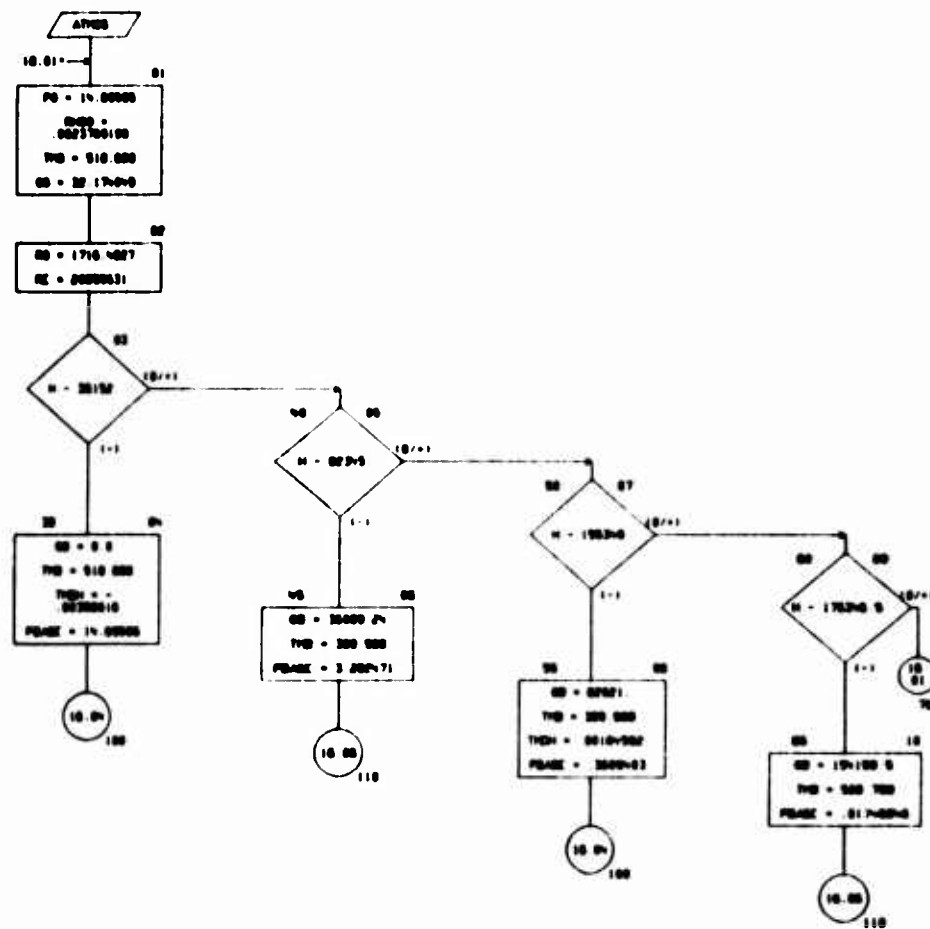


CHART TITLE - INTRODUCTORY COMMENTS

SUBMIT THE ATYDS

CHART TITLE - SUBROUTINE ATNDS(H,RENH,PH,OH)



```

graph TD
    10([10]) --> 11{11 - N > 0}
    11 -- 1 --> 12[12 - M = 1/2 * (N + 1)  
P = 1/2 * (N - 1)]
    11 -- 2 --> 13[13 - M = 1/2 * N  
P = 1/2 * N]
    12 --> 14[14 - AV = (M + P) / 2]
    13 --> 14
    14 --> 15([15 - AV])
    15 --> 16([16])
  
```

CHART TITLE - INTRODUCTORY COMMENTS

.....
SUBROUTINE END
.....

CHART TITLE - SUBROUTINE DLDS

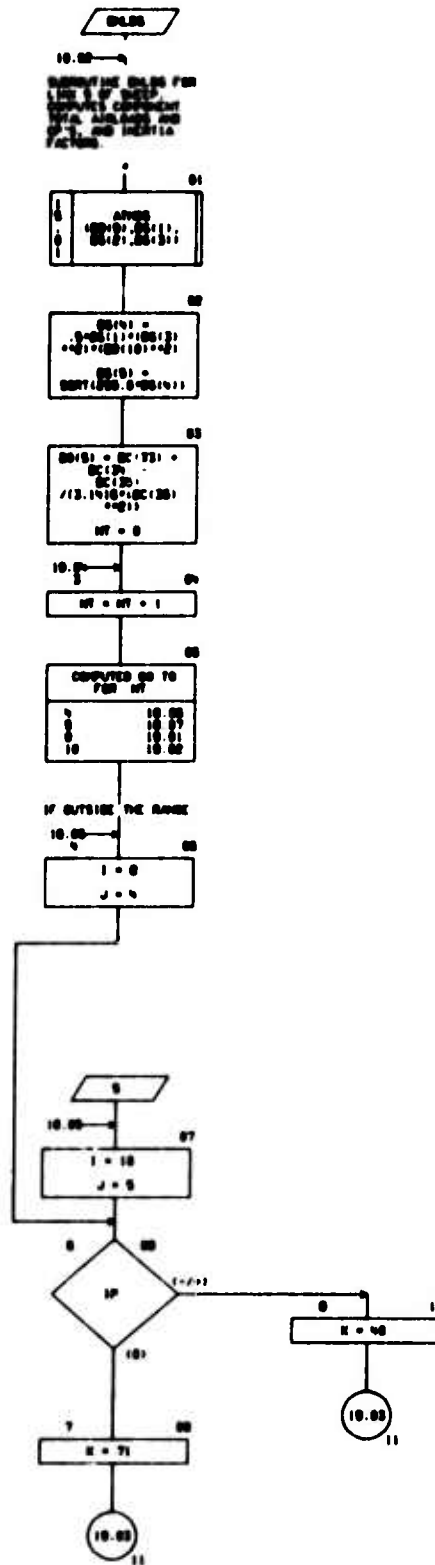


CHART TITLE - SUBROUTINE SWDS

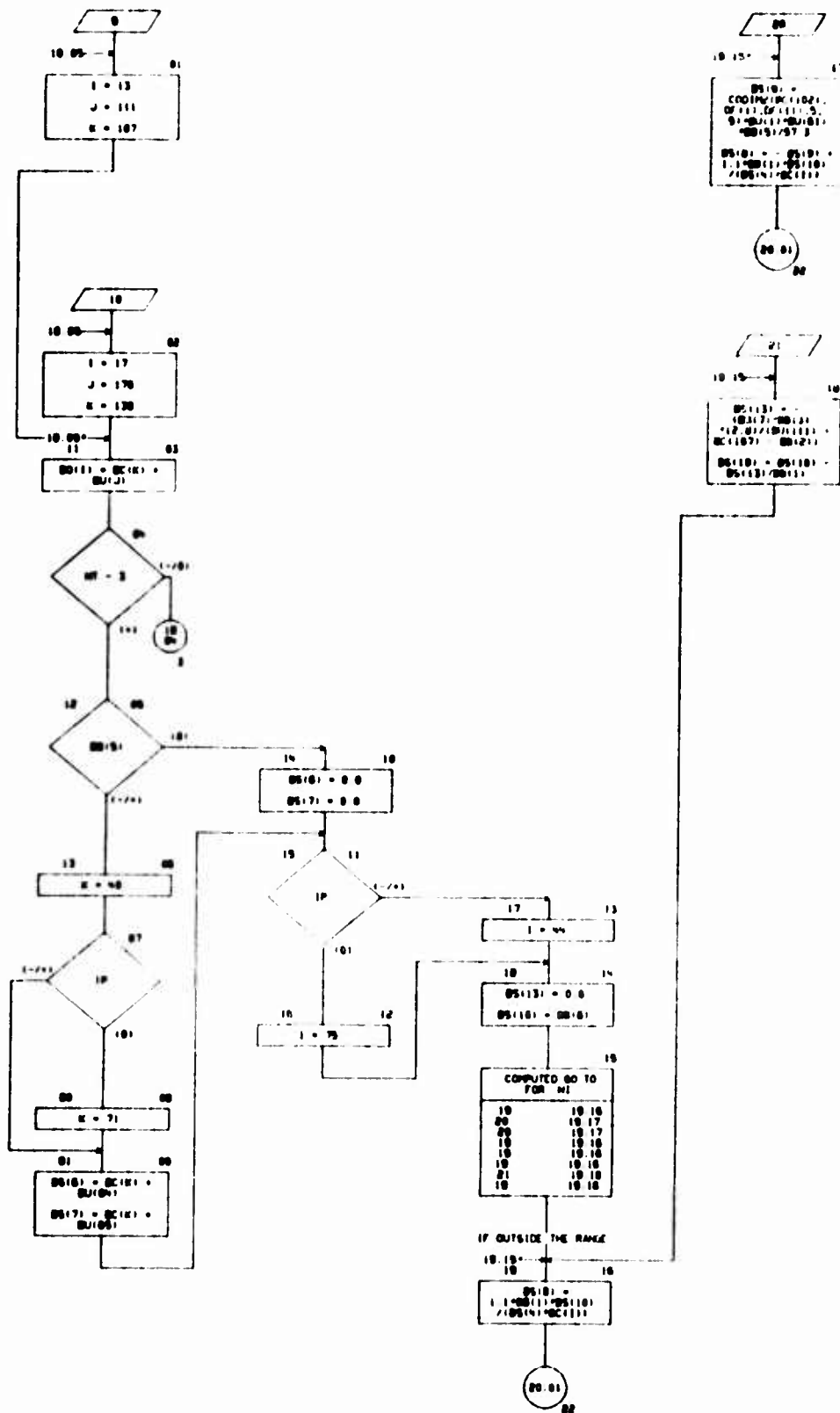
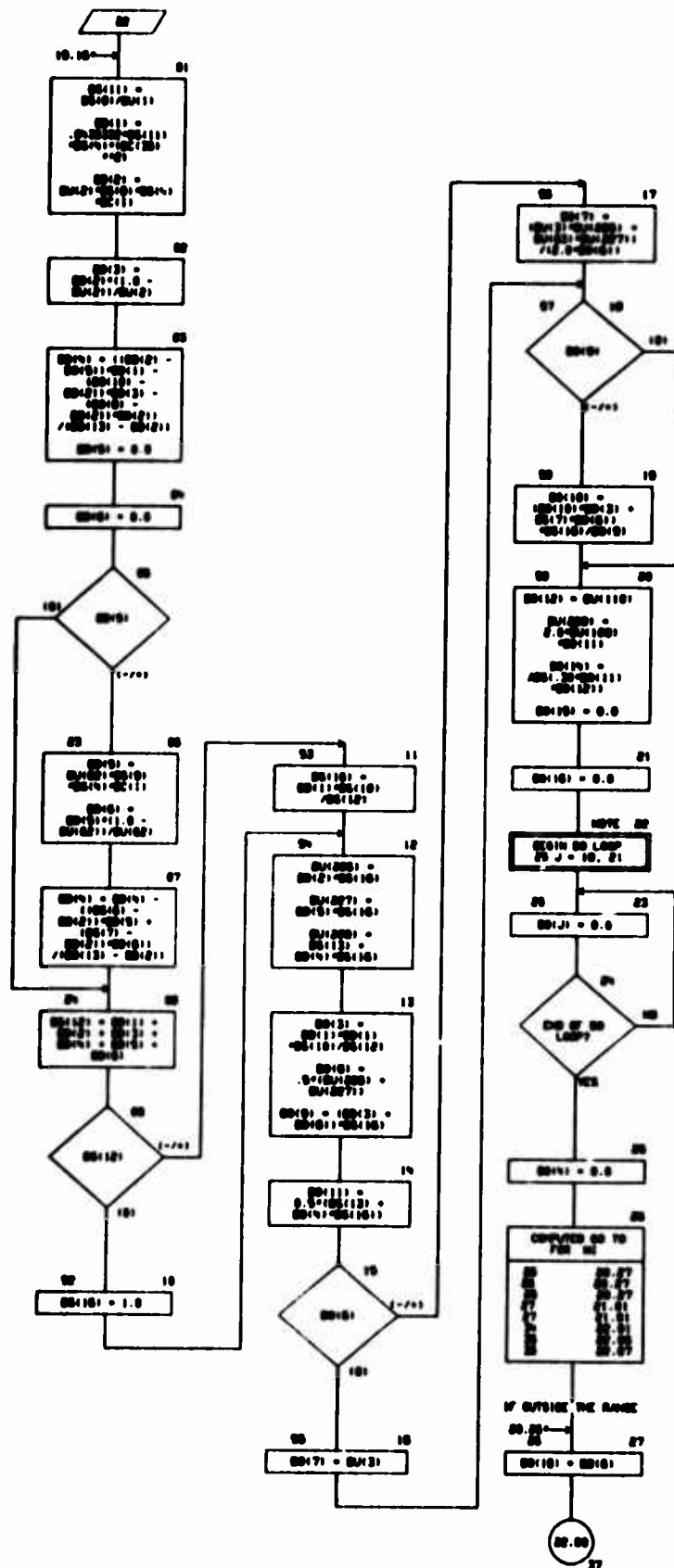


CHART TITLE - BUREAU OF BUREAU



```

graph TD
    Start([Start]) --> B7[07]
    B7 --> B1[01]
    B1 --> D1{1P}
    D1 --> B2[02]
    D1 --> B3[03]
    B2 --> B4[04]
    B4 --> B5[05]
    B5 --> B6[06]
    B6 --> B7
    B3 --> B8[08]
    B8 --> B9[09]
    B9 --> B10[10]
    B10 --> B11[11]
    B11 --> B12[12]
    B12 --> B13[13]
    B13 --> B14[14]
    B14 --> B15[15]
    B15 --> B16[16]
    B16 --> B17[17]
    B17 --> B18[18]
    B18 --> B19[19]
    B19 --> B20[20]
    B20 --> B21[21]
    B21 --> B22[22]
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    B23 --> B24[24]
    B24 --> B25[25]
    B25 --> B26[26]
    B26 --> B27[27]
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    B28 --> B29[29]
    B29 --> B30[30]
    B30 --> B31[31]
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    B32 --> B33[33]
    B33 --> B34[34]
    B34 --> B35[35]
    B35 --> B36[36]
    B36 --> B37[37]
    B37 --> B38[38]
    B38 --> B39[39]
    B39 --> B40[40]
    B40 --> B41[41]
    B41 --> B42[42]
    B42 --> B43[43]
    B43 --> B44[44]
    B44 --> B45[45]
    B45 --> B46[46]
    B46 --> B47[47]
    B47 --> B48[48]
    B48 --> B49[49]
    B49 --> B50[50]
    B50 --> B51[51]
    B51 --> B52[52]
    B52 --> B53[53]
    B53 --> B54[54]
    B54 --> B55[55]
    B55 --> B56[56]
    B56 --> B57[57]
    B57 --> B58[58]
    B58 --> B59[59]
    B59 --> B60[60]
    B60 --> B61[61]
    B61 --> B62[62]
    B62 --> B63[63]
    B63 --> B64[64]
    B64 --> B65[65]
    B65 --> B66[66]
    B66 --> B67[67]
    B67 --> B68[68]
    B68 --> B69[69]
    B69 --> B70[70]
    B70 --> B71[71]
    B71 --> B72[72]
    B72 --> B73[73]
    B73 --> B74[74]
    B74 --> B75[75]
    B75 --> B76[76]
    B76 --> B77[77]
    B77 --> B78[78]
    B78 --> B79[79]
    B79 --> B80[80]
    B80 --> B81[81]
    B81 --> B82[82]
    B82 --> B83[83]
    B83 --> B84[84]
    B84 --> B85[85]
    B85 --> B86[86]
    B86 --> B87[87]
    B87 --> B88[88]
    B88 --> B89[89]
    B89 --> B90[90]
    B90 --> B91[91]
    B91 --> B92[92]
    B92 --> B93[93]
    B93 --> B94[94]
    B94 --> B95[95]
    B95 --> B96[96]
    B96 --> B97[97]
    B97 --> B98[98]
    B98 --> B99[99]
    B99 --> End([End])
  
```

The flowchart illustrates the algorithm for finding the minimum number of elements in a subset S of natural numbers such that the sum of the elements equals N. The process begins with an initial value for N (07) and proceeds through a series of steps (01 to 99) involving calculations and decision points. Key steps include:

- Step 01: Initial value for N (07).
- Step 02: Decision point (1P).
- Step 03: Calculation of B1 = B0 + 1.
- Step 04: Calculation of B2 = B1 + B0.
- Step 05: Calculation of B3 = B2 + B1.
- Step 06: Calculation of B4 = B3 + B2.
- Step 07: Calculation of B5 = B4 + B3.
- Step 08: Calculation of B6 = B5 + B4.
- Step 09: Calculation of B7 = B6 + B5.
- Step 10: Calculation of B8 = B7 + B6.
- Step 11: Calculation of B9 = B8 + B7.
- Step 12: Calculation of B10 = B9 + B8.
- Step 13: Calculation of B11 = B10 + B9.
- Step 14: Calculation of B12 = B11 + B10.
- Step 15: Calculation of B13 = B12 + B11.
- Step 16: Calculation of B14 = B13 + B12.
- Step 17: Calculation of B15 = B14 + B13.
- Step 18: Calculation of B16 = B15 + B14.
- Step 19: Calculation of B17 = B16 + B15.
- Step 20: Calculation of B18 = B17 + B16.
- Step 21: Calculation of B19 = B18 + B17.
- Step 22: Calculation of B20 = B19 + B18.
- Step 23: Calculation of B21 = B20 + B19.
- Step 24: Calculation of B22 = B21 + B20.
- Step 25: Calculation of B23 = B22 + B21.
- Step 26: Calculation of B24 = B23 + B22.
- Step 27: Calculation of B25 = B24 + B23.
- Step 28: Calculation of B26 = B25 + B24.
- Step 29: Calculation of B27 = B26 + B25.
- Step 30: Calculation of B28 = B27 + B26.
- Step 31: Calculation of B29 = B28 + B27.
- Step 32: Calculation of B30 = B29 + B28.
- Step 33: Calculation of B31 = B30 + B29.
- Step 34: Calculation of B32 = B31 + B30.
- Step 35: Calculation of B33 = B32 + B31.
- Step 36: Calculation of B34 = B33 + B32.
- Step 37: Calculation of B35 = B34 + B33.
- Step 38: Calculation of B36 = B35 + B34.
- Step 39: Calculation of B37 = B36 + B35.
- Step 40: Calculation of B38 = B37 + B36.
- Step 41: Calculation of B39 = B38 + B37.
- Step 42: Calculation of B40 = B39 + B38.
- Step 43: Calculation of B41 = B40 + B39.
- Step 44: Calculation of B42 = B41 + B40.
- Step 45: Calculation of B43 = B42 + B41.
- Step 46: Calculation of B44 = B43 + B42.
- Step 47: Calculation of B45 = B44 + B43.
- Step 48: Calculation of B46 = B45 + B44.
- Step 49: Calculation of B47 = B46 + B45.
- Step 50: Calculation of B48 = B47 + B46.
- Step 51: Calculation of B49 = B48 + B47.
- Step 52: Calculation of B50 = B49 + B48.
- Step 53: Calculation of B51 = B50 + B49.
- Step 54: Calculation of B52 = B51 + B50.
- Step 55: Calculation of B53 = B52 + B51.
- Step 56: Calculation of B54 = B53 + B52.
- Step 57: Calculation of B55 = B54 + B53.
- Step 58: Calculation of B56 = B55 + B54.
- Step 59: Calculation of B57 = B56 + B55.
- Step 60: Calculation of B58 = B57 + B56.
- Step 61: Calculation of B59 = B58 + B57.
- Step 62: Calculation of B60 = B59 + B58.
- Step 63: Calculation of B61 = B60 + B59.
- Step 64: Calculation of B62 = B61 + B60.
- Step 65: Calculation of B63 = B62 + B61.
- Step 66: Calculation of B64 = B63 + B62.
- Step 67: Calculation of B65 = B64 + B63.
- Step 68: Calculation of B66 = B65 + B64.
- Step 69: Calculation of B67 = B66 + B65.
- Step 70: Calculation of B68 = B67 + B66.
- Step 71: Calculation of B69 = B68 + B67.
- Step 72: Calculation of B70 = B69 + B68.
- Step 73: Calculation of B71 = B70 + B69.
- Step 74: Calculation of B72 = B71 + B70.
- Step 75: Calculation of B73 = B72 + B71.
- Step 76: Calculation of B74 = B73 + B72.
- Step 77: Calculation of B75 = B74 + B73.
- Step 78: Calculation of B76 = B75 + B74.
- Step 79: Calculation of B77 = B76 + B75.
- Step 80: Calculation of B78 = B77 + B76.
- Step 81: Calculation of B79 = B78 + B77.
- Step 82: Calculation of B80 = B79 + B78.
- Step 83: Calculation of B81 = B80 + B79.
- Step 84: Calculation of B82 = B81 + B80.
- Step 85: Calculation of B83 = B82 + B81.
- Step 86: Calculation of B84 = B83 + B82.
- Step 87: Calculation of B85 = B84 + B83.
- Step 88: Calculation of B86 = B85 + B84.
- Step 89: Calculation of B87 = B86 + B85.
- Step 90: Calculation of B88 = B87 + B86.
- Step 91: Calculation of B89 = B88 + B87.
- Step 92: Calculation of B90 = B89 + B88.
- Step 93: Calculation of B91 = B90 + B89.
- Step 94: Calculation of B92 = B91 + B90.
- Step 95: Calculation of B93 = B92 + B91.
- Step 96: Calculation of B94 = B93 + B92.
- Step 97: Calculation of B95 = B94 + B93.
- Step 98: Calculation of B96 = B95 + B94.
- Step 99: Calculation of B97 = B96 + B95.

CHART TITLE - SUBROUTINE. 0455



CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(4400)
COMMON /IPRINT/IO(80)
DIMENSION BB(20),BD(100),BC(100),BS(20),BU(20),BO(20),OF(140),ND
(200)
EQUIVALENCE (BF(1),TCON(1000)),(BD(1),TCON(2003)),(BC(1),TCON(2750
)),(BU(1),TCON(3153)),(BS(1),TCON(3973)),(BO(1),TCON(3341)),(BD(1),
TCON(2003)),(ND(1),TCON(4201)),(NI,ND(100)),(NT,ND(150)),(I,ND(15
1)),(J,ND(152)),(K,ND(153)),(IP,ND(157))
40  FORMAT(1H,6X,13CONDITION NO=,F7.0,6X,3000=,F6.3,6X,4441=,F7.0,
    6X,300=,F6.2,10X,2000=,B4.00=,1P(50) **/)
20  FORMAT(1H,6X,13BODY LOADS/TK,4P2H=,F6.0,4X,44P2H=,F7.0,4X,
    4000=,F6.2/)
30  FORMAT(1H,6X,13WING PANEL LOAD/TK,8P2H(1)/2=,F6.0,4X,74P2H(1)=
    F7.0,4X,700H(1)=,F6.2/)
40  FORMAT(1H,6X,13WING CARRY-OVER LOAD/TK,7P2H(1)=,F6.0,4X,
    7000(1)=,F6.2/)
41  FORMAT(1H,6X,13HORIZONTAL TAIL LOADS/TK,8P2H/2=,F6.0,4X,44P2H=,
    F6.2,4X,4000=,F6.2,4X,50000=,F6.0/)
42  FORMAT(1H,6X,13VERTICAL TAIL LOAD/TK,4P2H=,F6.0,4X,44P2H=,F6.2,
    4X,4000=,F6.2/)
43  FORMAT(1H,6X,13WINGPLANE INERTIA FACTORS/TK,3042=,F6.2,4X,3000=,
    F6.2,4X,34000=,F7.3,4X,30000=,F7.3/)
44  FORMAT(1H,6X,13COMPONENT SPANWISE FACTORS/TK,8P2H(1)A=,F6.0,4X,
    8P2H(1)F=,F6.0,4X,7P2H(1)=,F6.0)

```

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AUTOFLEX CHART SET - SHEEP AIRLADS OVERLAY

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CHART TITLE - INTRODUCTORY COMMENTS

.....
FUNCTION CODING
.....

CHART TITLE - FUNCTION CODING(X,XI,YI,N,NKI)

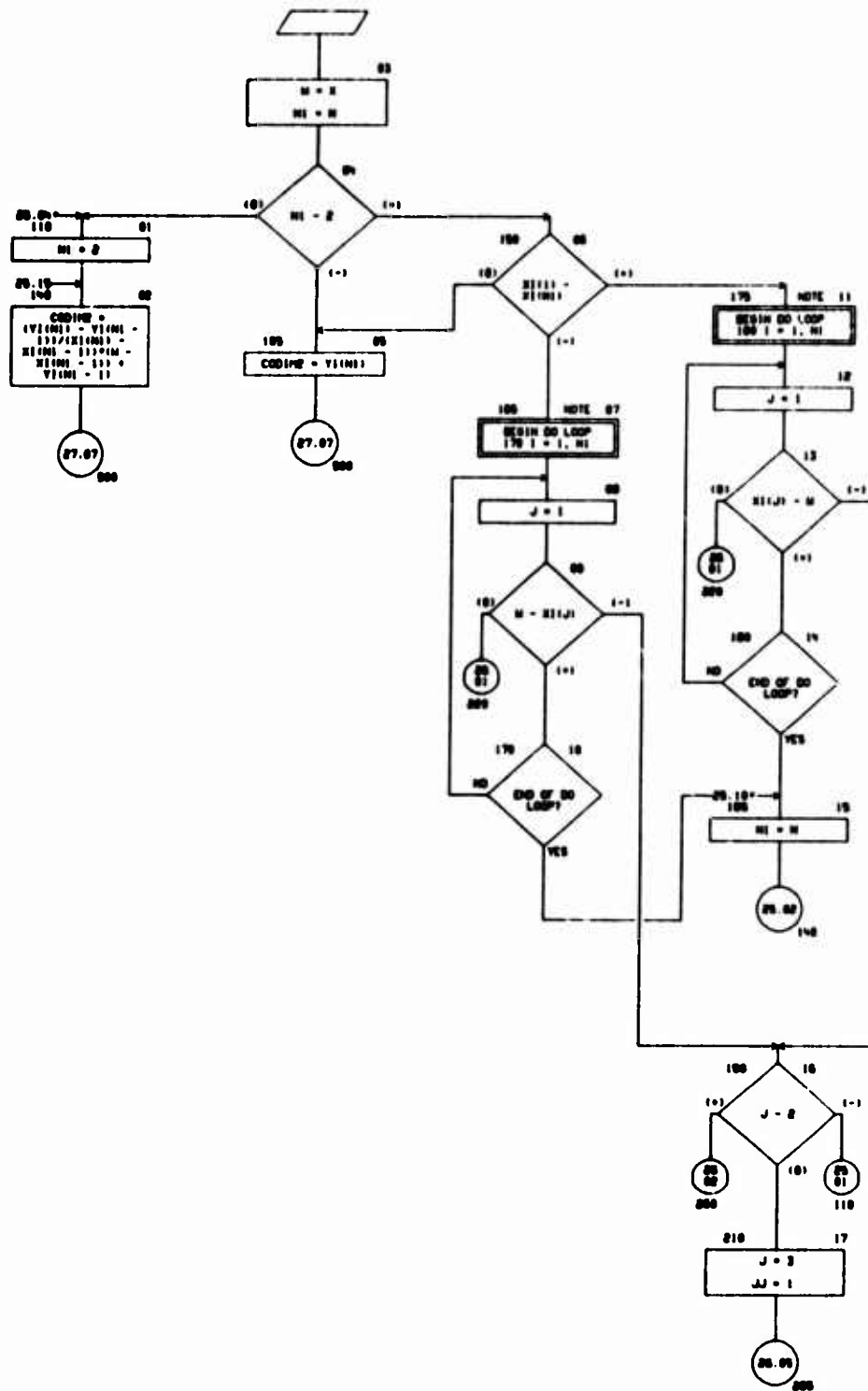
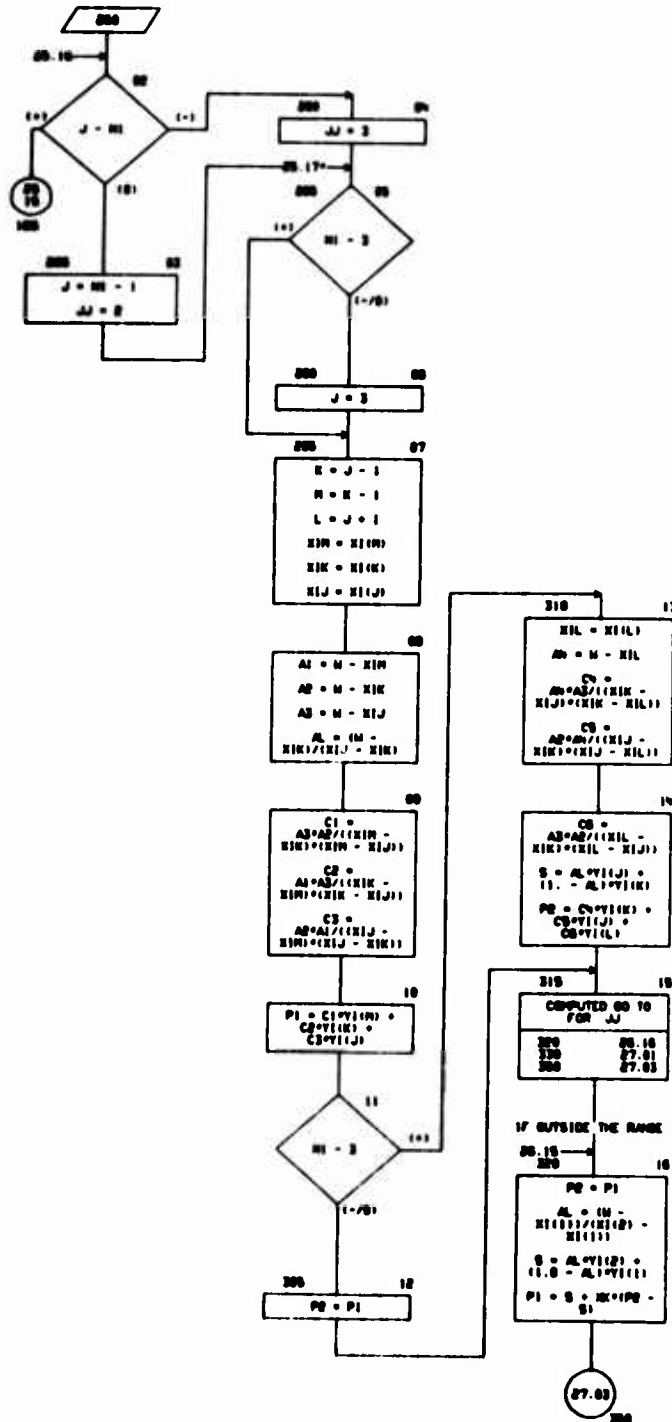
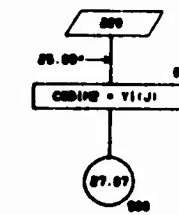
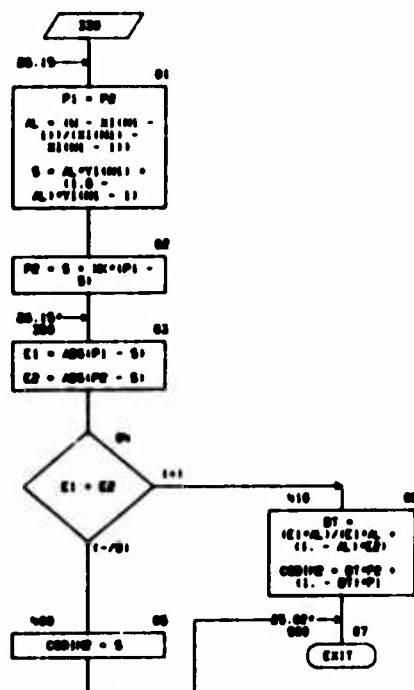


CHART TITLE - FUNCTION CODE: M2, N1, Y1, N, N2



ENTRY TITLE - FUNCTION CODES(X,XI,VI,M,EX)



07/08/74

AUTOLIN CHART SET - SHEEP AIRLINES OVERLAY

PAGE 00

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

CHART TITLE - NEW-PROCESSIONAL STATEMENTS

07:03 PM

AUTOLON CHART SET - SHEEP AIRWAYS OVERLAY

PAGE 20

CHART TITLE - INTERIOR CORE CURRENTS

.....
SUBROUTINE PATHS
.....

.....

CHART TITLE - SUBROUTINE - 1140

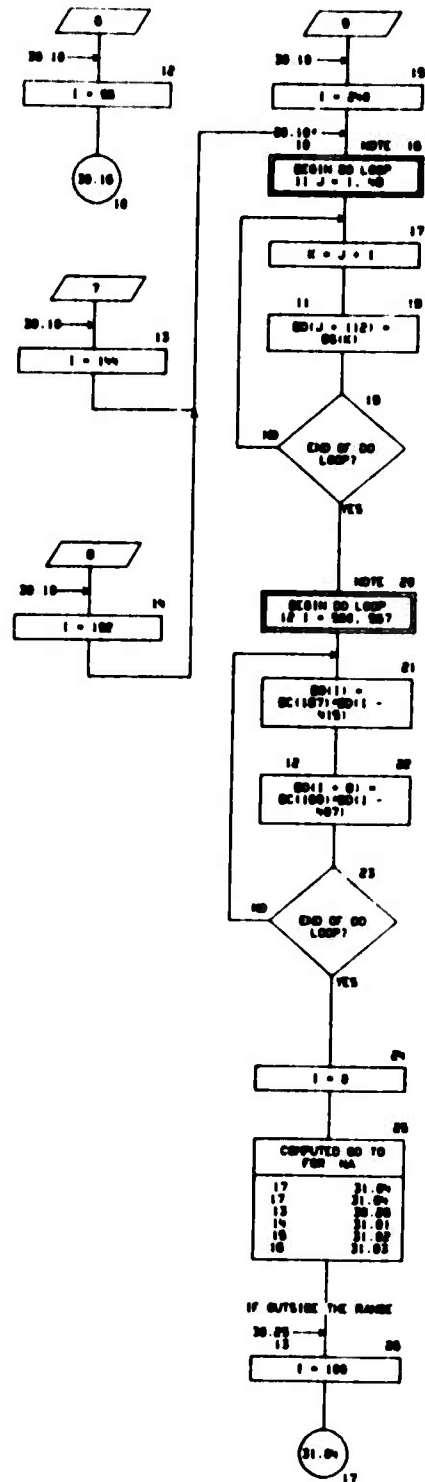
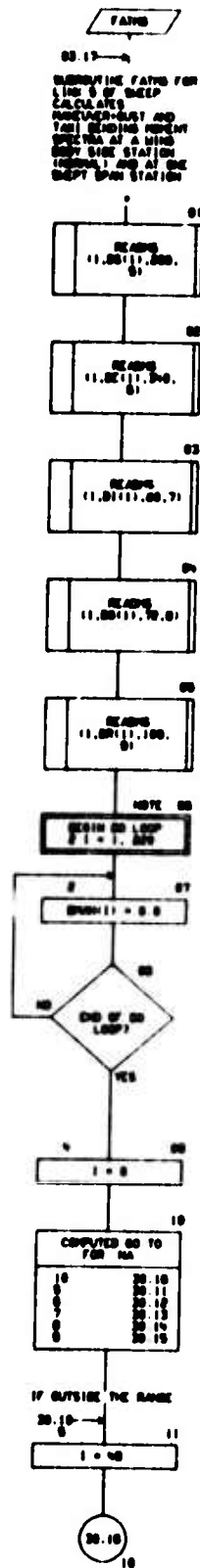
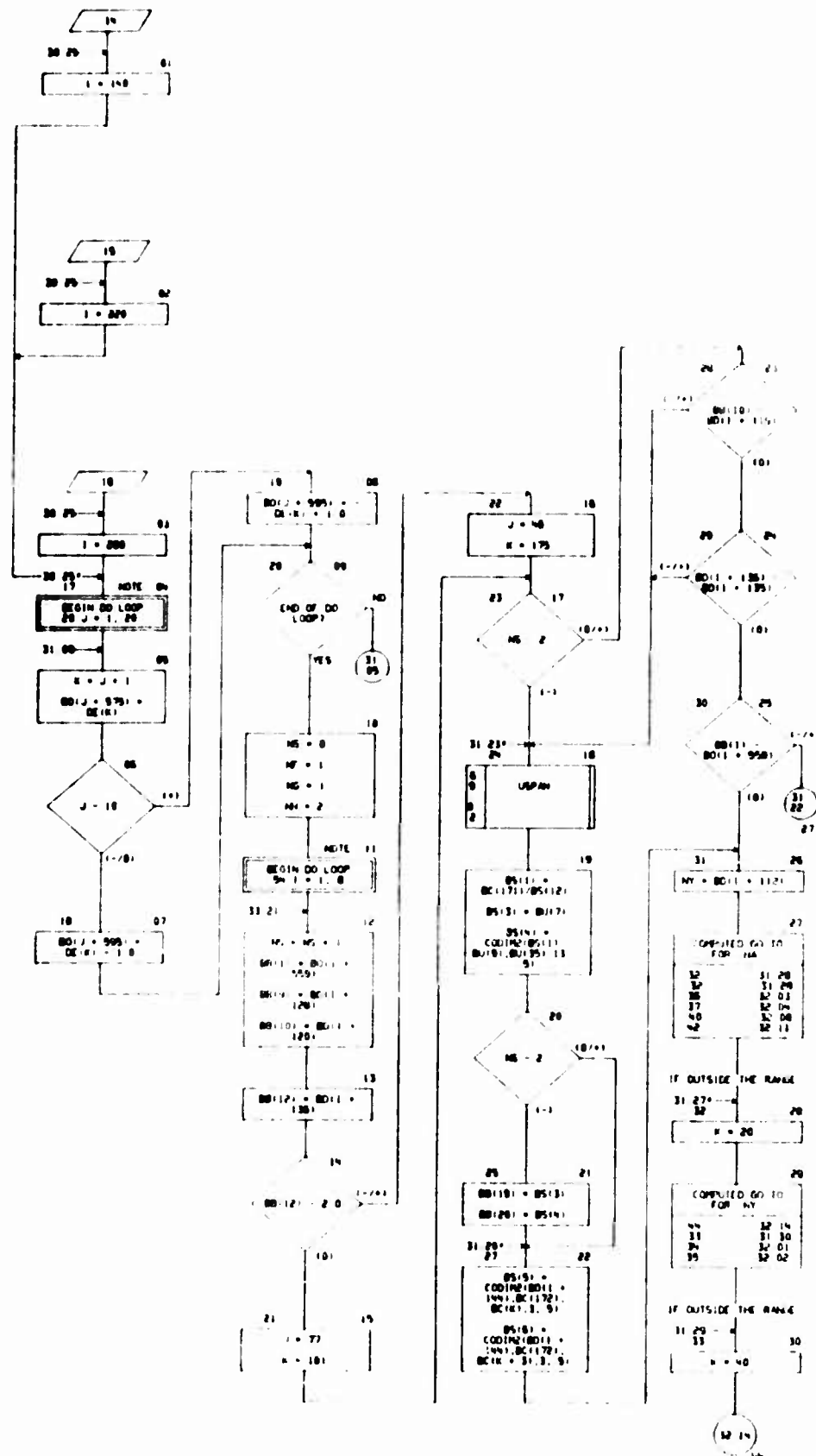


CHART TITLE - SUBROUTINE - ATMG



07/03/74

CHART TITLE - SUBROUTINE FATHS

AUTOFLOW CHART SET - DEEP AIRBORNE OVERLAY

PAGE 22

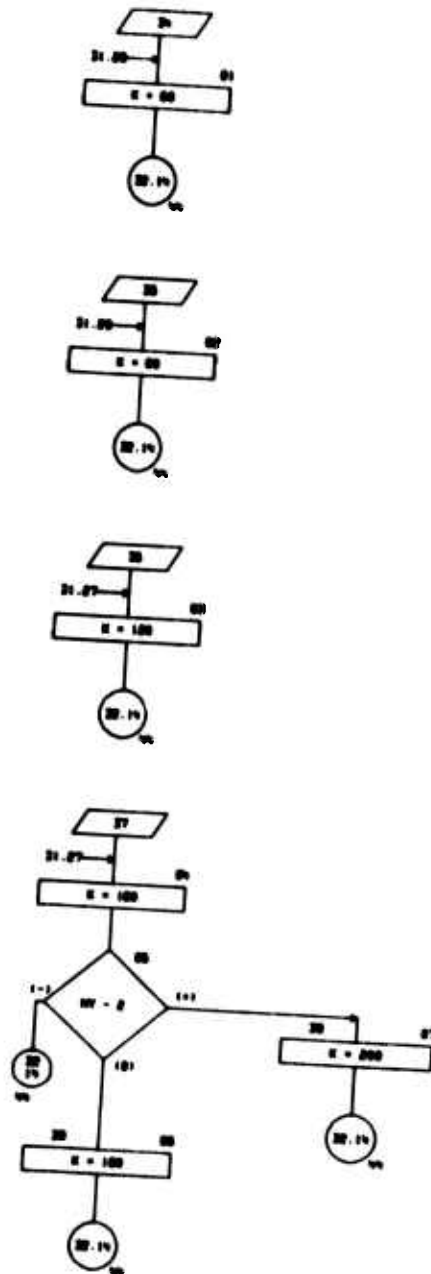


CHART TITLE - SUBROUTINE FATHG

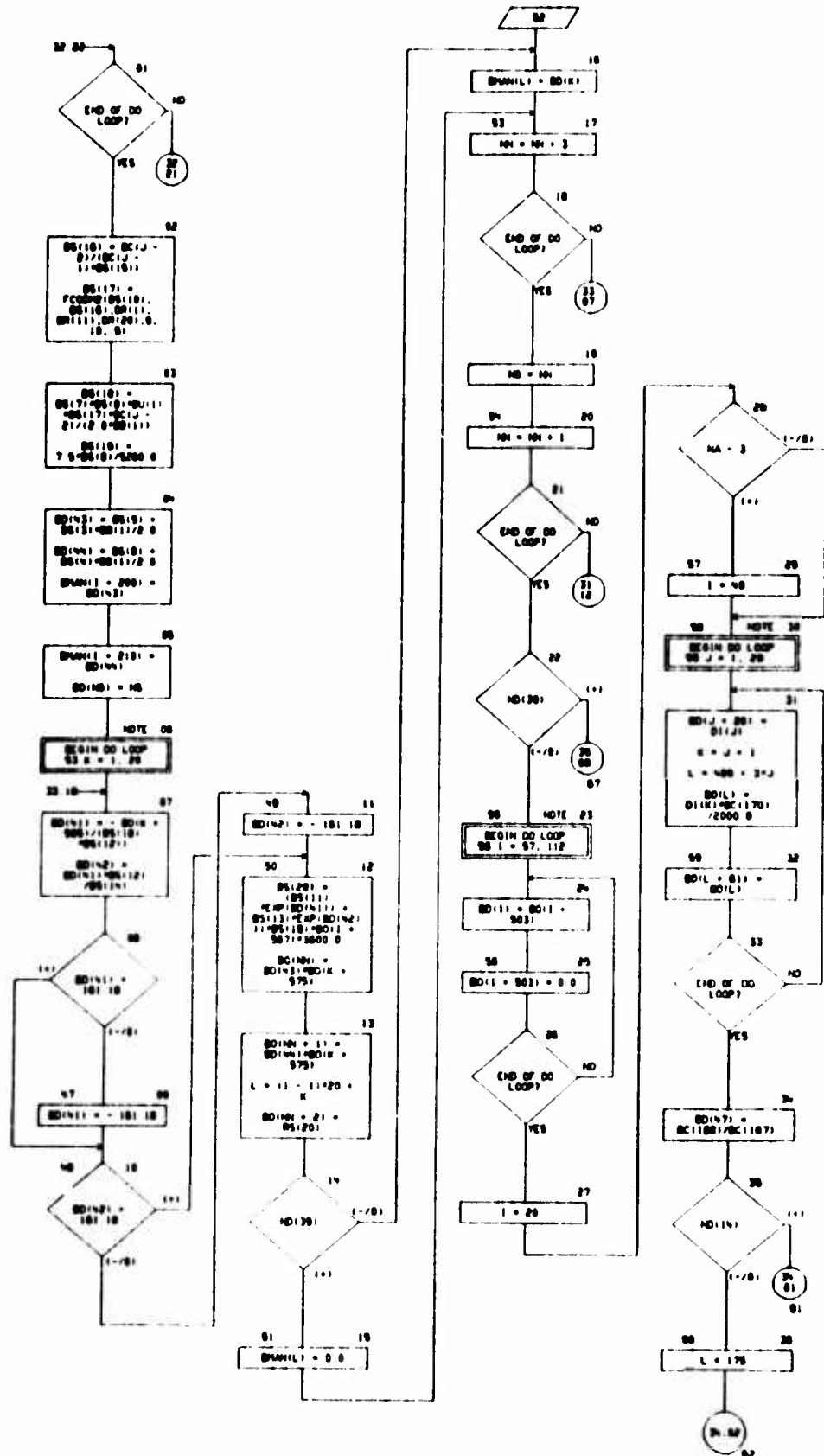


CHART TITLE - SUBROUTINE FATHS

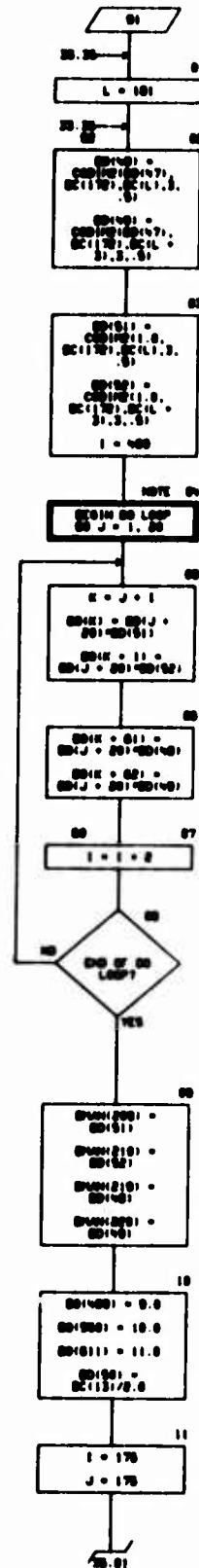


CHART TITLE - SUBROUTINE NAME

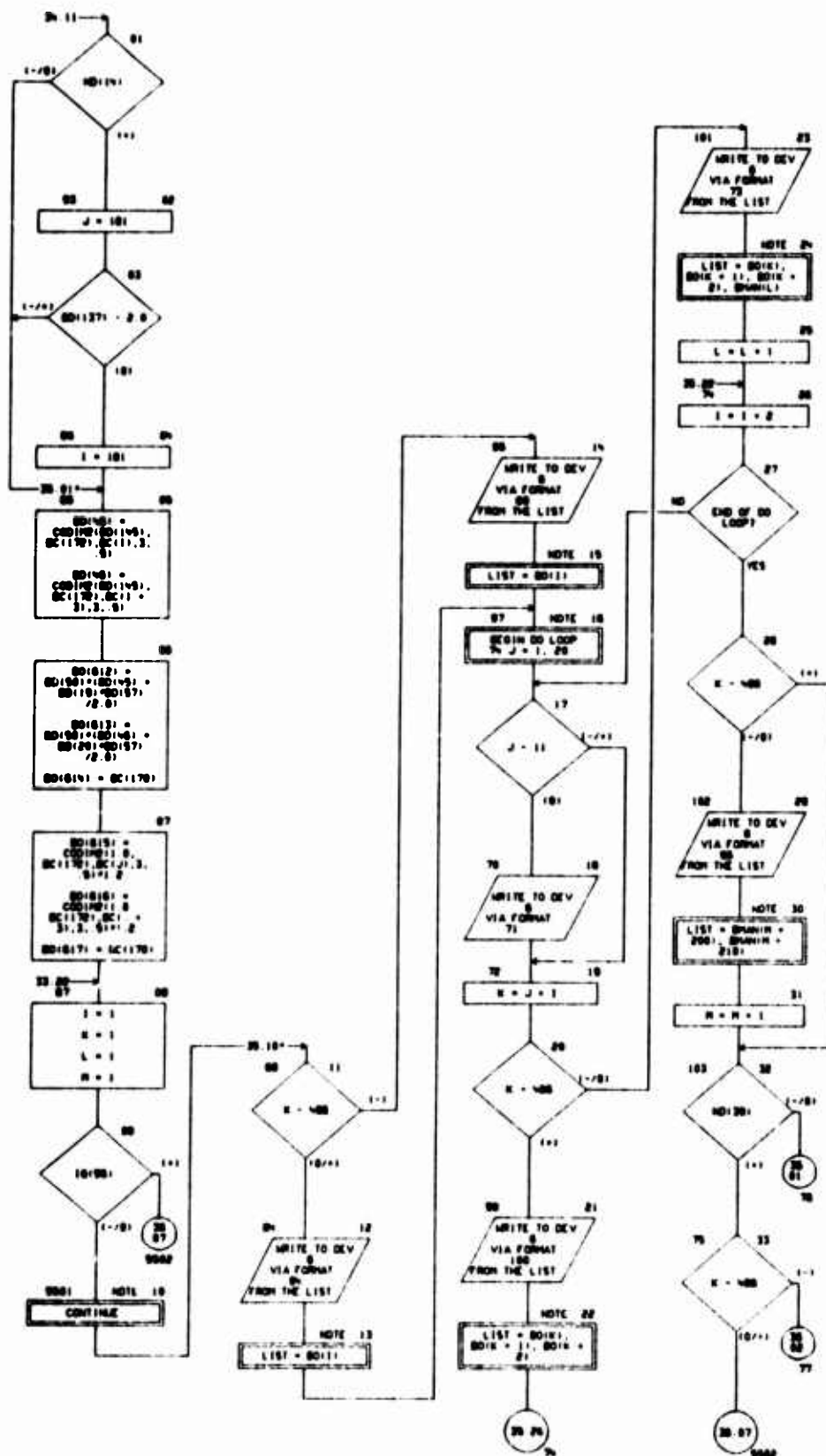


CHART TITLE - SUBROUTINE FAIRB

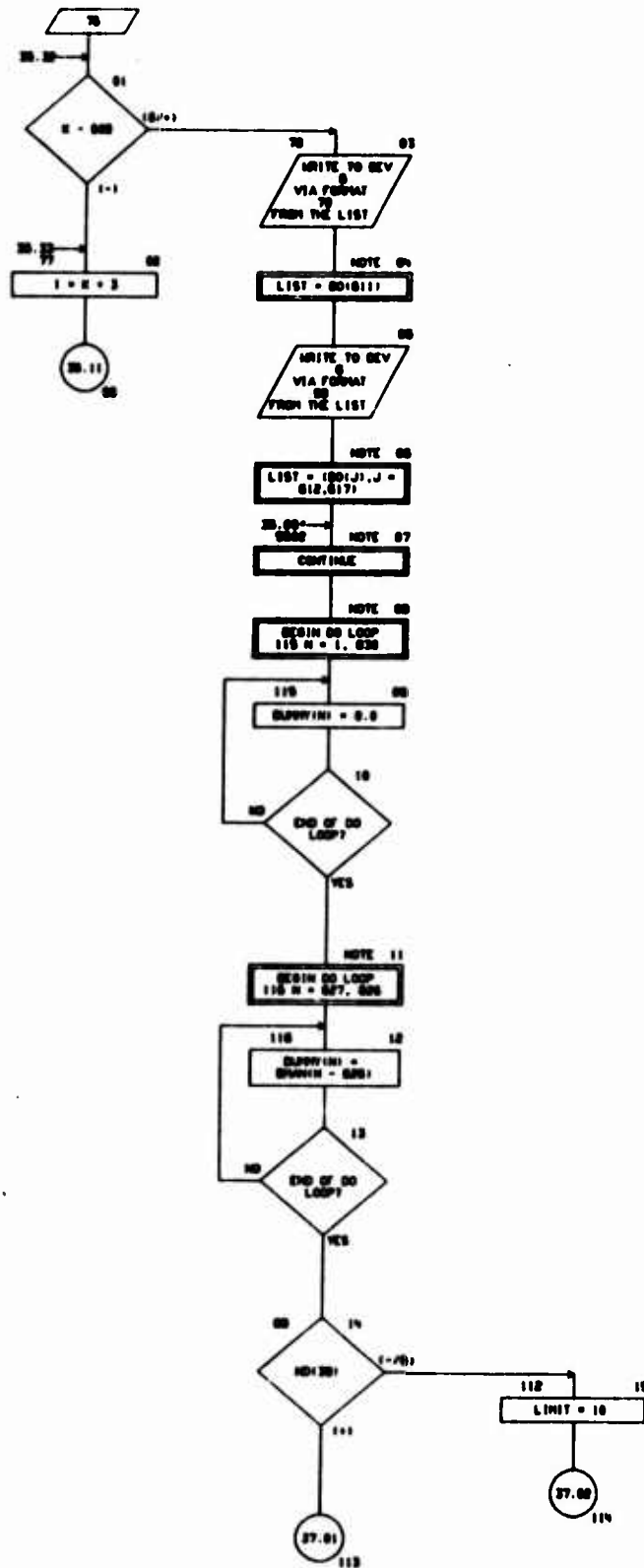


CHART TITLE - SUBROUTINE PAINS

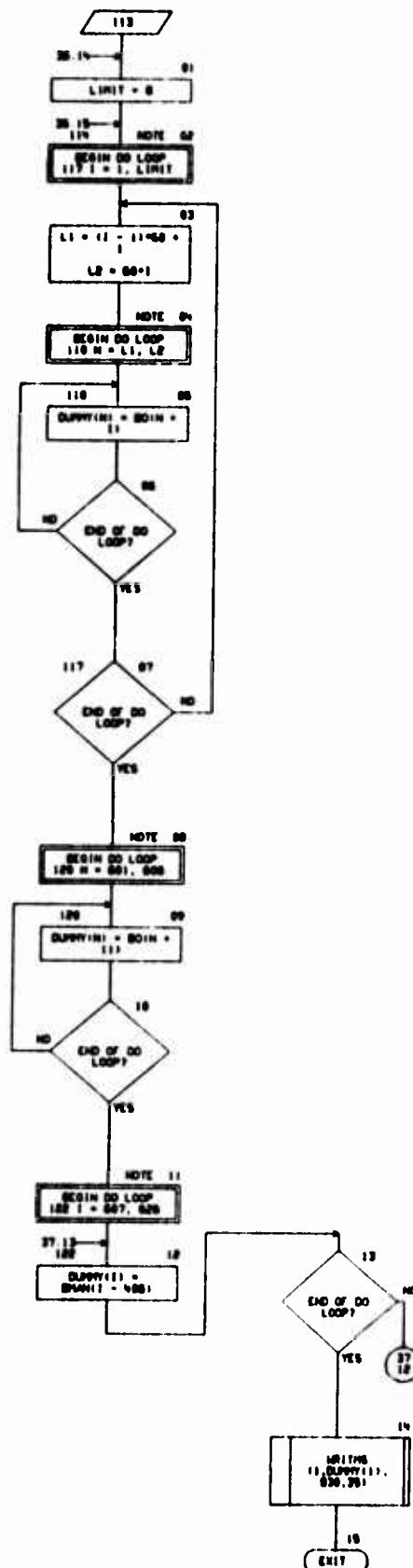


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(400)
COMMON /IPRINT/10(00)
DIMENSION DS(200),DE(240),DI(60),DO(70),DR(100),ND(200),DC(100),DS
(100),ED(100),DU(200),DD(60),DB(20),DMM(200)
DIMENSION SAMP(600)
EQUIVALENCE (DS(1),TCON(1000)),(DE(1),TCON(2177)),(DI(1),TCON(2517
)),(DO(1),TCON(2577)),(DR(1),TCON(2640)),(ND(1),TCON(4201)),(DC(1)
,TCON(2700)),(ED(1),TCON(2831)),(DS(1),TCON(2973)),(DD(1),TCON(3200
3)),(DU(1),TCON(3153)),(DB(1),TCON(3301)),(NA,ND(13)),(NF,ND(100))
,(NS,ND(100)),(NB,ND(101)),(NH,ND(102)),(NV,ND(103)),(I,ND(104)),(
J,ND(105)),(K,ND(106)),(L,ND(107)),(M,ND(108))
64 FORMAT(1H,DSK,2H*** FATHS - (P100) ****
      4X,10SPECTRA SEGMENT ND,P4.0//7X,
      10H00F SEND MDM,SK,10H40S SEND MDM,SK,10HCEEDANCES-TAKI)
66 FORMAT(1H,DSK,2H*** FATHS - (P100) ****
      4X,10SPECTRA SEGMENT ND,P4.0//7X,
      10H00F SEND MDM,SK,10H40S SEND MDM,SK,10HCEEDANCES-GUST,SK,
      10HCEEDANCES-VAN)
71 FORMAT(1H0)
100 FORMAT(1H 4HF10.0,5E12.0)
73 FORMAT(1H 4HF10.0,5E12.0,5E12.0)
65 FORMAT(//11H010 00F M =,F10.0,SK,10H10 MDS M =,F10.0)
70 FORMAT(1H,DSK,2H*** FATHS - (P100) ****
      4X,10SPECTRA SEGMENT ND,P4.0//SK,2HGROUND-AIR-ORD
      10H CYCLES//7X,27H00F SEND MDM 10S SEND MDM,7X,11H00CURRENCES)
60 FORMAT(1H ,4X,0F10.0,SK,1E12.0 )

```

CHART TITLE - INTRODUCTORY COMMENTS

FUNCTION ROOMS

CHART TITLE - FUNCTION FCODEN(X,Y,XI,YI,ZI,NI,ND,NK)

THE DIMENSION CURVE
 F17
 SUBPROGRAM... FCODE
 CALLING SEQUENCE
 F =
 FCODEN(X,Y,XI,YI,ZI,
 NI,ND,NK)
 X = ARGUMENT - 1ST
 VARIABLE
 Y = ARGUMENT - 2ND
 VARIABLE
 XI = ARGUMENT - 1ST
 VARIABLE
 YI = ARGUMENT - 2ND
 VARIABLE
 ZI = ARGUMENT - 3RD
 VARIABLE
 NI = NO. OF POINTS -
 1
 ND = NO. OF POINTS -
 1
 NK = END INTERNAL
 INTERPOLATION CONTROL
 CONSTANT

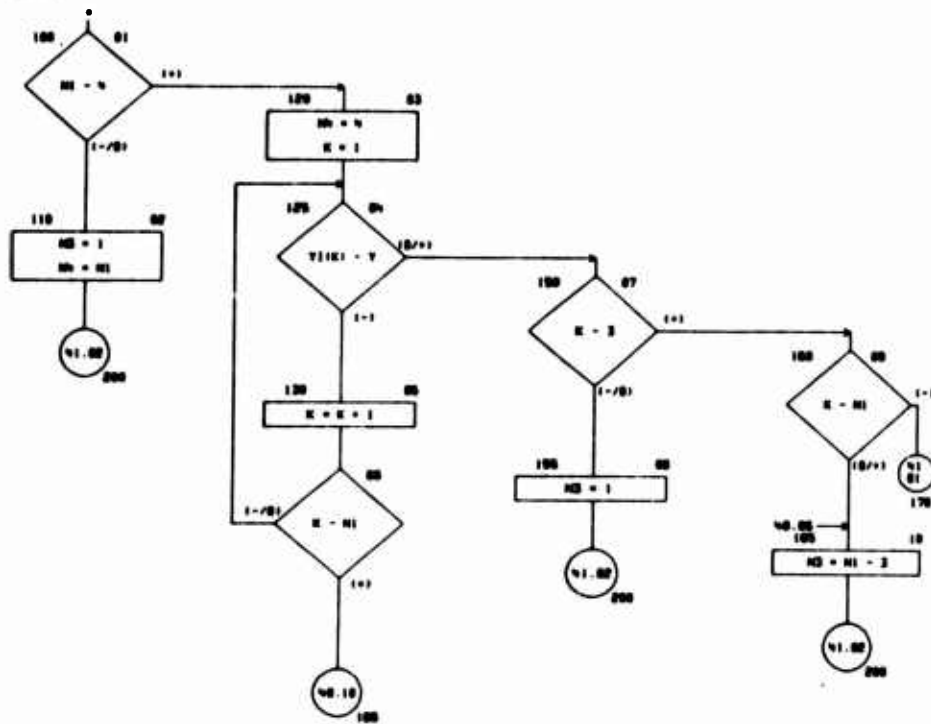
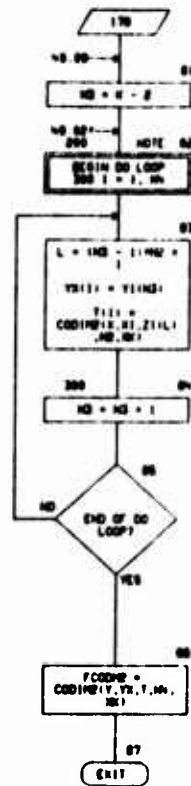


CHART TITLE - FUNCTION FCODEP(X,V,XI,YI,ZI,M1,M2,M3)



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AUTOFLOW CHART SET - SLEEP AIRLOADS OVERLAY

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CHART TITLE - NON-PROCEDURAL STATEMENTS

CONVEN TCON14001

DIMENSION NO1000, N1111, Y1111, Z1111, T141, Y141

EQUIVALENCE NO111, TCON1400111,

NO1, NO11011, NO1, NO11011, 11, NO114011, 11, NO114111, 11,

NO114011

07:03:24

AUTOFLOW CHART SET SWEEP AIRLOADS OVERLAY

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CHART TITLE INTERLUCKTORY CURRENTS

SCHEDULE TIME 07:04:01

CHART TITLE - SUBROUTINE FUSHE1

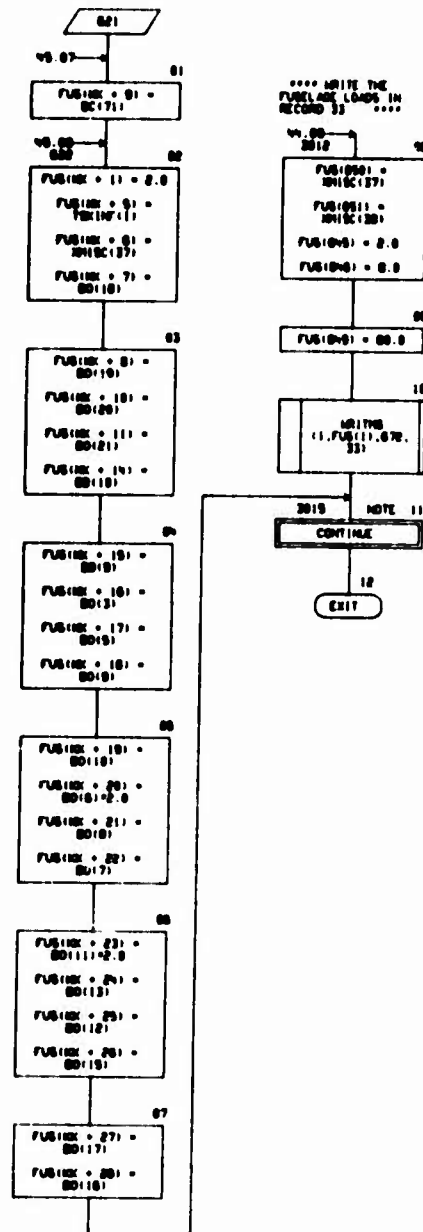


CHART TITLE - NON PROCEDURAL STATEMENTS

```
COMMON TCOM(400),MD(300),IDUM(132),SA(41132),INE(1132)
      RNZ(132),RC(44),SW(100),STEMP(23),STEMPH(23),STEMPV(23),
      DUM(100)
COMMON /MISC/ MISC(100)
DIMENSION MD(200),BO(620),BC(105)
DIMENSION TSWINF(23)
DIMENSION PLUS(670),BB(20)
EQUIVALENCE (MD(1),TCOM(400)),(BO(1),TCOM(300)),(BC(1),TCOM(270)
1),1,MD(10)),(BB(1),TCOM(295))
EQUIVALENCE (SW(10),TSWINF(1))
```

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AUTOFLEX CHART SET - SHEEP AIRLOADS OVERLAY

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CHART TITLE - INTERDUCTORY CONTENTS

.....
SUBROUTINE MAULDS
.....

[illegible]

CHART TITLE - SUBROUTINE MAULD

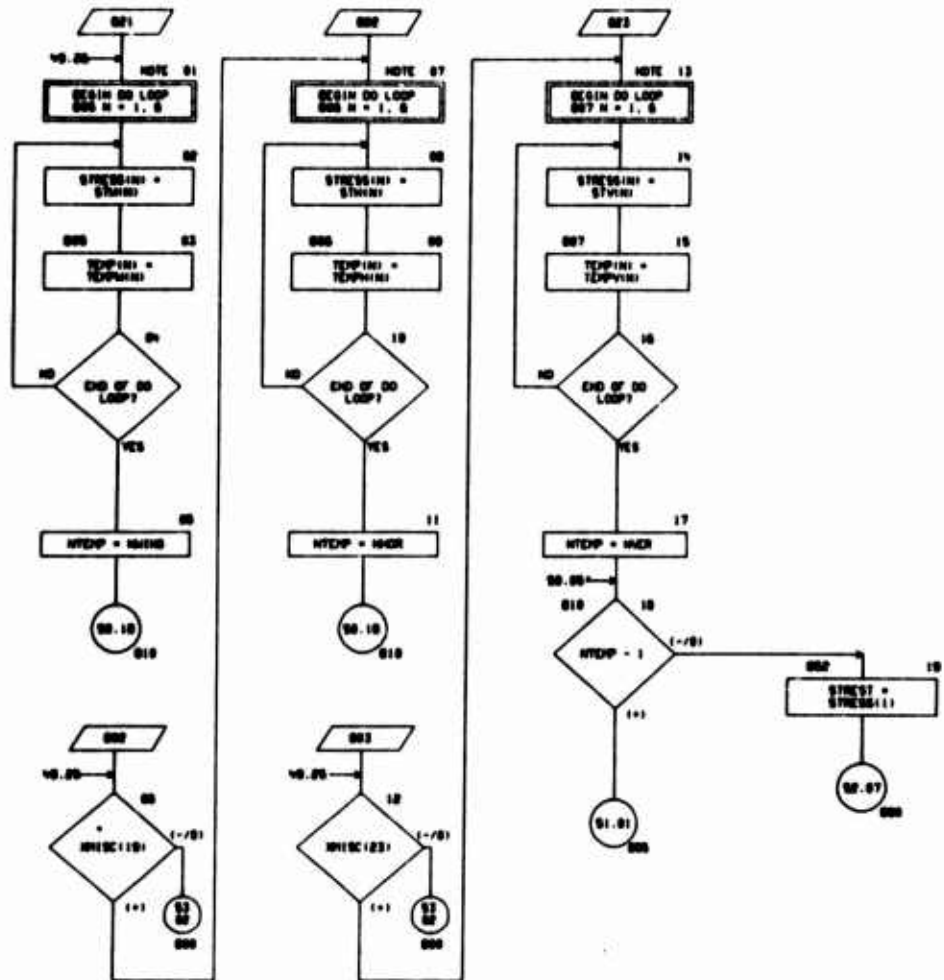
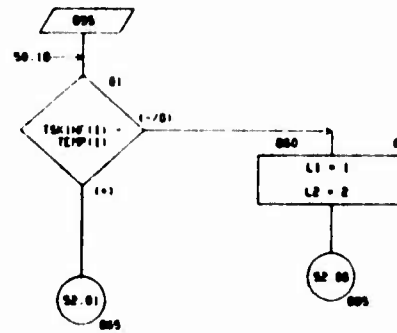


CHART TITLE - SUBROUTINE HARDS



```

graph TD
    000[000] -- 01.01 --> 01{TEMP(I) =  
TEMP(NTEMP)}
    01 -- 1 --> 02((02.00))
    01 -- 1 --> 07B[07B  
NOTE 02  
BEGIN DO LOOP  
DO N = 2, NTEMP]
    07B --> 03{TEMP(I) =  
TEMP(N)}
    03 -- 1 --> 04[L1 = N - 1  
L2 = N]
    03 -- 1 --> 05{END OF DO  
LOOP?}
    05 -- NO --> 06[01.02  
000]
    05 -- YES --> 07[07  
STRESS =  
STRESS(L2) -  
STRESS(L1) -  
STRESS(L1) -  
TEMP(L2) -  
TEMP(I) -  
TEMP(L2) -  
TEMP(L1)]
    07 -- 00.10 --> 08[08  
COMPUTED GO TO  
FOR K  
011 02.00  
012 02.10  
013 03.01]
    08 --> 09{IF OUTSIDE THE RANGE  
02.07  
011}
    09 --> 10[10  
STEPH(I) =  
STRESS  
RSH =  
RSH/STEPH(I)]
    10 --> 03.02((03.02))
    06 --> 012[012  
STEPH(I) =  
STRESS  
RSH =  
RSH/STEPH(I)]
    012 -- 03.07 --> 03.02
    03.02 --> 000
  
```

CHART TITLE - SUBROUTINE HANDS

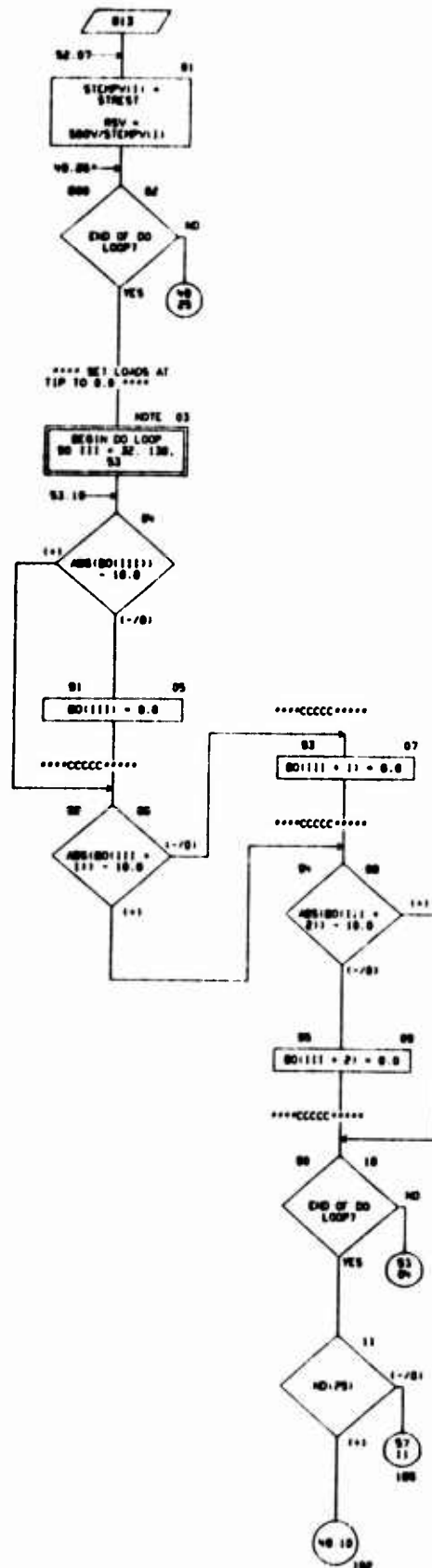


CHART TITLE - SUBROUTINE HAILDS

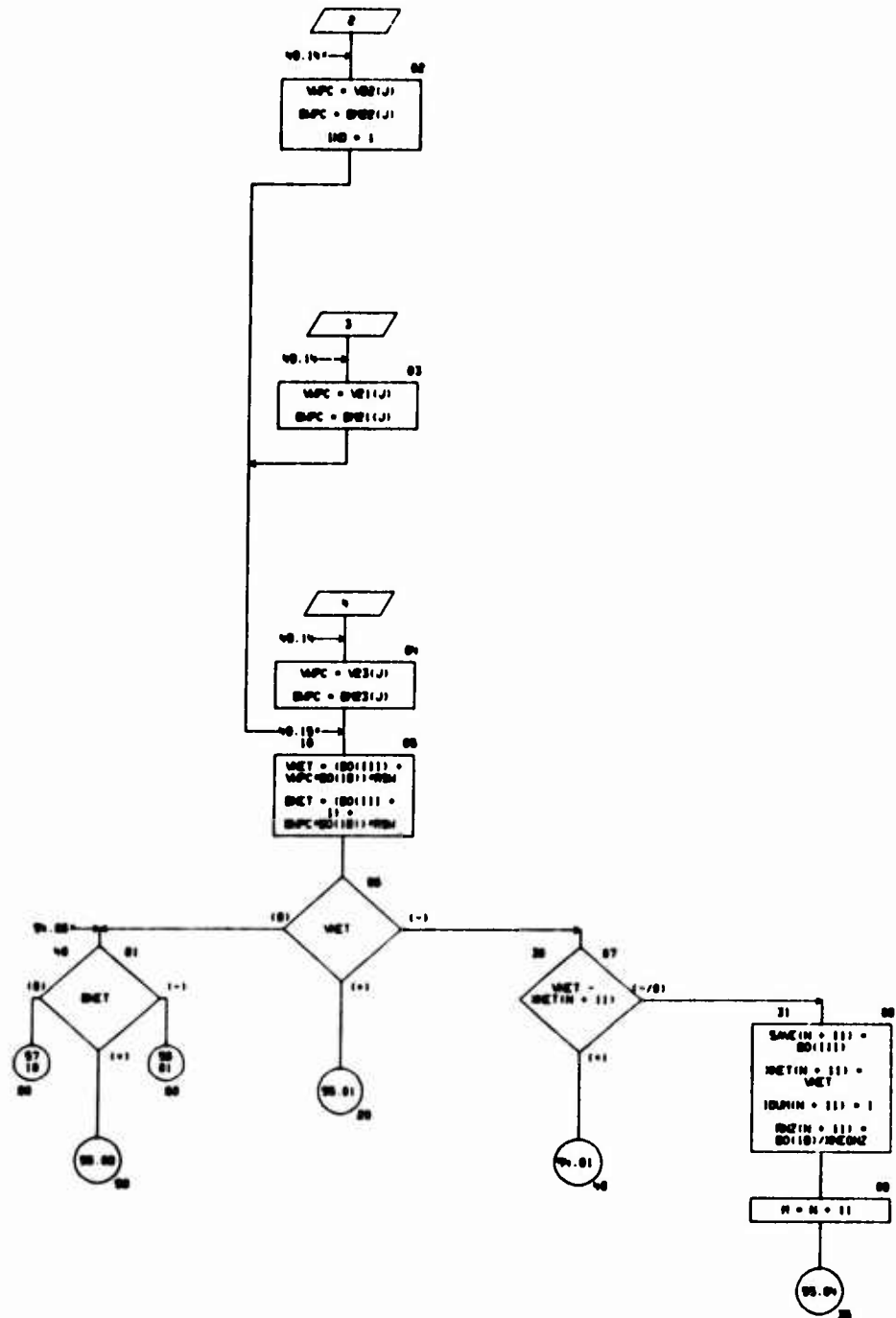


CHART TITLE - SUBROUTINE PARLOS

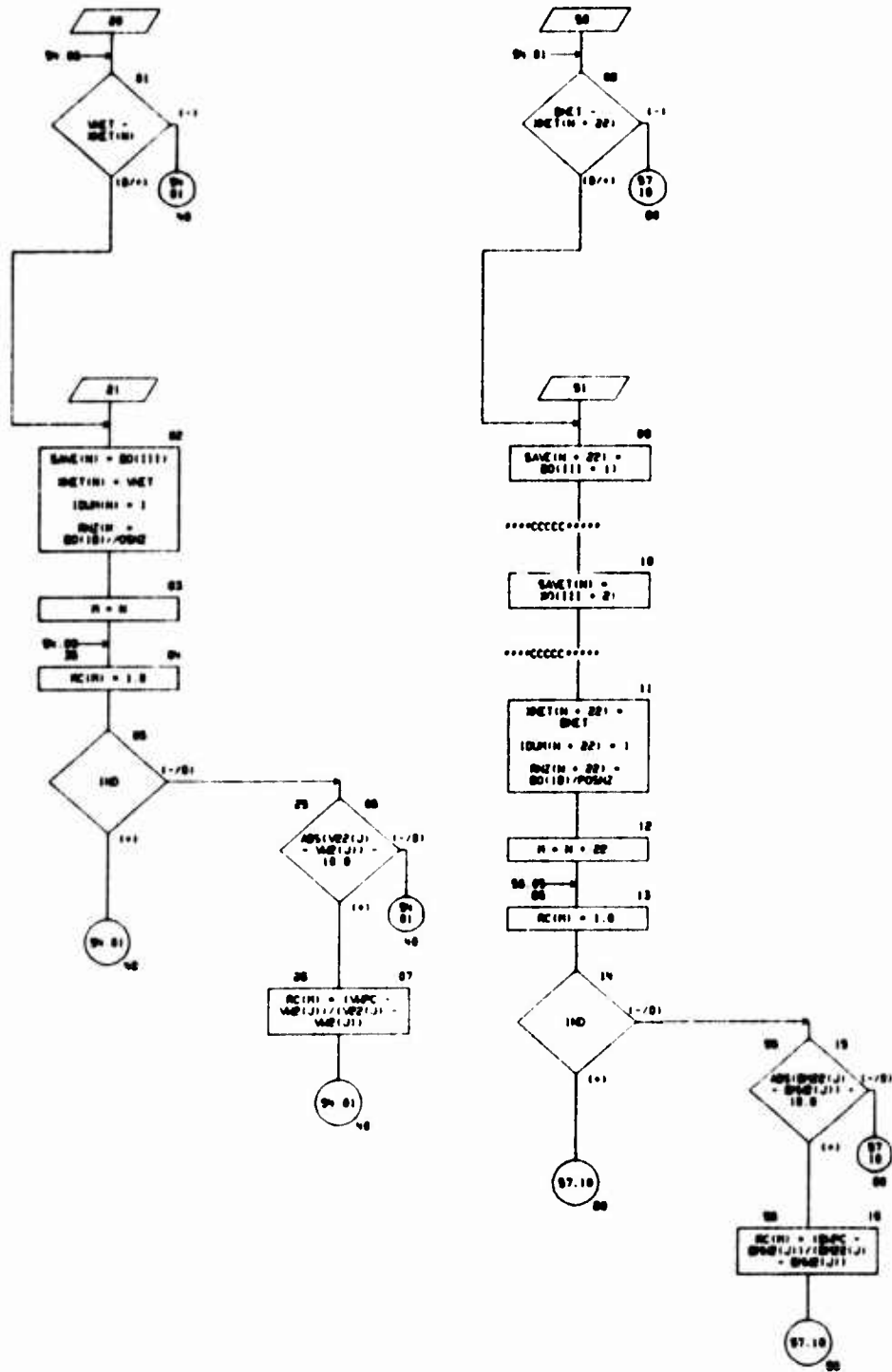


CHART TITLE - SUBROUTINE PARLO6

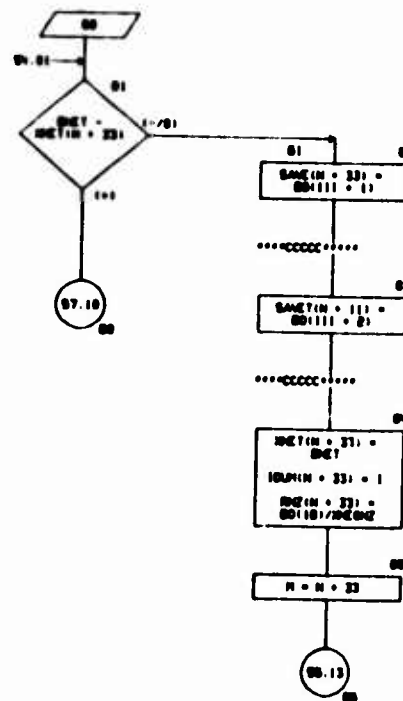


CHART TITLE - SUBROUTINE HAMBUS

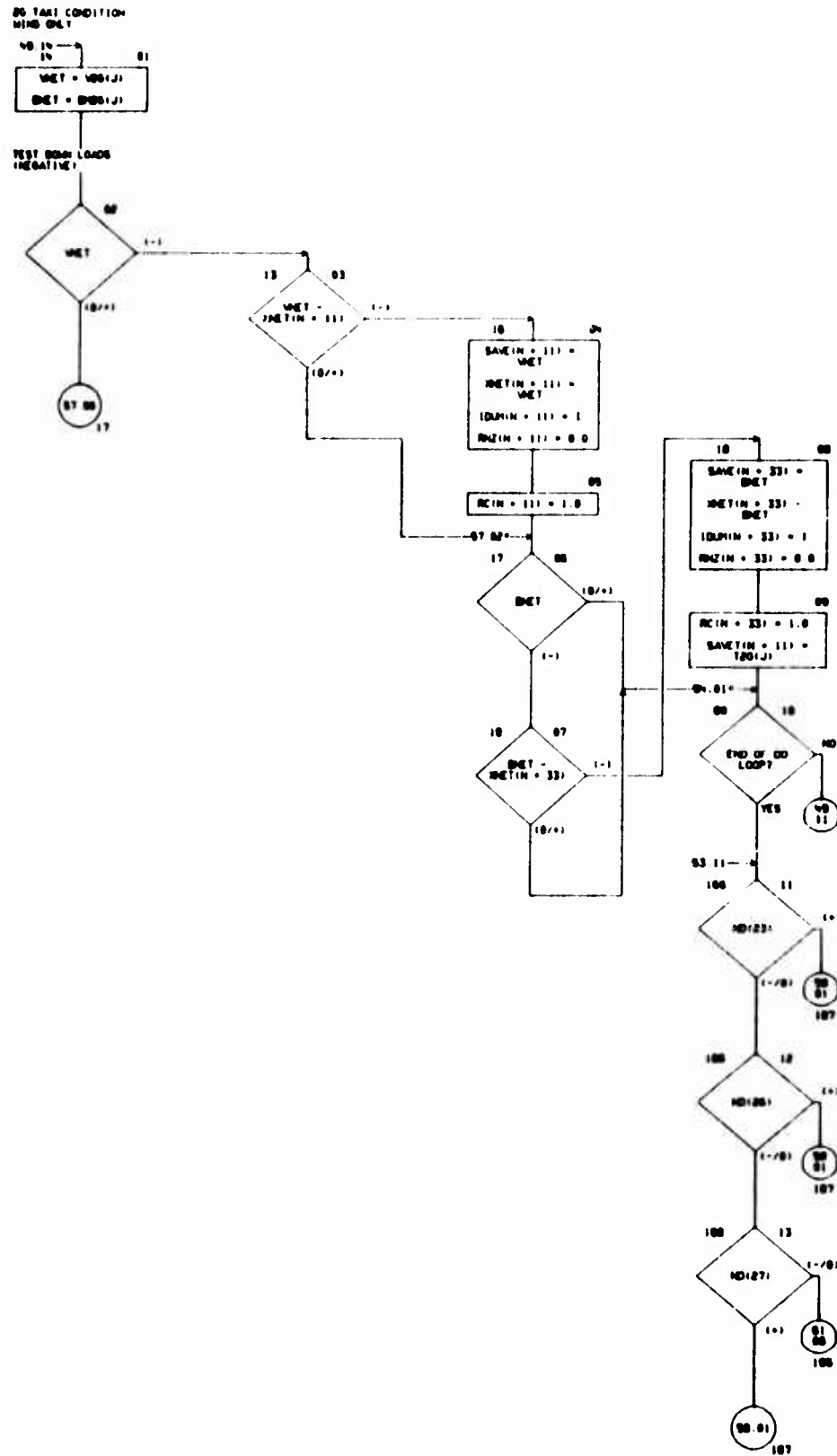


CHART TITLE - SUBROUTINE NAME(S)

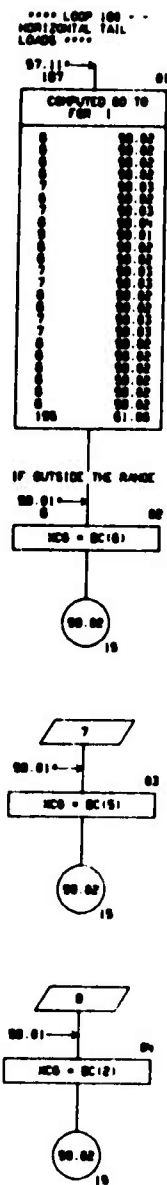


CHART TITLE - SUBROUTINE HADLOS

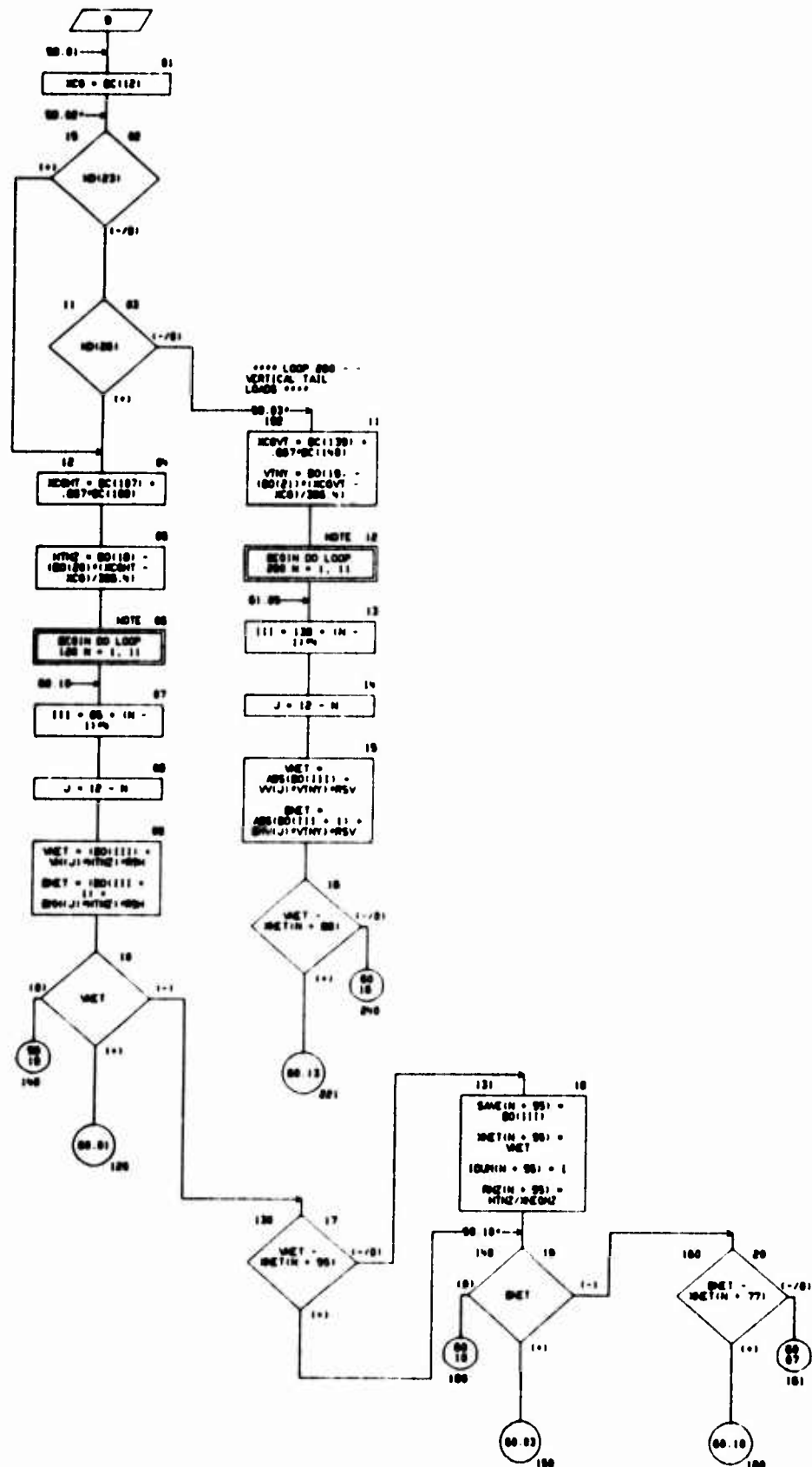


CHART TITLE - SURVEILLANCE MAILING

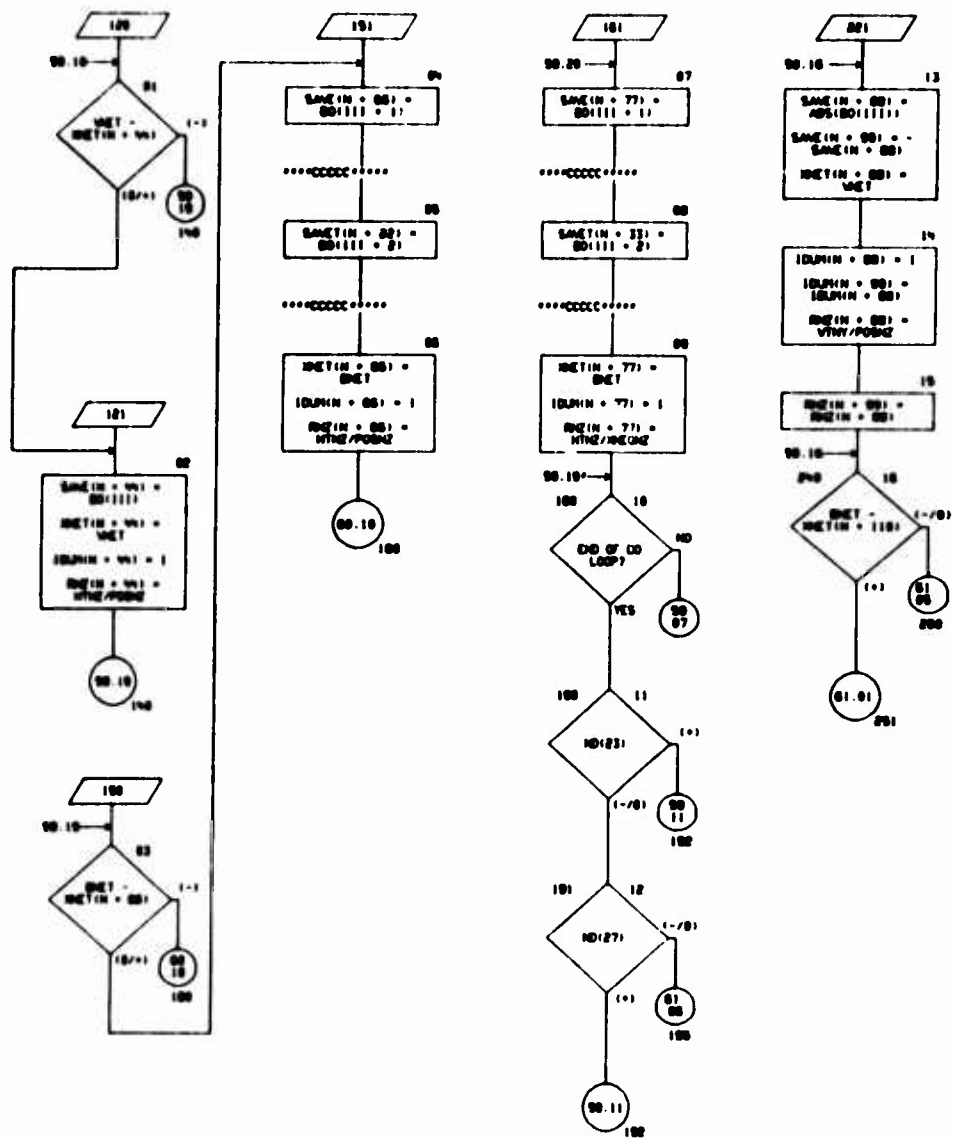


CHART TITLE - SUBROUTINE PARLOD

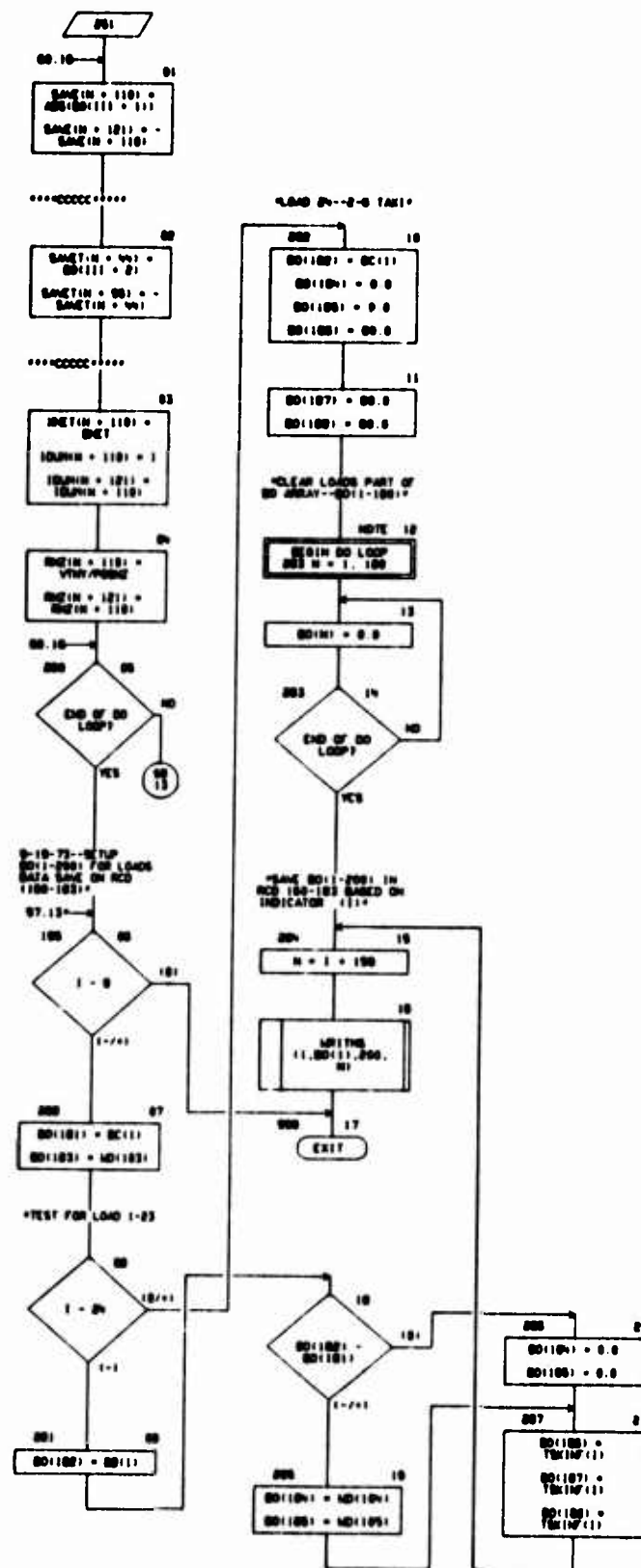


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

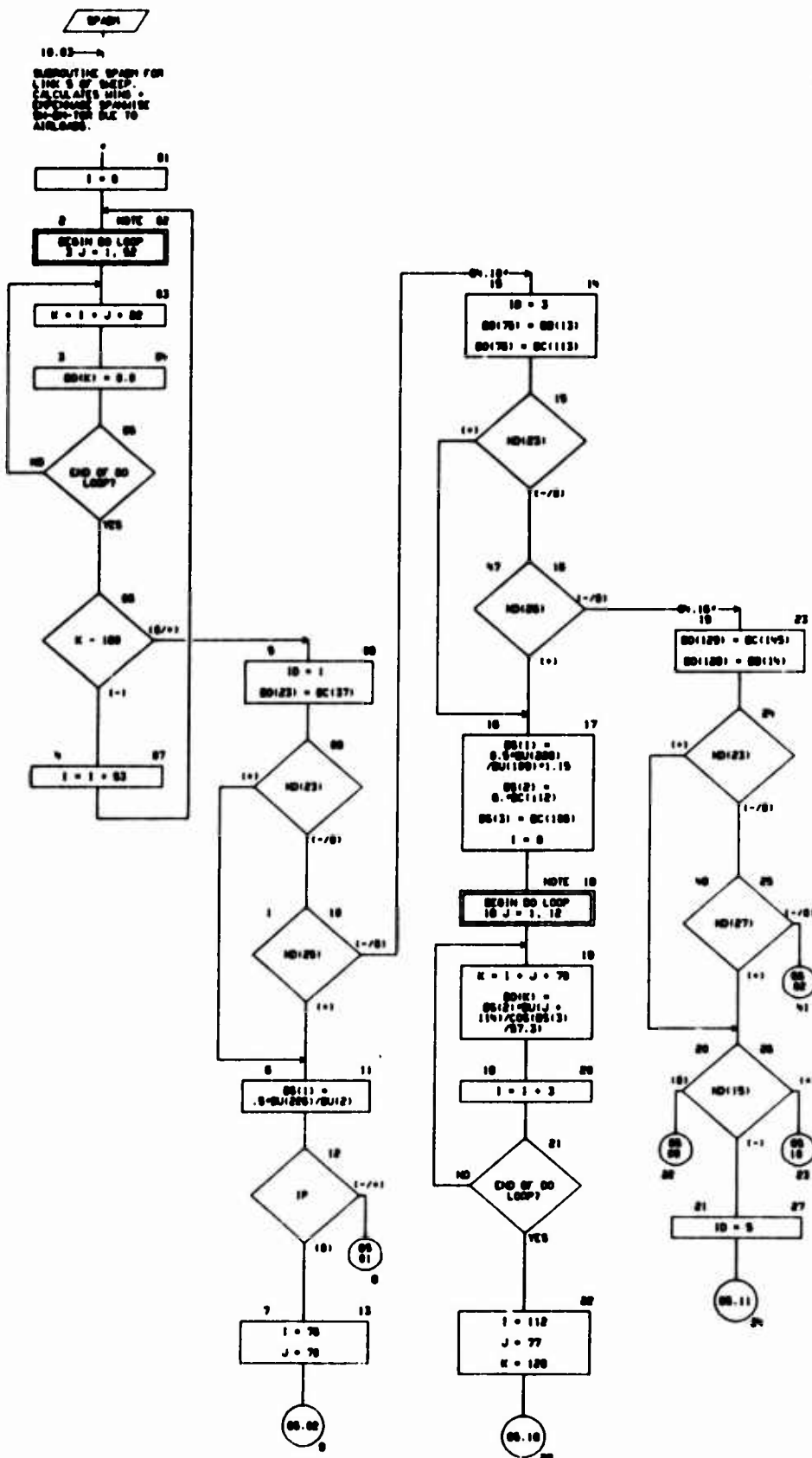
COMMON TCON(400),MLD(300),TDPM(132),SAC(132),HET(132),
      RHE(132),RC(44),SW(100),STEPH(23),STEPV(23),STEPV(23),
      ,BPM(100),SAC(100)
COMMON /MISC/ MISC(100)
DIMENSION MD(200),BD(620),TKINF(23),STRES(6),TEMP(6),
      ,STH(6),STV(6),STV(6),TEMPH(6),TEMPH(6),TEMPV(6)
DIMENSION BC(100)
DIMENSION BB(20)
DIMENSION MML(1024),MD(200)
DIMENSION VBS(11),BPS(11),T20(11),WM(11),BPM(11),
      ,VBS(11),BPS(11),VBS(11),BPS(11),
      ,VBS(11),BPS(11),VBS(11),BPS(11),
      ,VBS(11),BPS(11),WM(11),BPM(11),
      ,WM(11),BPM(11)
EQUIVALENCE (MD(1),TCON(420)), (BD(1),TCON(330)), (1,MD(10))
EQUIVALENCE
      (SW(116),TKINF(11)),
      (SW(130),STH (11)),
      (SW(146),TEMPH (11)), (SW(153),STH (11)), (SW(160),TEMPH (11)),
      (SW(167),STV (11)), (SW(173),TEMPV (11)), (SW(181),MINS (1)),
      (SW(188),HGR (1)), (SW(170),HGR (1)), (SW(152),BBH (1)),
      (SW(185),BBH (1)), (SW(180),BBV (1))
EQUIVALENCE (BBSH,MLD(1)), (PDSH,MLD(2)), (HDSH,MLD(3)),
      (VBS(1),MLD( 4)), (BPS(1),MLD( 5)), (T20(1),MLD( 20)),
      (WM(1),MLD( 27)), (BPM(1),MLD( 40)),
      (VBS(1),MLD( 70)), (BPS(1),MLD( 81)),
      (VBS(1),MLD(103)), (BPS(1),MLD(114)),
      (VBS(1),MLD(136)), (BPS(1),MLD(147)),
      (VBS(1),MLD(169)), (BPS(1),MLD(180)),
      (VBS(1),MLD(202)), (BPS(1),MLD(213)),
      (WM(1),MLD(235)), (BPM(1),MLD(246)),
      (WM(1),MLD(268)), (BPM(1),MLD(270))
EQUIVALENCE (BC(1),TCON(270))
EQUIVALENCE (BB(1),TCON(295))

```

CHART TITLE - INTRODUCTORY COMMENTS

SUBROUTINE SPAN
#####

CHART TITLE - SUBROUTINE SPASH



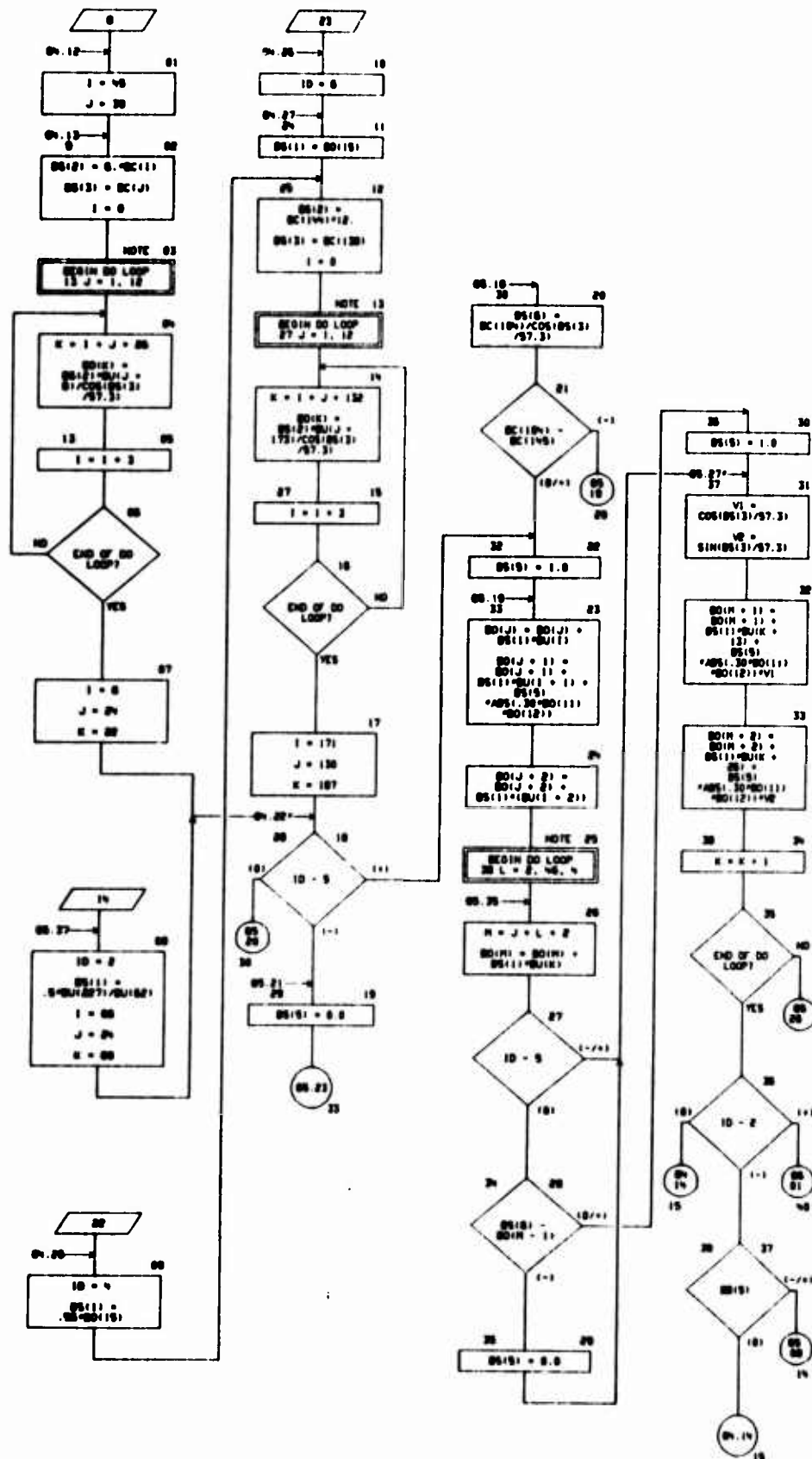


CHART TITLE - SUBROUTINE SPASH

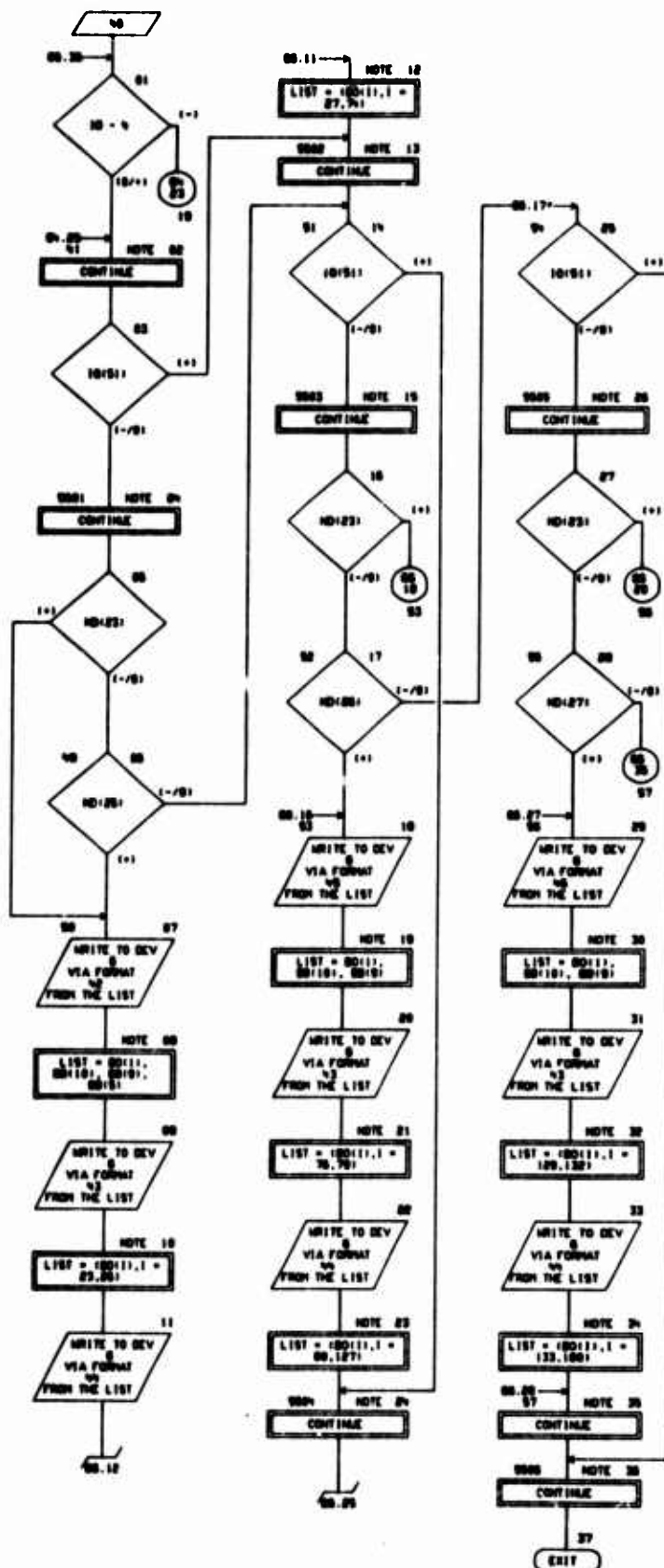


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(400)
COMMON /PRINT/IO(80)
DIMENSION BD(20),BC(100),BD(20),BS(20),BU(20),MD(200)
EQUIVALENCE (BC(1),TCON(275)),(BU(1),TCON(315)),(BD(1),TCON(330)
), (BS(1),TCON(297)),(BD(11),TCON(295)),(MD(1),TCON(420)),(IP,MD
(137)),(ID,MD(104)),(I,MD(150)),(J,MD(155)),(K,MD(157)),(L,MD(159)
), (M,MD(160))
42  FORMAT(1H,3X,10H100 LOADS,3X,8HCOND NO=,F7.0,3X,3HPPH,7B.3,
    3X,4HULT=,F7.0,3X,3HDEF=,F6.2,15X,20H** SPAN = (P(5)) **)
43  FORMAT(1H,3X,8HCOND DIST=,F7.0,3HIN,4X,10HAT SOB=,F6.0,2HLO,4X,
    10HAT SOB=,F12.0,3HIN-LO,4X,10HAT SOB=,F12.0,3HIN-LO///
    13X,10HSTATION,10X,8HBEAR,10X,8HEND ROM,21X,8HRTS ROM/
    14X,10HIN,21X,10HLO,15X,10HIN-LO,22X,10HIN-LO//
44  FORMAT(1H,3X,2H,0,F20.0,F20.0)
45  FORMAT(1H,3X,2HORIZONTAL TAIL LOADS,3X,8HCOND NO=,F7.0,3X,
    3HPPH,7B.3,3X,4HULT=,F7.0,10X,20H** SPAN = (P(5)) **)
46  FORMAT(1H,3X,10HVERTICAL TAIL LOADS,3X,8HCOND NO=,F7.0,3X,
    3HPPH,7B.3,3X,4HULT=,F7.0,21X,20H** SPAN = (P(5)) **)

```


07/03/74

AUTOFLEX CHART SET - SHEEP AIRLOADS OVERLAY

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CHART TITLE - INTRODUCTORY COMMENTS

[illegible]


 00.110-2
 DETERMINE JAPAN FOR
 LINK S OF WEAP.
 SERVICE MIND AND
 EQUIPMENT UNIT
 AIR AND DISTRIBUTION
 AND FACTORS.

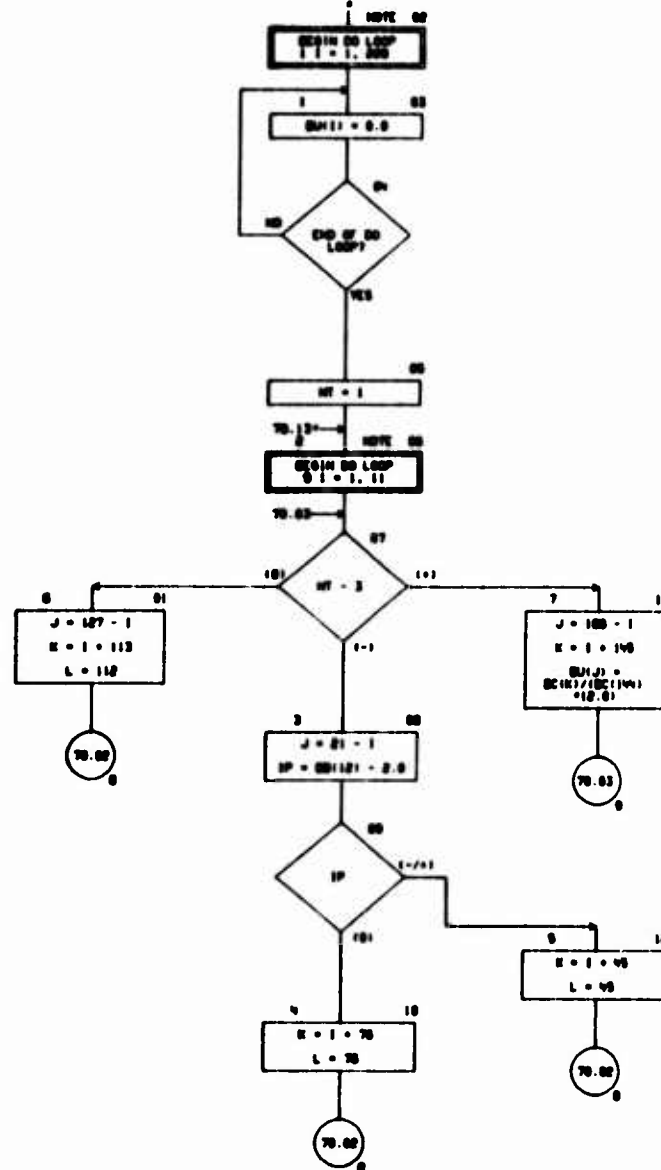
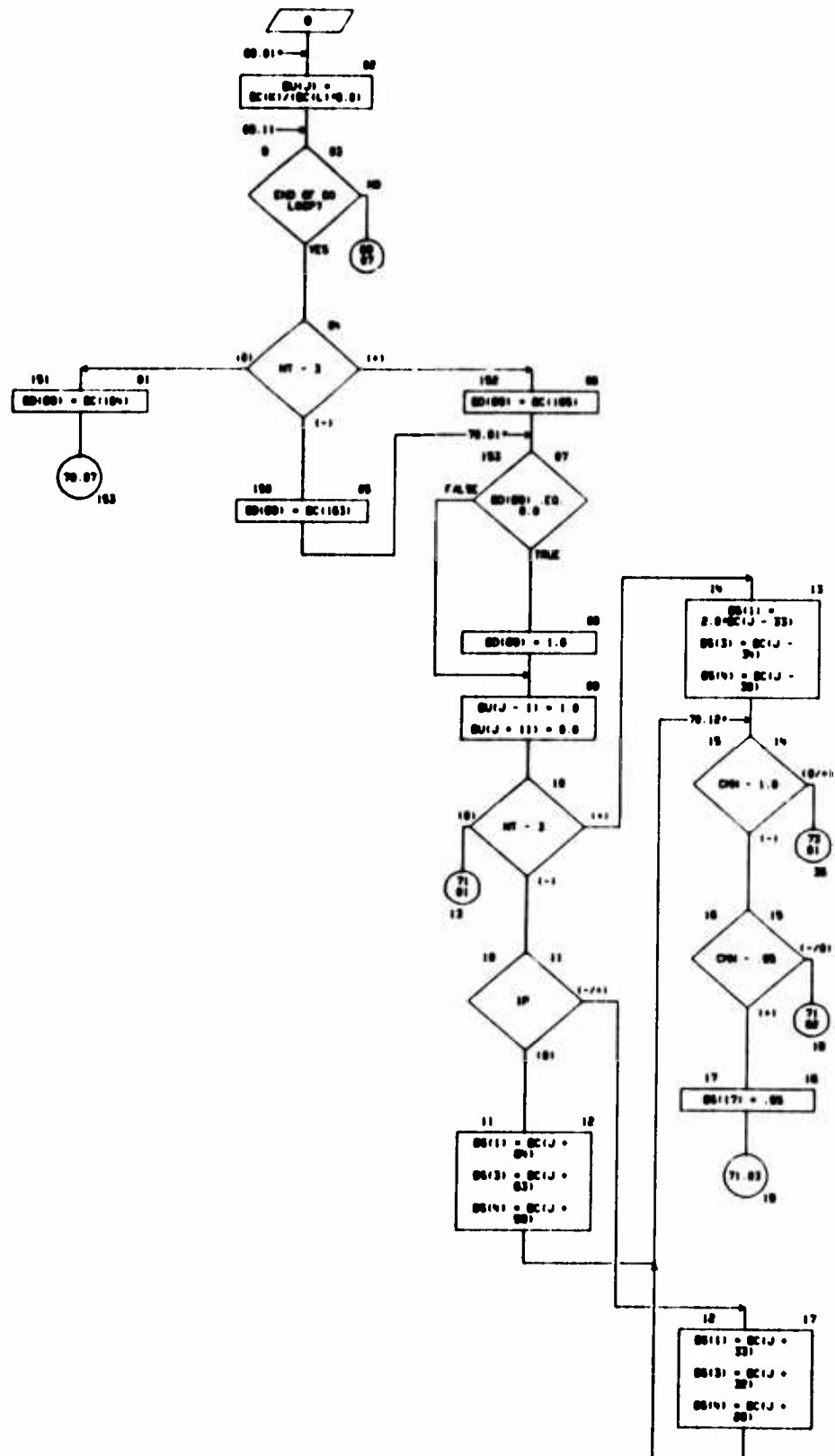


CHART TITLE - SUBROUTINE VOPAN



The flowchart is organized into three main vertical sections. The leftmost section contains steps 10 through 19, including calculations for $Q(1)$, $Q(2)$, $Q(3)$, and $Q(4)$, and a loop structure. The middle section contains steps 20 through 29, including a loop for i from 1 to n , a decision diamond for $i = 1$, and a loop for j from 1 to n . The rightmost section contains steps 30 through 39, including a loop for k from 1 to n , a decision diamond for $k = 1$, and a loop for l from 1 to n . The flowchart ends at step 40.

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CHART TITLE - SUBROUTINE USPHN

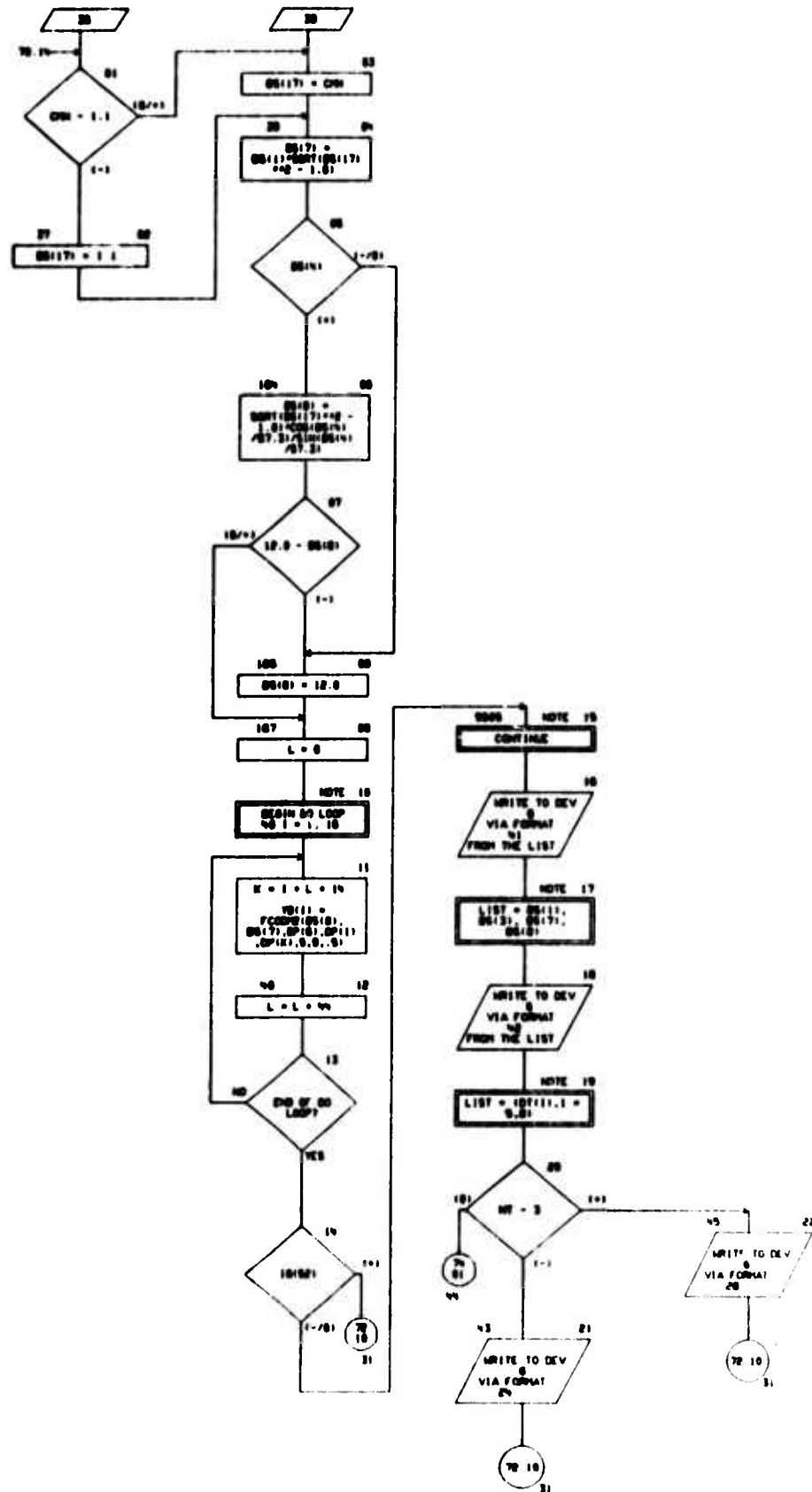


CHART TITLE - SUBROUTINE USPM

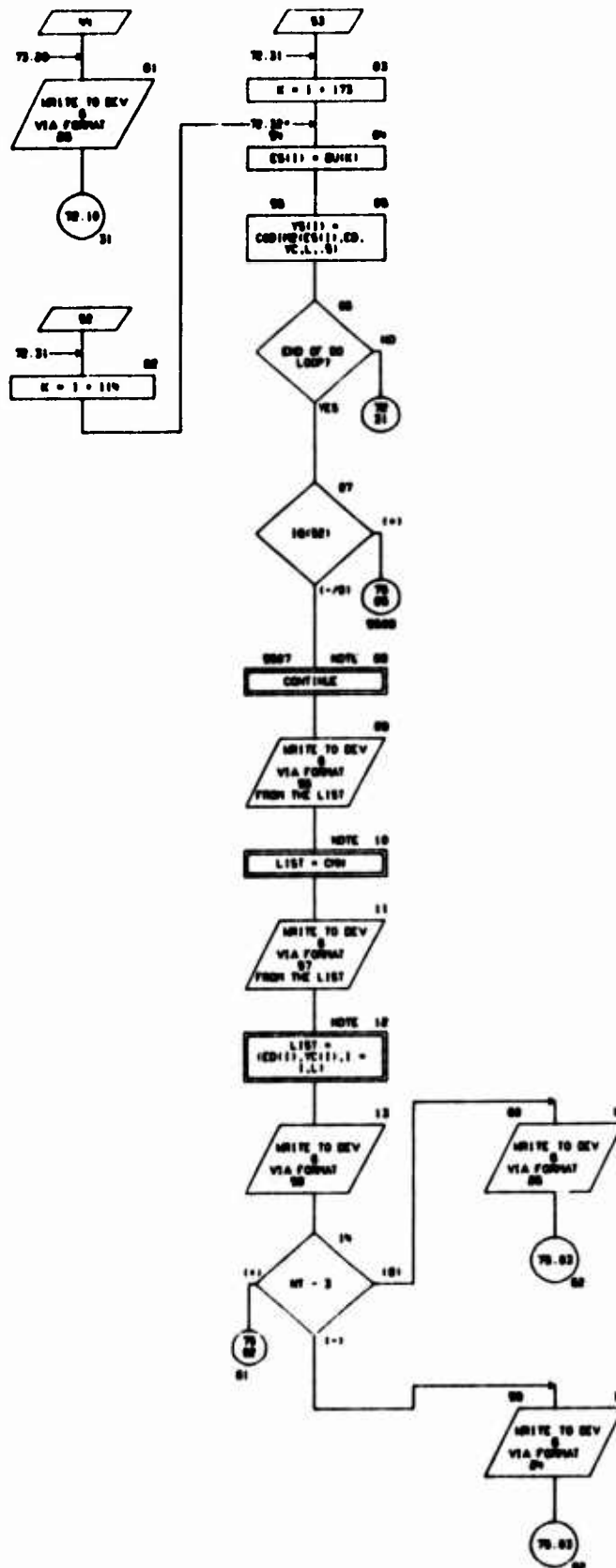


CHART TITLE - SUBROUTINE UJAPN

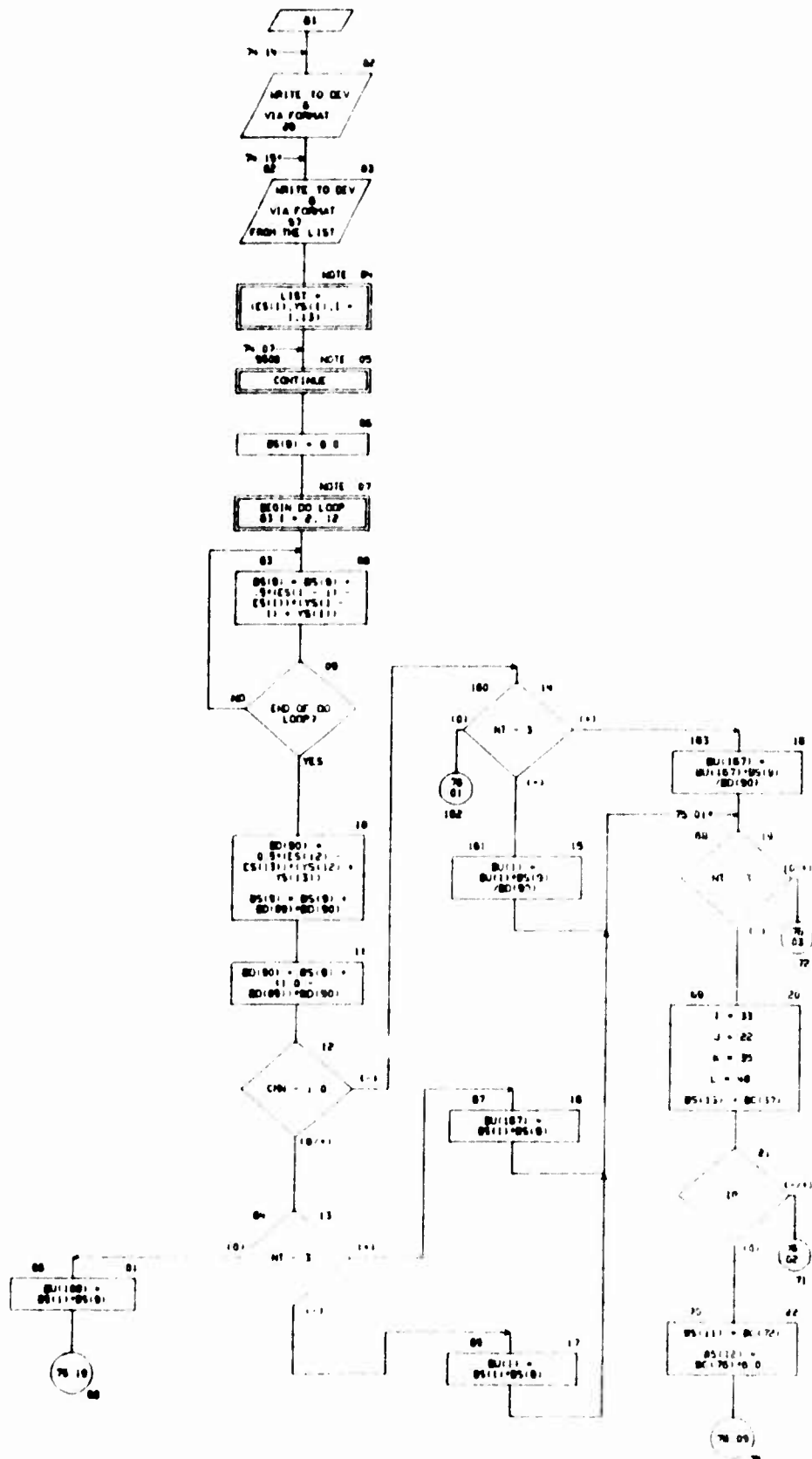


CHART TITLE - SUBROUTINE USFAN

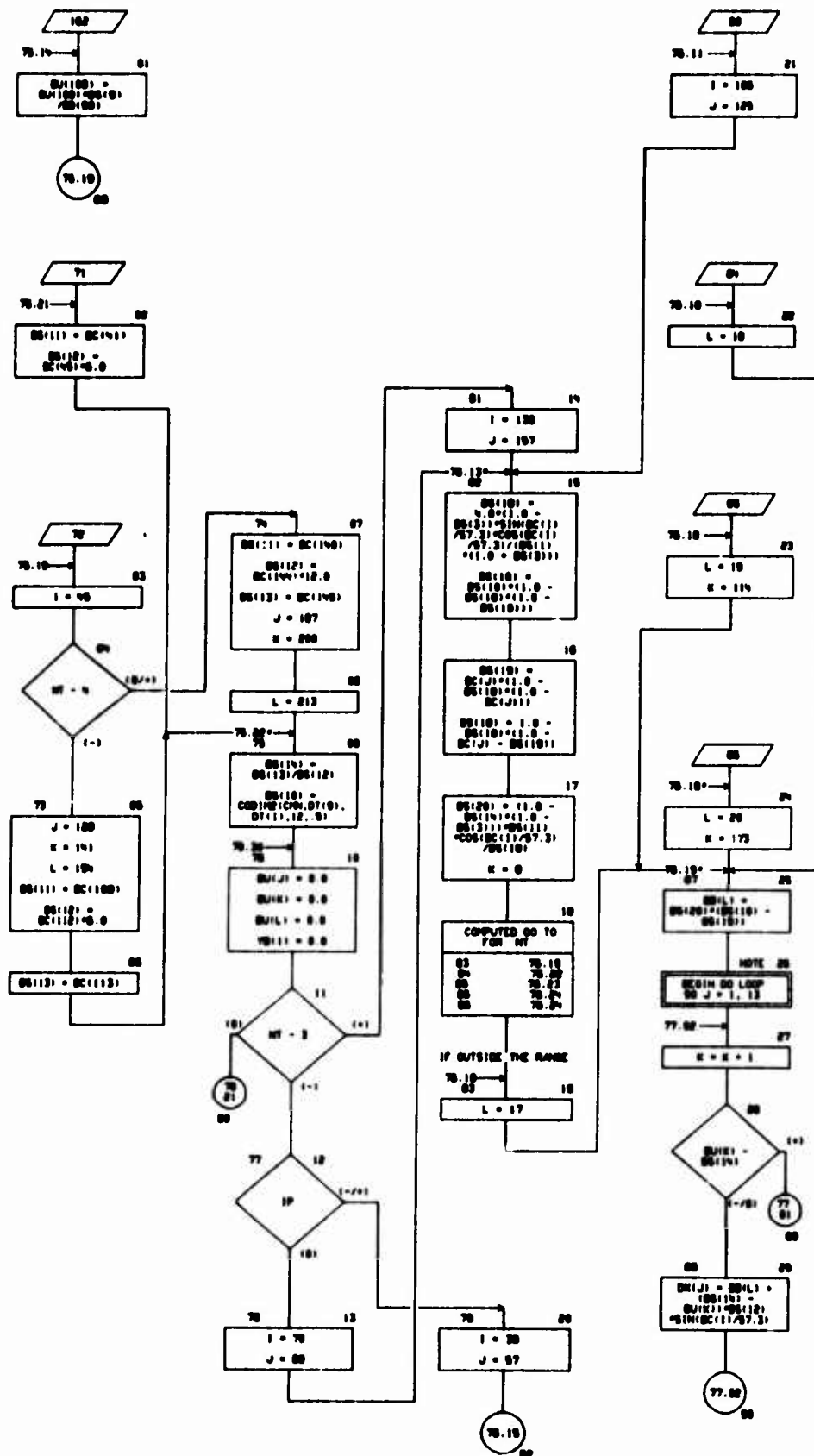


CHART TITLE - SUBROUTINE USPAH

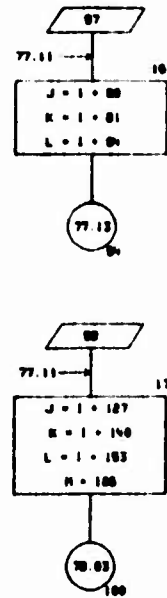
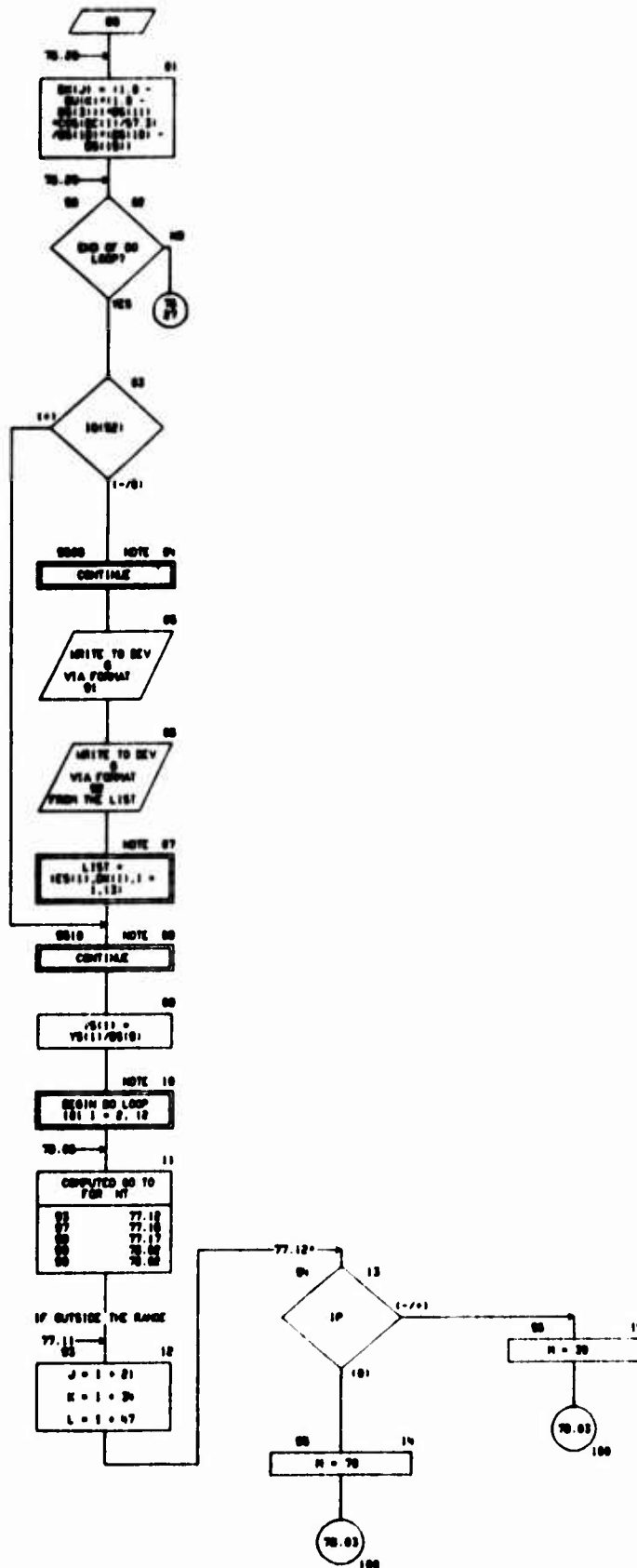


CHART TITLE - SUBROUTINE USPAN

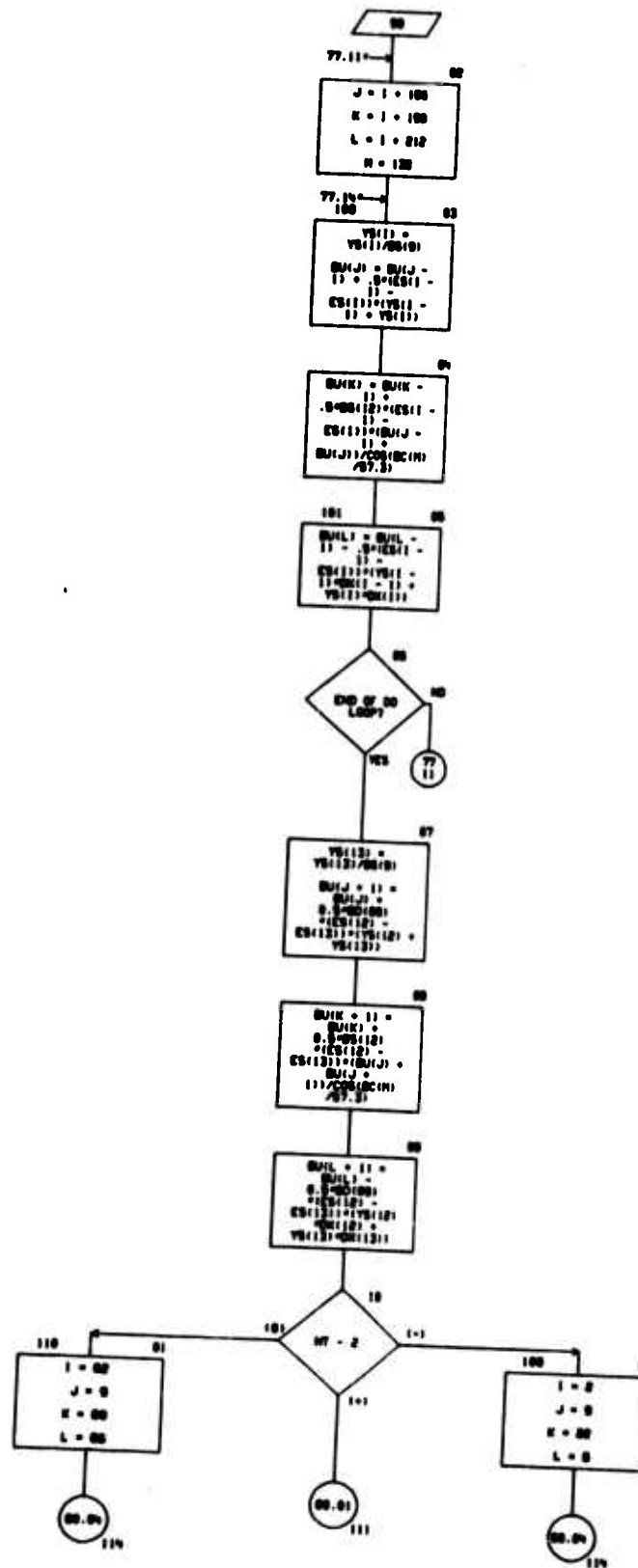


CHART TITLE - SUBROUTINE USPAN

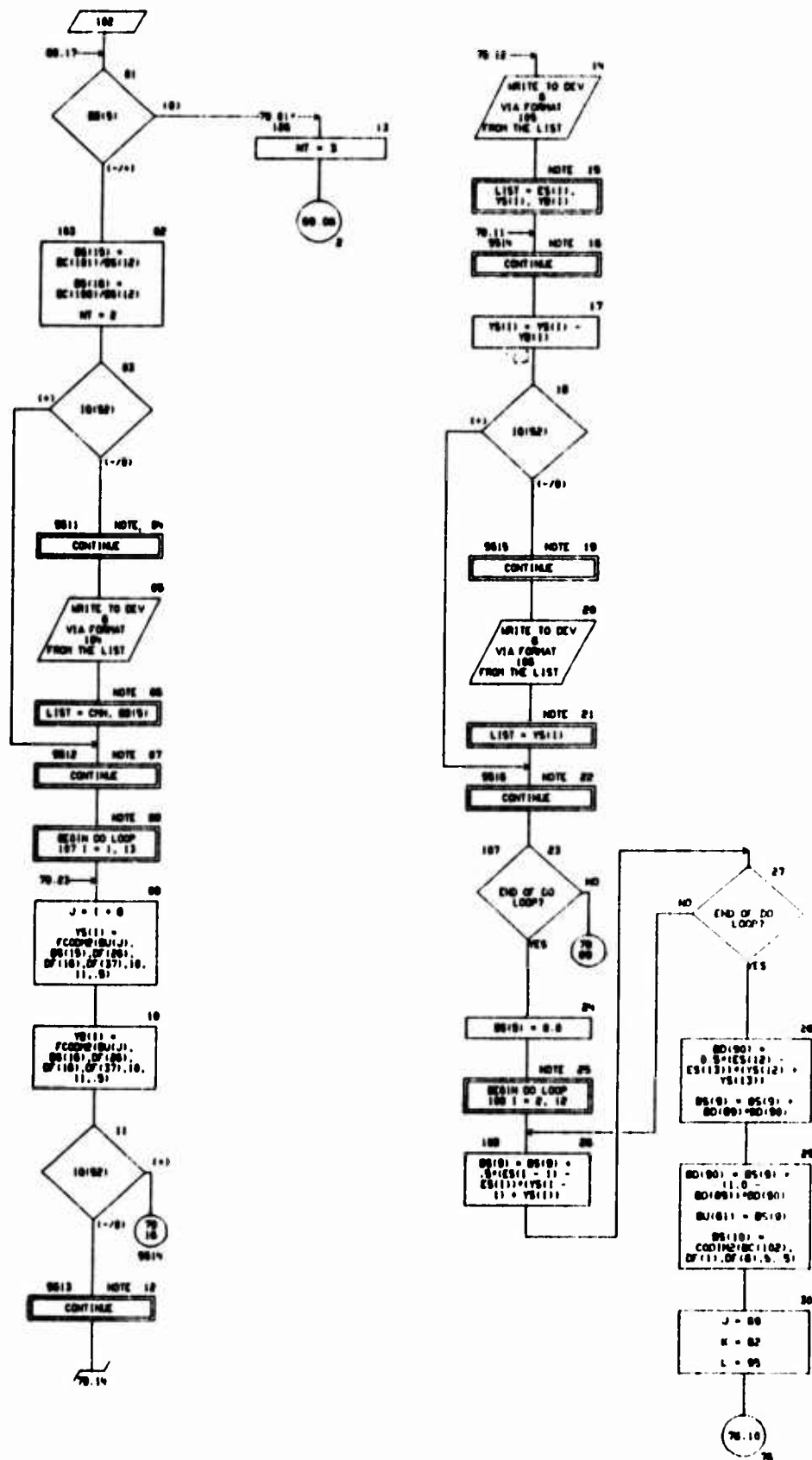


CHART TITLE - SUBROUTINE URPAN

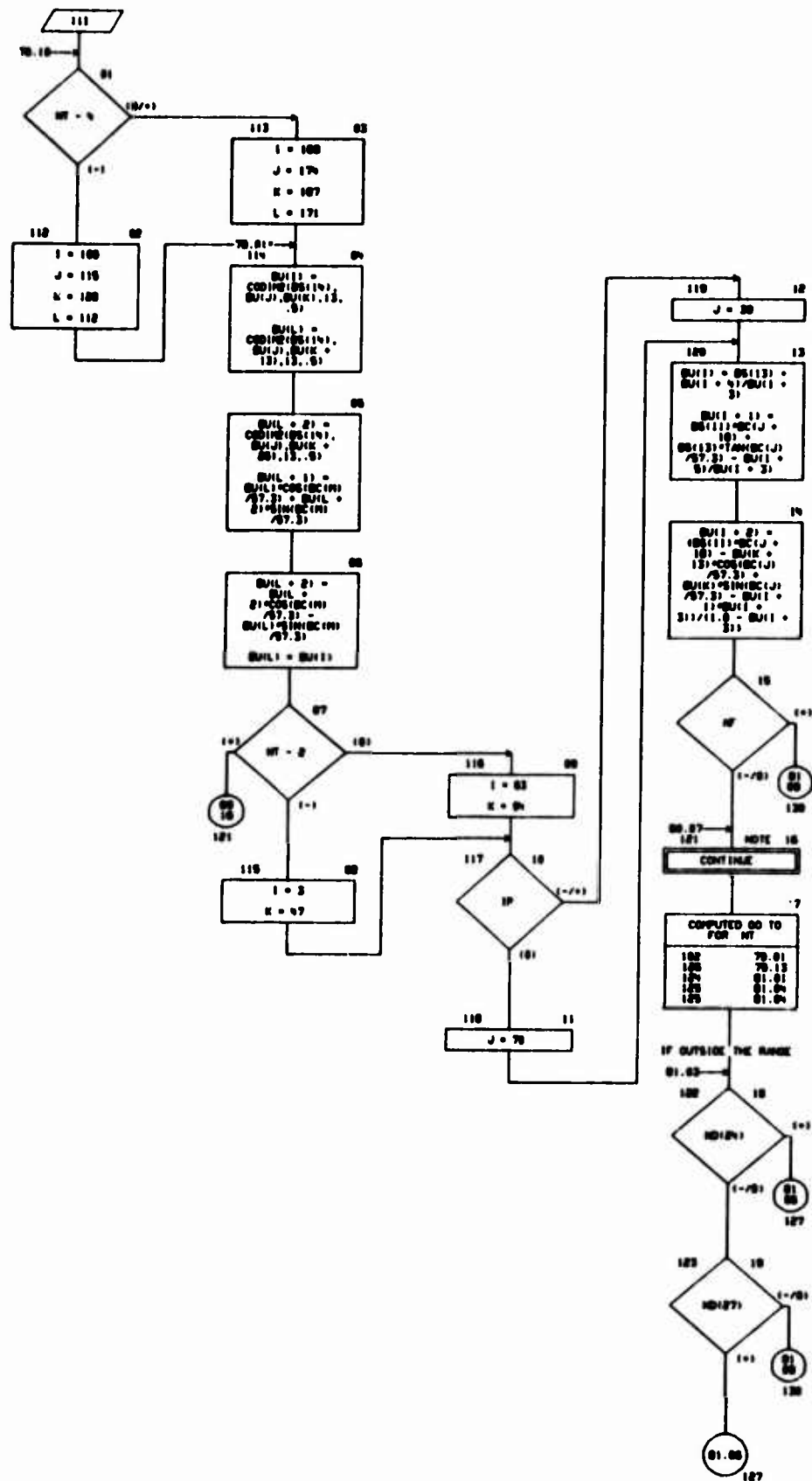


CHART TITLE - SUBROUTINE JAPAN

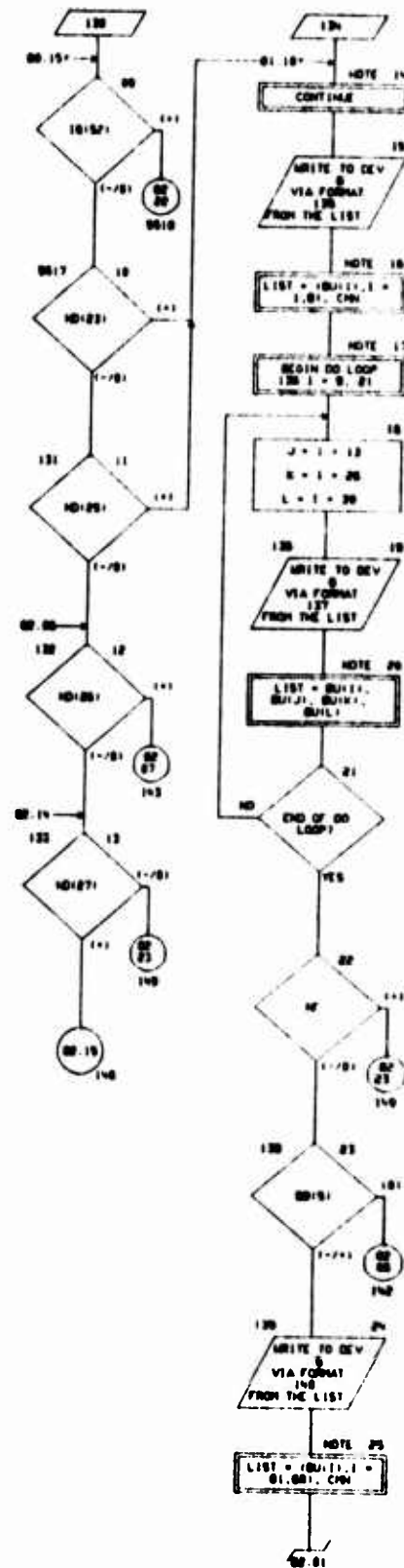
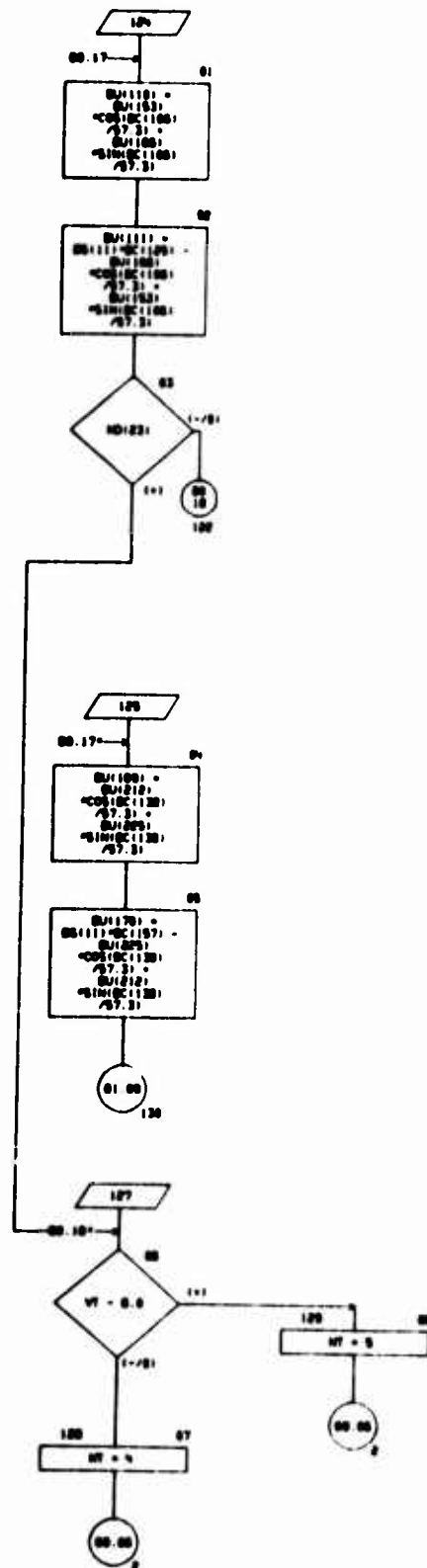


CHART TITLE - SUBROUTINE WSPAN

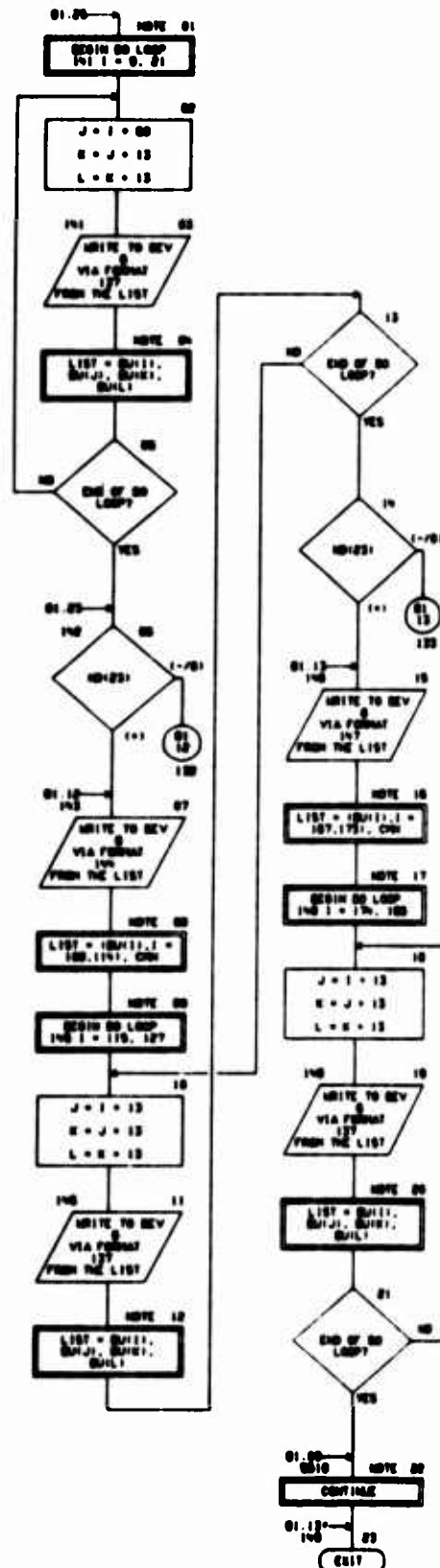


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(4000)
COMMON /PRINT/101001
DIMENSION MD(200),DT(50),DD(50),DF(140),DP(70),DC(100),DE(20),DS
(20),DD(100),DU(200),TA(5),VB(20),VC(5),VD(5),DE(5),DD(5),
EQUIVALENCE (DT(1),TCON(100)),(DD(1),TCON(150)),(DF(1),TCON(1000)),
(DP(1),TCON(1100)),(DD(1),TCON(1200)),(DC(1),TCON(1700)),(DE(1),T
CON(1800)),(DS(1),TCON(1900)),(DD(1),TCON(2000)),(DU(1),TCON(3100)),
TCON(3200),VB(1),DD(140),TA(1),DD(1),VD(1),DD(1),VC(1),DD(2
0),VD(1),DD(51),DD(52),DD(53),DD(54),DD(55),DD(56),DD(57),DD(58),DD
(59),DD(60),DD(61),DD(62),DD(63),DD(64),DD(65),DD(66),DD(67),DD(68),DD
(69),DD(70),DD(71),DD(72),DD(73),DD(74),DD(75),DD(76),DD(77),DD(78),DD
(79),DD(80),DD(81),DD(82),DD(83),DD(84),DD(85),DD(86),DD(87),DD(88),DD
(89),DD(90),DD(91),DD(92),DD(93),DD(94),DD(95),DD(96),DD(97),DD(98),DD
(99),DD(100),DD(101),DD(102),DD(103),DD(104),DD(105),DD(106),DD(107),DD
(108),DD(109),DD(110),DD(111),DD(112),DD(113),DD(114),DD(115),DD(116),DD
(117),DD(118),DD(119),DD(120)
21  FORMAT(10,5X,20H** UPAN - (P102) **
    11X,30HVALUES FROM SLAVE TABLES FOR AP=,F0 4,3X,4HWD=,F0 2,3X,
    3HDB=,F7 4,3X,5HBA=,F0 4)
22  FORMAT(10,7X,3HDB=,F10 2//11X,F10 5)
23  FORMAT(10,5X,5HMIN=)
24  FORMAT(10,5X,10HMINER TAIL)
25  FORMAT(10,5X,10HVERT TAIL)
26  FORMAT(10,10X,30HVALUES FROM LOADING TABLES FOR 100=,F0 3)
27  FORMAT(10,10X,F10 5,3X,4HSTA=,F0 3)
28  FORMAT(10,7X,3HDB=,F0 4,3X,3HDB=,F7 4,3X,3HDB=,F0 4,3X,3HDB=,
    F0 4,3X,20H** UPAN - (P102) **
29  FORMAT(10,7X,3HDB=,F10 2)
30  FORMAT(10,10X,30HSTATION,10X,20HLOADING AT DATA STATIONS,5X,30H=,
    F0 3,2X,20H** UPAN - (P102) **
31  FORMAT(10,F17 5)
32  FORMAT(10,10X,30HSTATION,10X,20HLOADING ANALYSIS STATIONS)
33  FORMAT(10,10X,30HSTATION,10X,50H SHEEP)
34  FORMAT(10,F17 5,F17 3)
35  FORMAT(10,5X,20H** UPAN - (P102) **
    11X,30HSTATION,10X,50HDB LONG,10X,50HMIN LONG,10X,50HAP INCR,
    5X,30H=,F0 3,5X,3HDB=,F0 2)
36  FORMAT(10,F17 5,F10 5,F10 5)
37  FORMAT(10,5X,F10 5)
38  FORMAT(10,10X,10HLOADING PARAMETERS,5X,20H** UPAN - (P102) **
    //7X,4HCLAR,F0 5,4X,30HDB(1)=,F7 5,4X,30HDB(1)=,F7 5,4X,
    30HDB(1)=,F0 2,4X,30HDB(1)=,F0 2//7X,10HDB OF BODY UNITS,//
    7X,4HCLAR,F0 5,4X,30HDB(1)=,F0 3,4X,30HDB(1)=,F0 3//7X,
    20HSPANNING UNIT DISTRIBUTIONS//5X,30HSTA,7X,30HSTA(1),4X,
    30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),5X,30H=,F0 3/5X,30HDB(1),10X,
    30HDB(1),5X,30HDB(1)
39  FORMAT(10,F17 5,F10 5,F10 5)
40  FORMAT(10,5X,20HDB DEFLECTED PARAMETERS,5X,
    20H** UPAN - (P102) **//7X,4HCLAR,F0 5,4X,30HDB(1)=,F7 5,4X,
    30HDB(1)=,F7 5,4X,30HDB(1)=,F0 2,4X,30HDB(1)=,F0 2//7X,
    10HDB OF BODY UNITS//7X,4HCLAR,F0 5,4X,30HDB(1)=,F0 3,4X,
    30HDB(1)=,F0 3//7X,20HSPANNING UNIT DISTRIBUTIONS//5X,30HSTA,7X,
    30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),5X,30H=,F0 3/5X,30HDB(1),10X,
    30HDB(1),5X,30HDB(1)
41  FORMAT(10,5X,10HMINER TAIL PARAMETERS,5X,20H** UPAN - (P102) **
    //7X,4HCLAR,F0 5,4X,30HDB(1)=,F7 5,4X,30HDB(1)=,F7 5,4X,30HDB(1)=,F0 2//7X,
    10HDB OF BODY UNITS//7X,4HCLAR,F0 5,4X,30HDB(1)=,F0 3,4X,
    30HDB(1)=,F0 3//7X,20HSPANNING UNIT DISTRIBUTIONS//5X,30HSTA,7X,
    30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),5X,30H=,F0 3/5X,30HDB(1),10X,
    30HDB(1),5X,30HDB(1)
42  FORMAT(10,5X,10HMINER TAIL PARAMETERS,5X,20H** UPAN - (P102) **
    //7X,4HCLAR,F0 5,4X,30HDB(1)=,F7 5,4X,30HDB(1)=,F7 5,4X,30HDB(1)=,F0 2//7X,
    10HDB OF BODY UNITS//7X,4HCLAR,F0 5,4X,30HDB(1)=,F0 3,4X,
    30HDB(1)=,F0 3//7X,20HSPANNING UNIT DISTRIBUTIONS//5X,30HSTA,7X,
    30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),4X,30HSTA(1),5X,30H=,F0 3/5X,30HDB(1),10X,
    30HDB(1),5X,30HDB(1)

```

CHART TITLE - INTRODUCTORY COMMENTS

.....
SUBROUTINE WMMET
.....

CHART TITLE - SUBROUTINE WAPPE1

WAPPE1

10 00 - n

SUBROUTINE TO ARRANGE
SURFACE AIRLOADS DATA
AND
NORMALIZING FACTORS

**** READ RECORD 37
THE WIND, HORIZONTAL
AND VERTICAL
LOADS ****

01

READS
(1, 01) (1, 00)
(1, 37)

**** READ RECORD 17
THE DEAD WEIGHT
RATIOS ****

(INITIALIZED AT 1.0
IN READ) ****

02

READS
(1, 01) (1, 00)
(1, 17)

LEARN LOADS WHICH ARE
TO BE COMPUTED

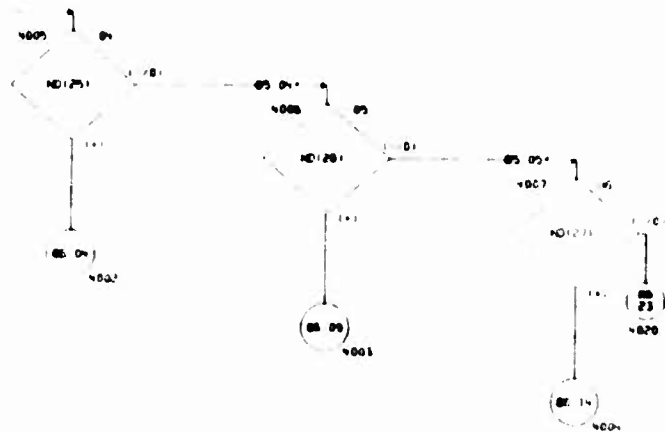
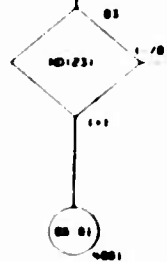
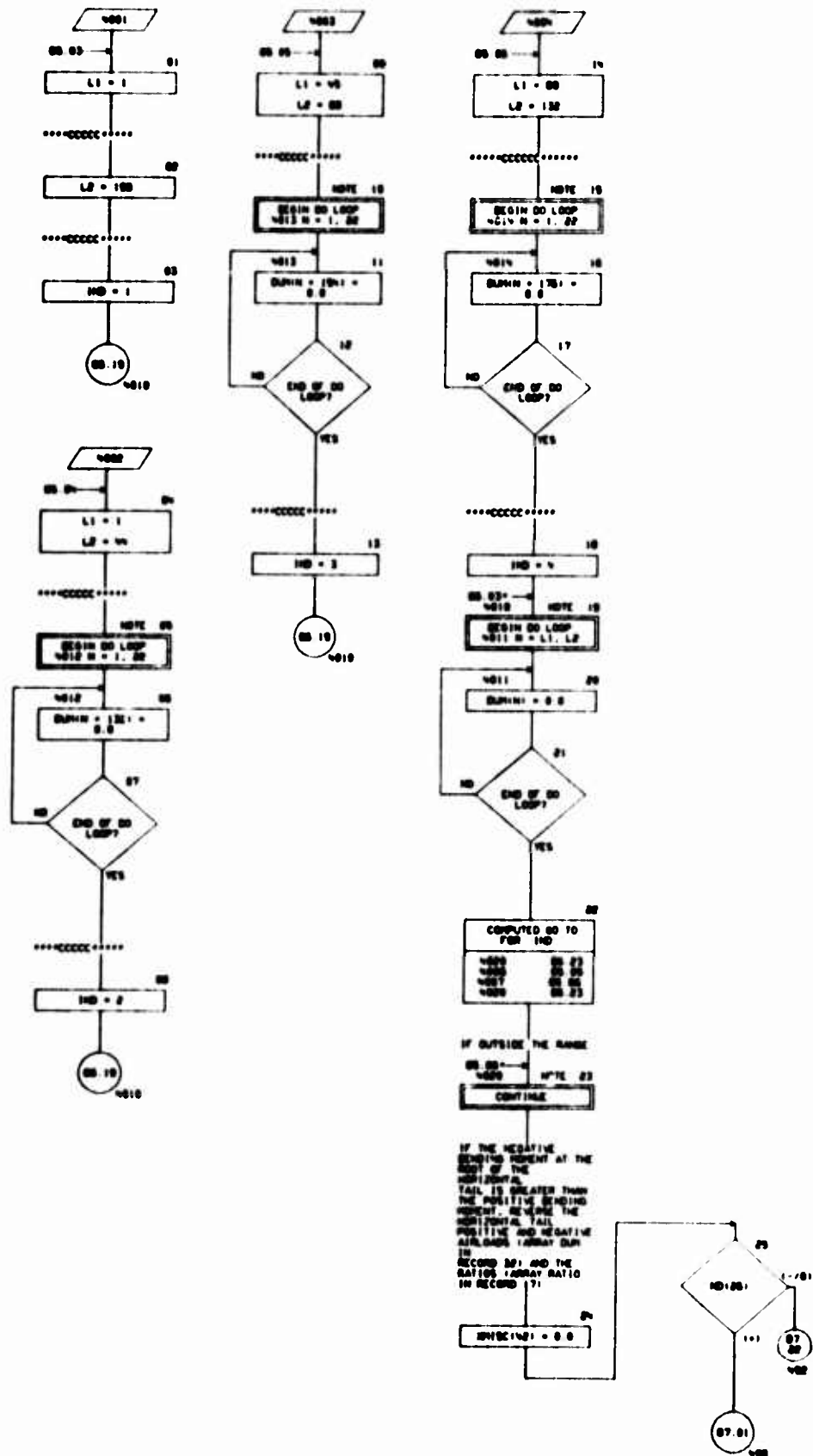


CHART TITLE - SUBROUTINE WANDT



[illegible]

393

CHART TITLE - SUBROUTINE NAME 1

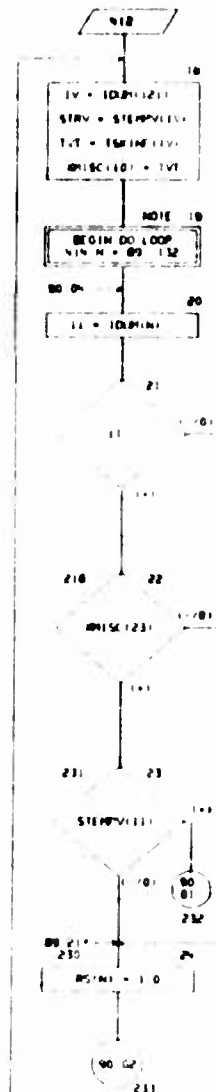
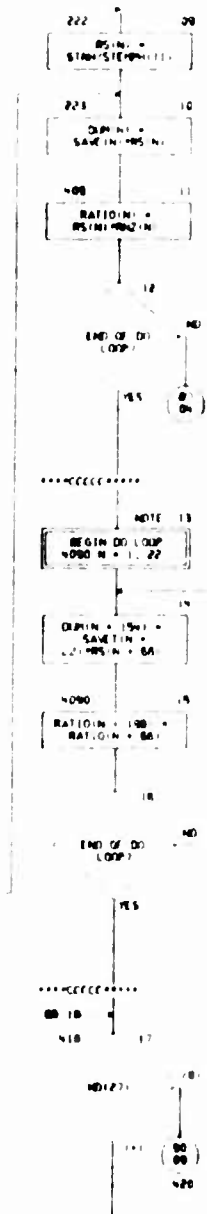


CHART TITLE - SUBROUTINE 100A01

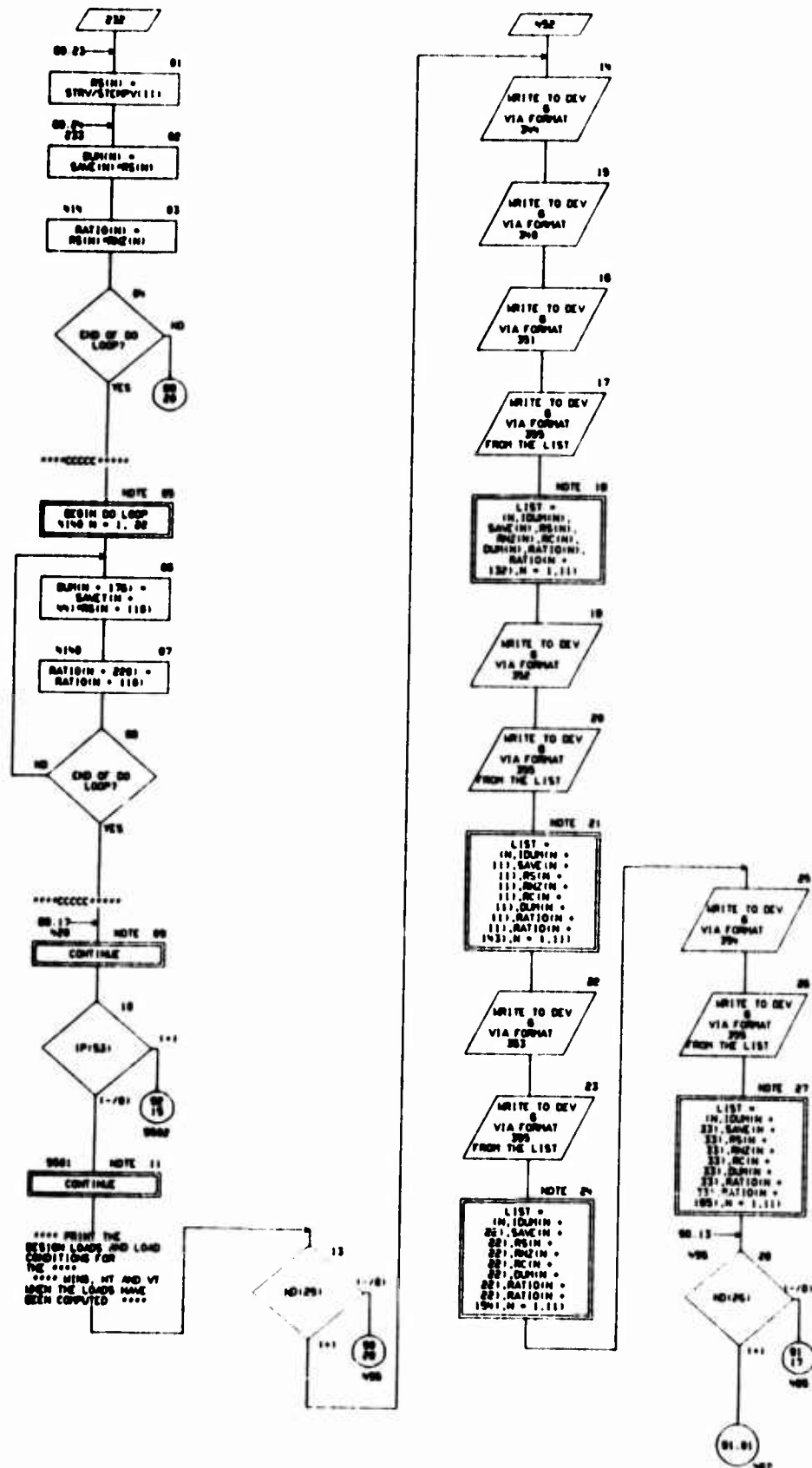


CHART TITLE - SUBROUTINE M0001

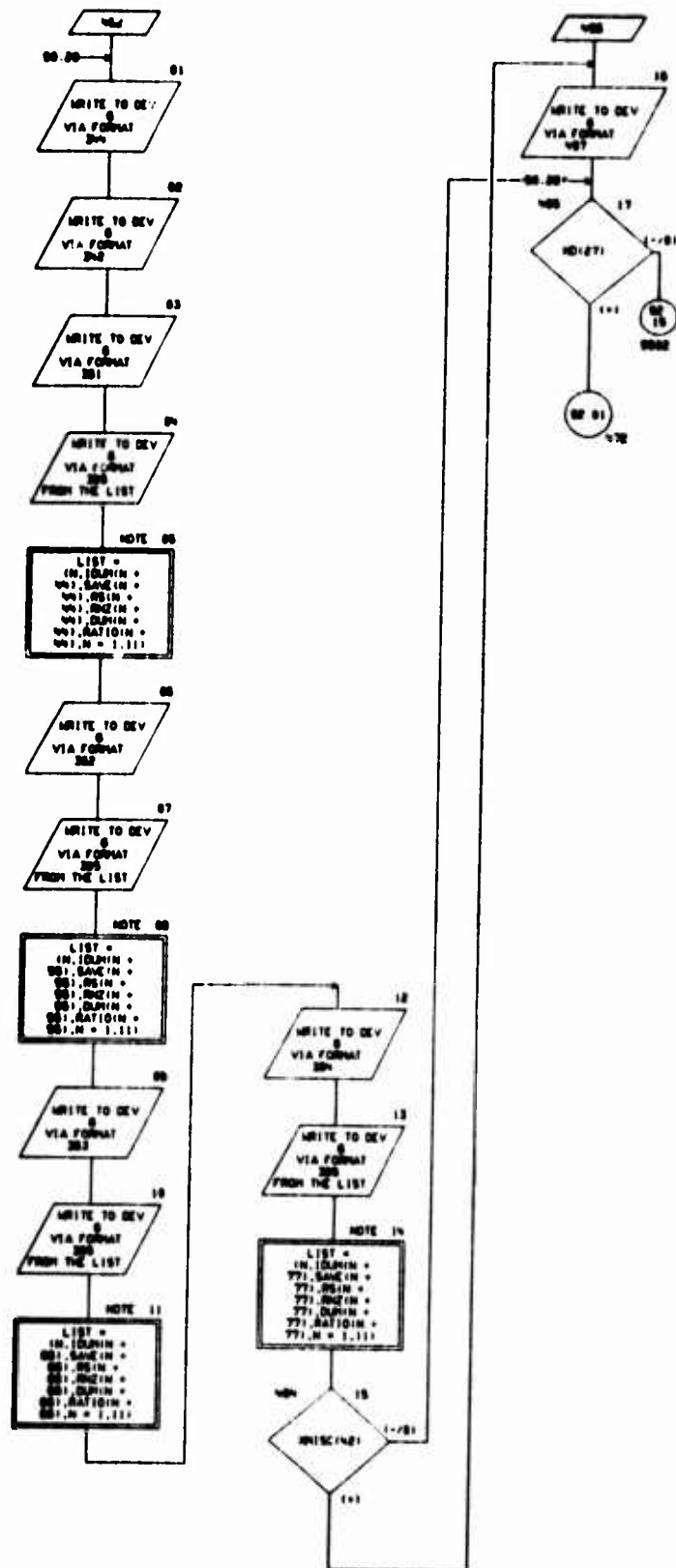


CHART TITLE - SUBROUTINE WMMET

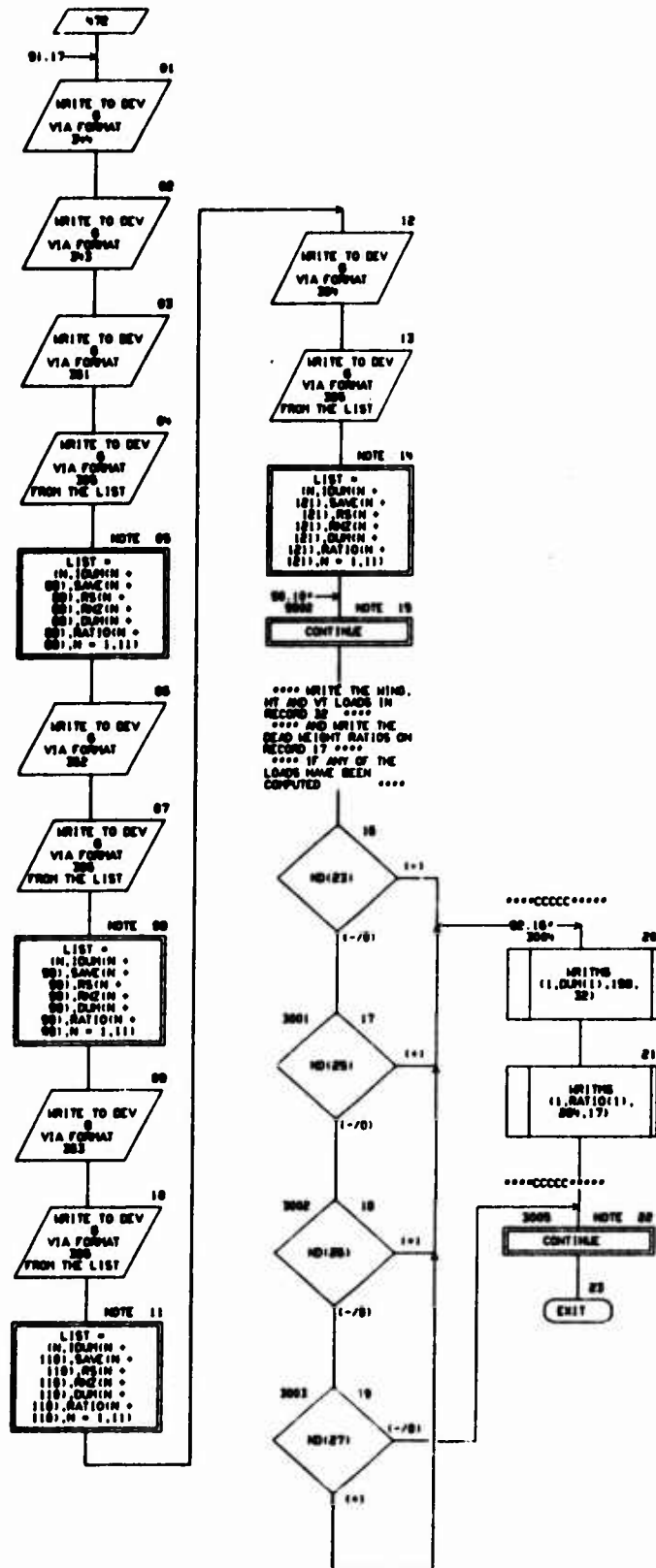


CHART TITLE - NON-PROCEDURAL STATEMENTS

```

COMMON TCON(400),L.D(300),LOPH(132),SAGE(132),JNET(132),
RUE(132),RC(144),SW(100),STEPH(23),STEPV(23),STEPV(23),
LPH(100),SAGE(100)
COMMON /IPRINT/ IP(80)
COMMON /MISC/ MISC(100)
DIMENSION RATIO(204),RS(132)
DIMENSION MD(200)
DIMENSION TSKINF(23)
EQUIVALENCE (MD(1),TCON(420))
EQUIVALENCE (SW(116),TSKINF(1)),(SW(152),SOM)
344 FORMAT(1H,BSX,BM*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD
17) ***BM,BM*** MOMENT - (P(63) ***)
345 FORMAT(4BX,10H* * * M O M E N T * * *)
351 FORMAT(//BSX,BMSTATION LC +V RS RUE RC
RS(1+V) RS+RUE RS+RUE*RC/)
355 FORMAT(4X,12,5X,12,F13.0,F8.3,F8.3,F13.0,F8.3,F8.3)
352 FORMAT(//BSX,BMSTATION LC -V RS RUE RC
RS(1-V) RS+RUE RS+RUE*RC/)
353 FORMAT(//BSX,BMSTATION LC +BM RS RUE RC
RS(1+BM) RS+RUE RS+RUE*RC/)
354 FORMAT(//BSX,BMSTATION LC -BM RS RUE RC
RS(1-BM) RS+RUE RS+RUE*RC/)
342 FORMAT(20H,23H*** HORIZONTAL TAIL ***
351 FORMAT(//BSX,BMSTATION LC +V RS RUE RS(1+
V) RS+RUE/)
355 FORMAT(4X,12,5X,12,F13.0,F8.3,F8.3,F13.0,F8.3)
352 FORMAT(//BSX,BMSTATION LC -V RS RUE RS(1-
V) RS+RUE/)
353 FORMAT(//BSX,BMSTATION LC +BM RS RUE RS(1+
BM) RS+RUE/)
354 FORMAT(//BSX,BMSTATION LC -BM RS RUE RS(1-
BM) RS+RUE/)
407 FORMAT(//17H,72HTHE POSITIVE AND NEGATIVE LOADS ON THE HORIZONTAL T
AIL HAVE BEEN REVERSED//14X,80HBECAUSE THE NEGATIVE BM AT THE ROOT
WAS GREATER THAN THE POSITIVE BM AT THE ROOT)
343 FORMAT(20H,21H*** VERTICAL TAIL ***

```

MODULE FORTRAN LISTING

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
FORTRAN MEDIA (LIST, AUTOGEN)			
CARD NO	****	CONTENTS	****
1		PROGRAM BLCHTL	
2	C		
3	C	PROGRAM BLCHTL IS LINK 4 OF SHEEP.	BLCH0001
4	C		
5	C	DETERMINES BASIC AIRLOAD CONDITIONS TO BE COMPUTED ON/AND WIND	BLCH0002
6	C	SENDING RECENT FATIGUE SPECTRA	BLCH0003
7	C	PROVIDES LOGIC AND CONTROL FOR THE AIRLOAD SUBROUTINES	BLCH0004
8	C		
9		CONVEN TCON(400),HL(100),LUM(100),SAME(100),HNET(100),	
10		• RE(100),RC(40),SW(100),STEPW(20),STEPH(20),STEPPV(20),	
11		• BLN(100)	
12		• SAME(100)	
13		C*****C*****	
14	C		
15		CONVEN /IPRINT/I(100)	
16		CONVEN /MISC/ MISC(100)	
17	C		
18		DIMENSION MD(200),BD(20),BC(100),BD(20),DT(50),DD(100),DF(140),DP(100)	BLCH0010
19		(170)	BLCH0011
20	C		
21		DIMENSION PS(20),LOCAL(20),TTOTAL(20),WFLUX(20),TSKINH(20)	
22	C		
23		DIMENSION IMACH(20),ALT(20),TSKINH(20)	
24	C		
25		EQUIVALENCE (MD(1),TCON(400)),(BD(1),TCON(300)),(BC(1),TCON(2700)),(BLCH0020	
26		01),((BD(1),TCON(200))),((1,MD(100))),	
27		2 (IN,MD(100)),IN,MD(100),IN,MD(100),DT(1),TCON(100),((BD(1)	
28		3),TCON(100),DT(1),TCON(100),DT(1),TCON(100))	BLCH0023
29	C		
30		EQUIVALENCE (ALT(1),BD(20)),(IMACH(1),BD(20))	
31	C		
32		EQUIVALENCE (SW(100),TSKINH(100))	
33		EQUIVALENCE (SW(1),PS(1)),(SW(2),LOCAL(1)),	
34		• (SW(4),TTOTAL(1),SW(70),WFLUX(1),SW(83),TSKINH(1))	
35	C		
36		RECHIND 24	
37	C		
38		C*****C*****	
39		BUFFER IN(24),TCON(1),TCON(200))	
40		C*****C*****	
41	C		
42		IF(UNIT(24)=1010,1010,1010	
43	C		
44	C	**** CLEAR ARRAYS ****	
45	C		
46		DO 3 I=1,20	
47		IMACH(I)=0.0	
48		ALT(I)=0.0	
49		STEPW(I)=0.0	
50		STEPH(I)=0.0	
51		700 STEPPV(I)=0.0	
52	C		
53		DO 3 I=1,20	BLCH0030
54		3 BD(I)=0.0	BLCH0035
55	C		
56		DO 1 I=1,100	
57		LUM(I)=0	
58		RE(I)=1.0	
59		SAME(I)=0.0	
60		1 HNET(I)=0.0	
61	C		
62		DO 4 I=1,40	
63		4 RE(I)=1.0	
64	C	**** READ RECORDS 1-4, LOADS PERMANENT DATA TABLES ****	
65		CALL READP(1,DT(1),50,1)	
66		CALL READP(1,DD(1),50,2)	
67		CALL READP(1,DF(1),140,3)	
68		CALL READP(1,DP(1),70,4)	
69	C		
70	C	*** SET UP MD(13-30), THE LOADS CONTROL INDICATORS ***	

07-03-74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLADDS OVERLAY
CARD NO	*****	CONTENTS	*****
71	C		
72	ND(13) = ND(BC(19))		
73	ND(14) = ND(BC(19))		
74	ND(15) = ND(BC(19))		
75	ND(16) = ND(BC(19))		
76	DO 9 N=23,30		
77	9 ND(14) = ND(BC(19))		
78	C		
79	IF(ND(23)=97,97,98)		
80	80 ND(24) = 1		
81	ND(25) = 1		
82	ND(26) = 1		
83	ND(27) = 1		
84	87 CONTINUE		
85	C		
86	ND(30) = ND(BC(100))		
87	C		
88	C **** READ RECORD 27, THE LOADS VARIABLE DATA ****		
89	CALL READR(1,BC(1),100,27)		
90	C		
91	C **** READ RECORD 31, THE STRESS, TEMPERATURE, WF AND G DATA ****		
92	CALL READR(1,97(1),100,31)		
93	C		
94	C **** READ RECORD 18, THE DEAD WEIGHT DATA (FROM OFATMS) ****		
95	CALL READR(1,MD(1),300,18)		
96	C		
97	DO 15 N=14,300,11		
98	IF(ABS(MD(N)) - 1.0) 15,15		
99	15 MD(N) = 0.0		
100	15 CONTINUE		
101	C		
102	IF(ND(16)=30,124,30)		ELCH227
103	30 DO 21 I=100,131		ELCH224
104	21 ND(11)=0		ELCH225
105	C		
106	IN=0		ELCH220
107	22 DO 67 I=1,8		ELCH223
108	IF(ND(1+27)=67,67,23)		ELCH225
109	23 IF(11+24,24,25)		ELCH224
110	24 IN=1		ELCH225
111	25 DO 10 (26,35,36,40,41,52,61,64,65),1		ELCH224
112	35 IF(ND(14)=20,20,27)		ELCH225
113	27 ND(11)=1		ELCH226
114	36 ND(10)=1		ELCH225
115	IF(BC(167)-BC(100)=30,20,29)		ELCH227
116	29 ND(11)=1		ELCH227
117	60 TO 31		ELCH220
118	30 ND(11)=1		ELCH225
119	31 IF(BC(100) - 1.0) 67,33,33		ELCH220
120	33 ND(11)=1		ELCH220
121	60 TO 67		ELCH220
122	35 IF(ND(14)=37,37,38)		ELCH224
123	38 ND(11)=1		ELCH225
124	37 ND(14)=1		ELCH220
125	60 TO 67		ELCH224
126	39 ND(11)=1		ELCH225
127	60 TO 67		ELCH220
128	40 ND(11)=1		ELCH220
129	60 TO 67		ELCH220
130	41 IF(ND(14)=44,44,43)		ELCH225
131	43 ND(11)=1		ELCH220
132	ND(12)=1		ELCH220
133	44 ND(11)=1		ELCH220
134	ND(11)=1		ELCH220
135	60 TO 67		ELCH220
136	52 IF(ND(14)=55,55,54)		ELCH220
137	54 ND(11)=1		ELCH220
138	ND(12)=1		ELCH220
139	55 ND(11)=1		ELCH220
140	ND(12)=1		ELCH220
141	60 TO 67		ELCH220

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07/00/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOAD OVERLAY
CARD NO	*****	CONTENTS	*****
004	GO TO 102		ELCH1200
005	110 MI=0		ELCH1200
006	GO TO 102		ELCH1270
007	100 MI=7		ELCH1275
008	GO TO 102		ELCH1280
009	101 MI=0		ELCH1285
010	C		
011	C **** COMPUTE THE LOADS FOR THIS CONDITION	****	
012	C **** THE WIND SHEAR IS STORED IN 00 30.30.40. .70	****	
013	C **** THE WIND DIR IS STORED IN 00 33.37.41. .75	****	
014	C **** THE WT SHEAR IS STORED IN 00 05.00.03. .105	****	
015	C **** THE WT DIR IS STORED IN 00 05.00.04. .105	****	
016	C **** THE VT SHEAR IS STORED IN 00 130.142.146. .170	****	
017	C **** THE VT DIR IS STORED IN 00 130.143.147. .170	****	
018	C		
019	102 CALL 04005		ELCH1290
020	CALL 04004		ELCH1290
021	C		
022	ALTI(1) = 00101		
023	WINDCH(1) = 001101		
024	C		
025	C		
026	C DETERMINE MAXIMUM MET DIR AT 005 AND AT STATION 2		
027	C WHEN THE WIND LOADS HAVE BEEN COMPUTED		
028	C STORE IN WINDC 30 AND 33		
029	C SAVE THE AIRLOAD DIR, CONDITION INDICATOR, LOAD FACTOR		
030	C AND DEAD WEIGHT PERCENT		
031	C		
032	WINDC(00114,14,405)		
033	C		
034	405 00 7010.0.0.0.7.0.7.0.0.0.7.7.0.0.7.7.0.0.0.0.0.1		
035	C		
036	0 WINDC(1) = WINDC(43)		
037	WINDC(1) = WINDC(44)		
038	GO TO 10		
039	C		
040	7 WINDC(1) = WINDC(40)		
041	WINDC(1) = WINDC(40)		
042	GO TO 10		
043	C		
044	0 WINDC(1) = WINDC(45)		
045	WINDC(1) = WINDC(45)		
046	GO TO 10		
047	C		
048	0 WINDC(1) = WINDC(47)		
049	WINDC(1) = WINDC(48)		
050	C		
051	10 WINDC(1) = 001251 + WINDC(1) * 001101		
052	WINDC(2) = 001001 + WINDC(1) * 001101		
053	C		
054	IF(WINDC(32) - ABS(WINDC(1)+11,12,12		
055	C		
056	11 WINDC(32) = WINDC(1)		
057	DIR1 = 001251		
058	DIR1 = 1		
059	DIR1 = 001101		
060	WINDC(1) = WINDC(1)		
061	C		
062	12 IF(WINDC(33) - ABS(WINDC(2)+13,14,14		
063	C		
064	13 WINDC(33) = WINDC(2)		
065	DIR2 = 001001		
066	DIR2 = 1		
067	DIR2 = 001101		
068	WINDC(2) = WINDC(2)		
069	14 CONTINUE		
070	C		
071	CALL PAR005		
072	C		
073	WINDC(1)=1125,125,010		
074	C		

AFTERNOON

07/05/74

INPUT LISTING

AUTOFLOW CHART SET - SHEEP

AIRLOADS OVERLAY

CARD NO

CONTENTS

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007      TWD=010.000                                A77000+0
008      SD=20 170+0                                A7700000
009      SD=1710.4007                                A7700000
010      RE=00000001                                A7700070
011      IF IN=20152.120.40.40                      00000100
012      SD=0 0                                         00000100
013      TWD=010.000                                000001+0
014      TWD=- 00700010                              00000100
015      PAGE=14.00000                               00000100
016      SD TO 100                                    00000100
017      40 IF IN=20730.140.00.00                    00000170
018      40 SD=20000.7+                              00000100
019      TWD=200.000                                  00000100
020      PAGE=3.000+71                                00000100
021      SD TO 110                                    00000200
022      50 IF IN=10030.100.00.00                    00000210
023      50 SD=00001.                                00000000
024      TWD=200.000                                  00000230
025      TWD=- 0010+002                              000002+0
026      PAGE=- 2000+03                              000002+0
027      SD TO 100                                    00000200
028      60 IF IN=17030.0+00.70.70                  00000200
029      60 SD=10+100.0                               00000270
030      .TWD=000 700                                00000200
031      PAGE=- 017+00+0                              00000000
032      SD TO 110                                    00000200
033      70 IF IN=2+0000 0.70.00.00                  00000200
034      70 SD=1700+0.0                               00000310
035      TWD=000.700                                  00000200
036      TWD=-. 000120000                             00000330
037      PAGE=- 000+0001                              00000100
038      SD TO 100                                    000002+0
039      80 SD=2+0001.                                00000200
040      TWD=20+0.3+0                                 00000200
041      PAGE= 0002000.000                            00000200
042      SD TO 110                                    00000270
043      100 SD=RE 40/(RE 40)                          00000200
044      TWD=TWD+TWD*(SD-00)                          00000200
045      VP=- (00/(00+TWD))+(AL00/(TWD/TWD))          00000+00
046      P=PAGE 40/VP)                                00000+10
047      SD=0. 001177+0001/TWD)                     00000+20
048      SD=0+000+0+TWD/(P0+TWD)                    00000+30
049      SD TO 120                                    00000+40
050      110 SD=RE 40/(AL 40)                          00000+00
051      TWD=TWD                                00000+00
052      VP=-00/(00-00)/(00+TWD)                     00000+70
053      P=PAGE 40/VP)                                00000+00
054      SD=0. 001177+0001/TWD)                     00000+00
055      SD=0+000+0+TWD/(P0+TWD)                    00000000
056      120 RETURN
057      END
058
059      C
060      C
061      C SUBROUTINE DLDS
062      C
063      SUBROUTINE DLDS                                27000001
064      C SUBROUTINE DLDS FOR LINK S OF SHEEP.          27000002
065      C COMPUTES COMPONENT TOTAL AIRLOADS AND CP'S. AND INERTIA FACTORS. 27000003
066      COMMON TCDH(400)                                27000004
067      COMMON /IPRINT/IO(00)
068      DIMENSION DD(20),DD(00),BC(100),DS(20),DU(20),DC(20),DF(140),ND(200000)
069      11200)
070      27000000
071      EQUIVALENCE (DF(1),TCDH(1000)),(DD(1),TCDH(200)),(BC(1),TCDH(27002000010
072      1), (DU(1),TCDH(310)), (DS(1),TCDH(207)), (DD(1),TCDH(330)), (DD(1),27000011
073      2,TCDH(200)), (ND(1),TCDH(400)), (IN),ND(100)), (IN),ND(150)), (1,ND(1532000012
074      31), (J,ND(102)), (K,ND(103)), (IP,ND(107))
075      27000013
076      CALL A77000(0),DS(1),DS(2),DS(3)
077      27000000
078      DS(1)=DS(1)+DS(3)+0*(DD(10)+0)
079      27000000
080      DS(1)=SQR(200.0 + DS(1))
081      27000100
082      DD(1)=BC(3)+BC(3)+BC(3)+1/3.1416*(BC(3)+DS(1)+0)
083      27000120

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07-03-74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLADDS OVERLAY
CARD NO	****	CONTENTS	****
000	MT=0		20000120
000	3 MT=MT+1		20000140
070	00 TO 14,0,0,101,MT		20000100
071	4 1=0		20000100
072	J=4		20000170
073	00 TO 0		20000100
074	0 1=10		20000100
075	J=0		20000200
076	0 IF(1P)0,7,0		20000210
077	7 0=71		20000200
078	00 TO 11		20000230
079	0 0=0		20000240
080	00 TO 11		20000250
081	0 1=13		20000300
082	J=11		20000270
083	0=107		20000300
084	00 TO 11		20000250
085	10 1=17		20000200
086	J=170		20000310
087	0=130		20000250
088	11 00(11)=0C(11)+0U(11)		20000320
089	IF(107)-313,3,12		20000240
090	12 IF(00(5)+113,14,13		20000350
091	13 0 = 00		20000250
092	IF(1P) 01,00,01		20000250
093	00 0 = 71		20000250
094	01 00(0) = 0C(0) + 0U(0)		20000200
095	00(7) = 0C(7) + 0U(00)		20000270
096	00 TO 10		20000200
097	14 00(0)=0.0		20000200
098	00(7)=0.0		20000200
099	15 IF(1P)17,10,17		20000210
000	16 1=70		20000250
001	00 TO 10		20000250
002	17 1=0		20000240
003	18 00(13)=0.0		20000250
004	00(10)=00(0)		20000200
005	00 TO 110,20,20,10,10,10,21,101,01		20000270
006	19 00(0)=1.1+00(11)+00(10)/00(4)+0C(11)		20000200
007	0U TO 20		20000200
008	20 00(0)=00(10)+0C(10)+07(11)+07(11)+0.5)+0U(11)+0U(0)+00(5)/07 3		20000200
009	00(0)=00(0)+1.1+00(11)+00(10)/00(4)+0C(11)		20000210
010	00 TO 20		20000250
011	21 00(13)= (00(7)+00(13)+12.0)/0U(11)+0C(107)+00(21)		20000230
012	00(10)+00(10)+00(13)/00(11)		20000240
013	00 TO 10		20000250
014	22 00(11)+00(0)/0U(1)		20000200
015	00(11)= 0+2030+00(11)+00(4)+0C(10)+02		20000270
016	00(21)+0U(21)+00(0)+00(4)+0C(11)		20000200
017	00(31)+00(21)+1.0-0U(21)/0U(2)		20000200
018	00(4)=1+00(21)+00(0)+00(11)+00(10)+00(21)+00(31)+00(0)+00(21)+00(23)+00(00)		20000200
019	111/00(13)+00(21)		20000210
020	00(0)=0.0		20000250
021	00(0)=0.0		20000230
022	IF(00(0)+23,24,23		20000240
023	23 00(0)+0U(02)+00(0)+00(4)+0C(11)		20000200
024	00(0)+00(0)+1.0-0U(02)/0U(02)		20000200
025	00(4)+00(4)+1+00(0)+00(21)+00(0)+00(17)+00(21)+00(0)+1/00(13)+00(23)+00(00)		20000200
026	111		20000200
027	24 00(12)+00(11)+00(21)+00(31)+00(4)+00(0)+00(0)		20000200
028	IF(00(12)) 53,52,53		20000200
029	00 00(10) = 1.0		20000200
030	00 TO 04		20000200
031	03 00(10) = 00(11)+00(10)/00(12)		20000200
032	04 0U(000) = 00(21)+00(10)		20000200
033	0U(007) = 00(0)+00(10)		20000210
034	0U(000) = 00(13) + 00(4)+00(10)		20000200
035	00(3)+00(11)+00(11)+00(10)/00(12)		20000230
036	00(0)+0.0+0U(000)+0U(007)		20000240
037	00(0) = (00(3) + 00(0)+00(10)		20000250
038	00(11) = 0.0+00(13) + 00(4)+00(10)		20000200

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409

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	APPLDADS OVERLAY
CARD NO	****	CONTENTS	****
710	I F7.2,NX,WRITE(1)=,F8.2//		32001300
711	WRITE(6,N)END(1),END(1)		32001310
712	N6 FORMATTING,BX,2000000 CARRY-OVER LOAD/7X,DPZ(1)=,F8.0,NX,		32001320
713	I WRITE(1)=,F8.2//		32001330
714	WRITE(6,N)END(1),END(1),END(1),END(1),END(1)		32001340
715	N1 FORMATTING,BX,21000000000 TAIL LOADS/7X,DPZ(2)=,F8.0,NX,WRITE(=,32001350		
716	I F7.2,NX,WRITE(=,F8.2,NX,END(1)=,F8.0//		32001360
717	WRITE(6,N)END(1),END(1),END(1)		32001370
718	N2 FORMATTING,BX,20000000000 TAIL LOAD/7X,DPZ(3)=,F8.0,NX,WRITE(=,F8.2,32001380		
719	I NX,WRITE(=,F8.2//		32001390
720	WRITE(6,N)END(1),END(1),END(1),END(1),END(1)		32001400
721	N3 FORMATTING,BX,20000000000 INERTIA FACTORS/7X,DOZ(=,F8.2,NX,DOY(=,32001410		
722	I F7.2,NX,DOZ(=,F7.3,NX,DOZ(=,F7.3//		32001420
723	WRITE(6,N)END(1),END(1),END(1)		32001430
724	N4 FORMATTING,BX,20000000000 SPACIAL FACTORS/7X,DPZ(1)=,F8.0,NX,32001440		
725	I DPZ(1)=,F8.0,NX,DPZ(1)=,F8.0//		32001450
726	END CONTINUE		
727	N6 RETURN		32001460
728	END		32001470
729	C		
730	C !!!		
731	C FUNCTION CODING		
732	C !!!		
733	C		
734	FUNCTION CODING(X,XI,YI,M,N)		COD10010
735	COMMON TCN(400)		COD10015
736	DIMENSION ND(200),XI(1),YI(1)		COD10020
737	EQUIVALENCE (ND(1),TCN(400)),(M,ND(143)),(J,ND(144)),(JJ,ND(145))COD10025		
738	(I),(K,ND(147)),(L,ND(147)),(N,ND(148))		COD10030
739	M = X		COD10037
740	N1 = N		COD10040
741	IF (M-2) 100,110,100		COD10039
742	100 CODING = YI(M)		COD10036
743	GO TO 200		COD10040
744	110 N1 = 2		COD10045
745	140 CODING = (YI(N1)-YI(N1-1))/(XI(N1)-XI(N1-1)) + M-XI(N1-1) *		COD10046
746	I YI(N1-1)		COD10050
747	GO TO 200		COD10050
748	150 IF (XI(1) - XI(N1))100,100,170		COD10055
749	160 DO 170 I = 1,N1		COD10070
750	J = I		COD10072
751	IF (M - XI(J)) 100,200,170		COD10075
752	170 CONTINUE		COD10080
753	GO TO 100		COD10085
754	170 DO 180 I = 1,N1		COD10090
755	J = I		COD10092
756	IF (XI(J) - M) 100,200,100		COD10095
757	180 CONTINUE		COD10100
758	180 N1 = N		COD10105
759	GO TO 140		COD10110
760	190 IF (J - 2) 110,210,200		COD10115
761	210 J = 3		COD10120
762	JJ = I		COD10125
763	GO TO 200		COD10130
764	200 CODING = YI(J)		COD10135
765	GO TO 200		COD10140
766	200 IF (J - N1) 200,200,100		COD10145
767	200 J = N1-1		COD10150
768	JJ = 2		COD10155
769	GO TO 200		COD10160
770	200 JJ = 3		COD10165
771	200 IF(N1-3)200,200,200		COD10170
772	200 J = 3		COD10175
773	200 K = J-1		COD10180
774	K = K-1		COD10185
775	L = J+1		COD10190
776	XI(K) = XI(M)		COD10195
777	XI(K) = XI(K)		COD10200
778	XI(J) = XI(J)		COD10205
779	AJ = M-XI(K)		COD10210
780	AD = M-XI(K)		COD10215

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07/03/74	INPUT LISTING	AUTOFLIGHT CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
002	GO TO 10		FATH0100
003	9 I=240		FATH0105
004	10 DO 11 J=1,40		FATH0110
005	K=J+1		FATH0115
006	11 DO 12 I=1,40		FATH0120
007	DO 12 J=000,007		FATH0125
008	DO 13 I=0C(107)+0D(1+415)		FATH0130
009	12 DO 14 I=0C(100)+0D(1+07)		FATH0135
010	I=0		FATH0140
011	GO TO 117,17,13,14,10,101,NA		FATH0145
012	13 I=100		FATH0150
013	GO TO 17		FATH0155
014	14 I=140		FATH0160
015	GO TO 17		FATH0165
016	15 I=020		FATH0170
017	GO TO 17		FATH0175
018	16 I=000		FATH0180
019	17 DO 20 J=1,20		FATH0185
020	K=J+1		FATH0190
021	DO 21 J=570+0C(K)		FATH0195
022	IF (J-10)10,10,10		FATH0200
023	10 DO 22 J=000+0C(K)-1.0		FATH0205
024	GO TO 20		FATH0210
025	10 DO 23 J=000+0C(K)+1.0		FATH0215
026	20 CONTINUE		FATH0220
027	MS=0		FATH0225
028	MS=1		FATH0230
029	MS=1		FATH0235
030	MS=2		FATH0240
031	DO 34 I=1,0		FATH0245
032	MS=MS+1		FATH0250
033	00(1)=00(1)+000		FATH0255
034	00(9)=00(1)+120		FATH0260
035	00(10)=00(1)+120		FATH0265
036	00(12)=00(1)+130		FATH0270
037	IF (00(12)-2.0)20,21,22		FATH0275
038	21 J=77		FATH0280
039	K=101		FATH0285
040	GO TO 23		FATH0290
041	22 J=0		FATH0295
042	K=170		FATH0300
043	23 IF (MS-2)24,20,20		FATH0305
044	24 CALL UEPAN		FATH0310
045	05(1)=0C(171)/05(12)		FATH0315
046	05(3)=0J(7)		FATH0320
047	05(4)=CODIMP(05(1),0J(0),0J(3),13,5)		FATH0325
048	IF (MS - 2)25,27,27		FATH0330
049	25 00(10)=05(3)		FATH0335
050	00(20)=05(4)		FATH0340
051	27 05(5)=CODIMP(00(1)+140,0C(170),0C(K),3,5)		FATH0345
052	05(6)=CODIMP(00(1)+140,0C(170),0C(K+3),3,5)		FATH0350
053	GO TO 31		FATH0355
054	28 IF (00(10)-00(1)+10)24,20,24		FATH0360
055	28 IF (00(1)+130)-00(1)+10)24,30,24		FATH0365
056	30 IF (00(1)-00(1)+90)127,31,27		FATH0370
057	31 NY=00(1)+112		FATH0375
058	GO TO 32,32,30,37,40,42,NA		FATH0380
059	32 K=20		FATH0385
060	GO TO 144,33,34,35,NY		FATH0390
061	33 K=0		FATH0395
062	GO TO 44		FATH0400
063	34 K=00		FATH0405
064	GO TO 44		FATH0410
065	35 K=00		FATH0415
066	GO TO 44		FATH0420
067	36 K=120		FATH0425
068	GO TO 44		FATH0430
069	37 K=100		FATH0435
070	IF (NY-2144,30,30)		FATH0440
071	38 K=100		FATH0445
072	GO TO 44		FATH0450

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
004	00 00 J=1.20		FATH0700
005	K=J+1		FATH0705
006	00(K)=00(J-20)*00(51)		FATH0710
007	00(K+1)=00(J-20)*00(52)		FATH0715
008	00(K+51)=00(J-20)*00(40)		FATH0740
009	00(K+52)=00(J-20)*00(40)		FATH0745
1000	00 I=1+2		FATH0750
1001	00(M+200)=00(51)		FATH0751
1002	00(M+210)=00(52)		FATH0752
1003	00(M+210)=00(40)		FATH0753
1004	00(M+200)=00(40)		FATH0754
1005	00(400)=0.0		FATH0755
1006	00(500)=10.0		FATH0760
1007	00(511)=11.0		FATH0765
1008	00(50)=00(131)/2.0		FATH0770
1009	I=175		FATH0810
1010	J=175		FATH0812
1011	M=00(141)*00.00.03		FATH0813
1012	00 J=101		FATH0814
1013	M=00(137)-2.0*00.00.05		FATH0815
1014	00 I=101		FATH0820
1015	00 00(40)=00(100)*00(140),00(170),00(11.3..5)		FATH0825
1016	00(40)=00(100)*00(140),00(170),00(11.3..5)		FATH0830
1017	00(512)=00(50)*00(40)+00(10)*00(57)/2.0		FATH0835
1018	00(513)=00(50)*00(40)+00(20)*00(57)/2.0		FATH0840
1019	00(514)=00(170)		FATH0845
1020	00(515)=00(100)*00(170),00(11.3..5)*1.2		FATH0850
1021	00(516)=00(100)*00(170),00(11.3..5)*1.2		FATH0855
1022	00(517)=00(170)		FATH0860
1023	07 I=1		FATH0862
1024	K=1		FATH0864
1025	L=1		FATH0867
1026	M=1		FATH0868
1027	M=10(50)*00(1.0001.9502		
1028	0001 CONTINUE		
1029	00 IF K=00(50),04,04		FATH0870
1030	04 WRITE(6,04)*00(1)		FATH0872
1031	04 FORMAT(1H),00H,20H** FATH0 - (P100) ****		FATH0873
1032	I 4X,10#SPECTRA SEGMENT NO,F4 0//7X,		FATH0874
1033	I 1#DEF SEND MEM,3X,10#S05 SEND MEM,5X,10#NEEDANCES-TAKI		FATH0875
1034	00 TO 07		FATH0876
1035	00 WRITE(6,00)*00(1)		FATH0878
1036	00 FORMAT(1H),00H,20H** FATH0 - (P100) ****		FATH0879
1037	I 4X,10#SPECTRA SEGMENT NO,F4 0//7X,		FATH0880
1038	I 1#DEF SEND MEM,3X,10#S05 SEND MEM,5X,10#NEEDANCES-OUT,3X,		FATH0880
1039	2 10#NEEDANCES-RUN)		FATH0884
1040	07 00 74 J=1.20		FATH0891
1041	IF J=1170,70,70		FATH0892
1042	70 WRITE(6,71)		FATH0894
1043	71 FORMAT(1H)		FATH0895
1044	72 K=J+1		FATH0896
1045	IF K=00(101),101,00		FATH0898
1046	00 WRITE(6,100)*00(K),00(K+1),00(K+2)		FATH0899
1047	100 FORMAT(1H 4#F15.0,5#E12.5)		FATH0903
1048	00 TO 74		FATH0904
1049	101 WRITE(6,73)*00(K),00(K+1),00(K+2),00(M+L)		FATH0905
1050	73 FORMAT(1H 4#F15.0,5#E12.5,5#E12.5)		FATH0907
1051	L=L+1		FATH0908
1052	74 I=1+2		FATH0909
1053	IF K=00(102),102,103		FATH0911
1054	102 WRITE(6,00)*00(M+200),00(M+210)		FATH0912
1055	00 FORMAT(1H 10#F15.0,F15.0,3X,10#10 MOS M=F15.0)		FATH0913
1056	M=M+1		FATH0914
1057	103 IF 10(30)170,70,70		FATH0915
1058	70 IF K=00(77),9502,9502		
1059	70 IF K=00(77),70,70		FATH0918
1060	77 I=K+3		FATH0919
1061	00 TO 00		FATH0919
1062	70 WRITE(6,70)*00(511)		FATH0920
1063	70 FORMAT(1H),00H,20H** FATH0 - (P100) ****		
1064	I 4X,10#SPECTRA SEGMENT NO,F4 0//3X,2#GROUND-AIR-GRD		

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
1130	100 NS = N1-3		FC000100
1131	00 TO 000		FC000100
1132	170 NS = E-2		FC000170
1133	200 00 200 1 = 1.04		FC000170
1140	L = (NS-1) * 42 + 1		FC000100
1141	VS(1) = V(LNS)		FC000100
1142	T(1) = CDBTIME(1,1),Z(L),NS,NS		FC000100
1143	300 NS = N3-1		FC000100
1144	FC0002 = CDBTIME(1,VS,1,NS,NS)		FC000200
1145	RETURN		FC000200
1146	END		FC000210
1147	C		
1148	C (.....)		
1149	C SUBROUTINE FUEL1		
1150	C (.....)		
1151	C		
1152	C SUBROUTINE FUEL1		
1153	C		
1154	C		
1155	C SUBROUTINE TO STORE TOTAL AIRLOADS DATA FOR FUELAGE		
1156	C IN RECORD 33		
1157	C		
1158	COMMON TCON(400),MLD(300),IDUM(132),SAGE(132),MDE(132),		
1159	* RHE(132),RC(44),SW(100),STEPH(23),STEPM(23),STEPP(23),		
1160	* BLPH(100)		
1161	C		
1162	COMMON /MISC/ MISC(100)		
1163	C		
1164	DIMENSION MD(200),DD(620),SC(100)		
1165	DIMENSION TCON(23)		
1166	DIMENSION FUS(67),DD(20)		
1167	C		
1168	EQUIVALENCE (MD(1),TCON(420)),(DD(1),TCON(330)),(SC(1),TCON(270)		
1169	* (1,MD(101)),(DD(1),TCON(205))		
1170	EQUIVALENCE (SW(110),TCON(11))		
1171	C		
1172	IF(1) = 1) 700,700,0		
1173	0 K = 1 - 1		
1174	00 10 J=1,K		
1175	IF(MD(J+100)) 10,10,710		
1176	10 CONTINUE		
1177	C		
1178	700 00 701 N=1,070		
1179	701 FUS(N) = 0		
1180	C		
1181	710 IF(1) = 23) 019,019,0012		
1182	C		
1183	019 NK = 1 - 1 = 20		
1184	C		
1185	IF(1) = 0)001,002,003		
1186	001 FUS(NK+2) = 1		
1187	00 TO 012		
1188	002 FUS(NK+2) = 2		
1189	00 TO 012		
1190	003 IF(1) = 0)04,04,045		
1191	004 FUS(NK+2) = 3		
1192	FUS(230) = MISC(35)		
1193	FUS(237) = MISC(35)		
1194	00 TO 012		
1195	005 IF(1) = 0)000,000,007		
1196	005 FUS(NK+2) = 4		
1197	00 TO 012		
1198	007 IF(1) = 20)000,000,010		
1199	007 FUS(NK+2) = 5		
1200	00 TO 012		
1201	009 FUS(NK+2) = 6		
1202	00 TO 012		
1203	010 IF(1) = 02)000,011,011		
1204	011 FUS(NK+2) = 7		
1205	C		
1206	012 IF(MD(137))020,021,020		

07/05/74	INPUT LISTING	AUTOFLOW CHART SET - MEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
1007	C		
1008	GO FUSION(0) = SC(10)		
1009	GO TO 002		
1010	C		
1011	GO FUSION(0) = SC(11)		
1012	C		
1013	GO FUSION(1) = 2.0		
1014	FUSION(0) = TSC(10)		
1015	FUSION(0) = TSC(11)		
1016	FUSION(1) = SC(10)		
1017	FUSION(0) = SC(10)		
1018	FUSION(10) = SC(20)		
1019	FUSION(11) = SC(21)		
1020	FUSION(10) = SC(10)		
1021	FUSION(15) = SC(0)		
1022	FUSION(16) = SC(3)		
1023	FUSION(17) = SC(0)		
1024	FUSION(18) = SC(0)		
1025	FUSION(19) = SC(10)		
1026	FUSION(20) = SC(0) = 2.0		
1027	FUSION(21) = SC(0)		
1028	FUSION(22) = SC(1)		
1029	FUSION(23) = SC(11) = 2.0		
1030	FUSION(24) = SC(13)		
1031	FUSION(25) = SC(12)		
1032	FUSION(26) = SC(15)		
1033	FUSION(27) = SC(17)		
1034	FUSION(28) = SC(16)		
1035	GO TO 3015		
1036	C		
1037	C **** WRITE THE FUSELAGE LOADS IN RECORD 33 ****		
1038	C		
1039	3012 FUSION(0) = TSC(17)		
1040	FUSION(1) = TSC(18)		
1041	FUSION(0) = 2.0		
1042	FUSION(0) = 0.0		
1043	FUSION(0) = 0.0		
1044	C		
1045	CALL WRITE(1,FUSION,070,33)		
1046	C		
1047	3015 CONTINUE		
1048	RETURN		
1049	END		
1050	C		
1051	C		
1052	C SUBROUTINE PARLOADS		
1053	C		
1054	C		
1055	C SUBROUTINE PARLOADS		
1056	C		
1057	COMMON TCON(400),MID(300),IDUP(132),SANE(132),MDET(132),		
1058	*,RDE(132),RC(144),SV(100),STEPW(23),STEPH(23),STEPV(23)		
1059	*,SLPH(100),SANE(100)		
1060	C****CCCCC*****		
1061	C		
1062	C		
1063	COMMON /MID/ MID(100)		
1064	C		
1065	DIMENSION MD(200),SD(620),TSC(10),STRESS(0),TEMP(0),		
1066	*,STH(0),STH(0),STV(0),TEMPH(0),TEMPH(0),TEMPV(0)		
1067	C		
1068	DIMENSION SC(100)		
1069	DIMENSION SD(20)		
1070	DIMENSION MD(100),IDUP(200)		PARLOADS
1071	C		
1072	DIMENSION VDS(11),SDS(11),TDS(11),WDS(11),SDH(11),		
1073	*,VDS(11),SDS(11),TDS(11),SDH(11),		
1074	*,VDS(11),SDS(11),TDS(11),SDH(11),		
1075	*,VDS(11),SDS(11),TDS(11),SDH(11),		
1076	*,WDS(11),SDH(11)		
1077	C		

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07/05/74      INPUT LISTING      AUTOFLON CART SET - SHEEP      AIRLOADS OVERLAY

CARD NO      ****      CONTENTS      ****

1270      EQUIVALENCE (1001,TCOM+201), (1001,TCOM+202), (1,100101)
1275      C
1280      EQUIVALENCE
1281      * (1001101,TEMP111),
1282      *
1283      * (1001101,TEMP111), (1001101,STN111), (1001101,TEMP111),
1284      * (1001101,STV111), (1001101,TEMP111), (1001101,TEMP111),
1285      * (1001101,TEMP111), (1001101,TEMP111), (1001101,TEMP111),
1286      * (1001101,TEMP111), (1001101,TEMP111), (1001101,TEMP111),
1287      C
1288      EQUIVALENCE (1001,MLD111), (1001,MLD112), (1001,MLD113),
1289      * (1001,MLD114), (1001,MLD115), (1001,MLD116), (1001,MLD117),
1290      * (1001,MLD118), (1001,MLD119), (1001,MLD120), (1001,MLD121),
1291      * (1001,MLD122), (1001,MLD123), (1001,MLD124), (1001,MLD125),
1292      * (1001,MLD126), (1001,MLD127), (1001,MLD128), (1001,MLD129),
1293      * (1001,MLD130), (1001,MLD131), (1001,MLD132), (1001,MLD133),
1294      * (1001,MLD134), (1001,MLD135), (1001,MLD136), (1001,MLD137),
1295      * (1001,MLD138), (1001,MLD139), (1001,MLD140), (1001,MLD141),
1296      * (1001,MLD142), (1001,MLD143), (1001,MLD144), (1001,MLD145),
1297      * (1001,MLD146), (1001,MLD147), (1001,MLD148), (1001,MLD149),
1298      * (1001,MLD150), (1001,MLD151), (1001,MLD152), (1001,MLD153),
1299      C
1300      EQUIVALENCE (1001,TCOM+2700)
1301      EQUIVALENCE (1001,TCOM+2003)
1302      C
1303      C
1304      C
1305      C
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1906      C
1907      C
1908      C
1909      C
1910      C
1911      C
1912      C
1913      C
1914      C
1915      C
1916      C
1917      C
1918      C
1919      C
1920      C
1921      C
1922      C
1923      C
1924      C
1925      C
1926      C
1927      C
1928      C
1929      C
1930      C
1931      C
1932      C
1933      C
1934      C
1935      C
1936      C
1937      C
1938      C
1939      C
1940      C
1941      C
1942      C
1943      C
1944      C
1945      C
1946      C
1947      C
1948      C
1949      C
1950      C
1951      C
1952      C
1953      C
1954      C
1955      C
1956      C
1957      C
1958      C
1959      C
1960      C
1961      C
1962      C
1963      C
1964      C
1965      C
1966      C
1967      C
1968      C
1969      C
1970      C
1971      C
1972      C
1973      C
1974      C
1975      C
1976      C
1977      C
1978      C
1979      C
1980      C
1981      C
1982      C
1983      C
1984      C
1985      C
1986      C
1987      C
1988      C
1989      C
1990      C
1991      C
1992      C
1993      C
1994      C
1995      C
1996      C
1997      C
1998      C
1999      C
2000      C

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07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AMPLITUDE OVERLAY
CARD NO	CONTENTS		
1340	001 IF IHTIME(10)1000.000.001		
1350	C		
1351	001 DO 005 N=1.0		
1352	STRESS(1) = STMIN		
1353	005 TEMP(1) = TEMPMIN		
1354	NTMP = NMIN		
1355	GO TO 010		
1356	C		
1357	002 IF IHTIME(10)1000.000.002		
1358	C		
1359	002 DO 006 N=1.0		
1360	STRESS(1) = STMIN		
1361	006 TEMP(1) = TEMPMIN		
1362	NTMP = NMIN		
1363	GO TO 010		
1364	C		
1365	003 IF IHTIME(20)1000.000.003		
1366	C		
1367	003 DO 007 N=1.0		
1368	STRESS(1) = STMIN		
1369	007 TEMP(1) = TEMPMIN		
1370	NTMP = NMIN		
1371	C		
1372	010 IF IHTMP = 1000.000.000		
1373	C		
1374	004 STRES1 = STRESS(1)		
1375	GO TO 000		
1376	C		
1377	005 IF IHTIME(1) = TEMP(1)1000.000.005		
1378	C		
1379	005 LI = 1		
1380	LB = 2		
1381	GO TO 000		
1382	C		
1383	006 IF IHTIME(1) = TEMP(1)1000.000.006		
1384	C		
1385	006 LI = IHTMP - 1		
1386	LB = IHTMP		
1387	GO TO 000		
1388	C		
1389	075 DO 002 N=2, IHTMP		
1390	IF IHTIME(1) = TEMP(1)1000.000.002		
1391	075 LI = N - 1		
1392	LB = N		
1393	GO TO 000		
1394	002 CONTINUE		
1395	C		
1396	005 STRES1 = STRESS(LB) - (STRESS(LB) - STRESS(LI)) *		
1397	*(TEMP(LB) - TMIN(1)) / (TEMP(LB) - TEMP(LI))		
1398	C		
1399	C		
1400	000 GO TO (011,012,013),N		
1401	C		
1402	011 STEPM(1) = STRES1		
1403	STN = 5004 / STEPM(1)		
1404	GO TO 000		
1405	C		
1406	012 STEPM(1) = STRES1		
1407	STN = 5004 / STEPM(1)		
1408	GO TO 000		
1409	C		
1410	013 STEPM(1) = STRES1		
1411	STN = 5004 / STEPM(1)		
1412	C		
1413	000 CONTINUE		
1414	C		
1415	C **** SET LOADS AT TIP TO 0.0 ****		
1416	C		
1417	00 00 111 = 32,130,93		
1418	IF IADS(001111) = 10.0191.01.02		
1419	01 00111 = 0.0		

07/03/74	INPUT LISTING	AUTOTON CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
1400	C*****CCCC*****		
1401	00 IF (ABS(BD(111)+1)) - 10.01 03.03.04		
1402	C*****CCCC*****		
1403	00 BD(111)+1 = 0.0		
1404	C*****CCCC*****		
1405	04 IF (ABS(BD(111)+2)) - 10.01 05.05.06		
1406	00 BD(111)+2 = 0.0		
1407	C*****CCCC*****		
1408	00 CONTINUE		
1409	C		
1410	IF (ABS(BD(110),100,100		
1411	C		
1412	C **** LOOP 00 - - WIND LOADS ****		
1413	C		
1414	100 00 00 N=1,11		
1415	C		
1416	J = 10 - N		
1417	C		
1418	111 = 10 + (N-1) * 4		
1419	C		
1420	100 = 0		
1421	C		
1422	00 101,1,1,1,0,1,0,1,0,1,1,0,0,1,1,0,0,1,1,1,1,1,1,1,1		
1423	C		
1424	1 WPC = V2(J)		
1425	00PC = 001(J)		
1426	00 TO 10		
1427	C		
1428	2 WPC = V2(J)		
1429	00PC = 002(J)		
1430	100 = 1		
1431	00 TO 10		
1432	C		
1433	3 WPC = V2(J)		
1434	00PC = 003(J)		
1435	00 TO 10		
1436	C		
1437	4 WPC = V2(J)		
1438	00PC = 004(J)		
1439	C		
1440	10 WET = (BD(111)) + WPC + BD(10) * RM		
1441	00ET = (BD(111)+1) + 00PC + BD(10) * RM		
1442	C		
1443	IF (WET) 20,40,20		
1444	C		
1445	20 IF (WET - 00ET) 40,21,21		
1446	C		
1447	21 SAME(11) = BD(111)		
1448	00ET(11) = WET		
1449	100(11) = 1		
1450	00ET(11) = BD(10) / 100(11)		
1451	C		
1452	N = N		
1453	C		
1454	20 RE(11) = 1.0		
1455	C		
1456	IF (100) 25,25,40		
1457	C		
1458	25 IF (ABS(W2(J) - W2(J)) - 10.0140,40,05		
1459	00 RE(11) = (WPC - W2(J)) / (W2(J) - W2(J))		
1460	00 TO 40		
1461	C		
1462	3 IF (WET - 00ET(11)) 31,31,40		
1463	C		
1464	31 SAME(11) = BD(111)		
1465	00ET(11) = WET		
1466	100(11) = 1		
1467	00ET(11) = BD(10) / 00ET(11)		
1468	C		
1469	N = N + 11		
1470	00 TO 35		

07/03/76	INPUT LISTING	AUTOLON CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
1401	C		
1402		40 IF (BNET) 100.00.00	
1403	C		
1404		50 IF (BNET) - BNETIN=20:100.01.01	
1405	C		
1406		51 SAMEIN=20) = 00:111:11	
1407	C	*****	
1408		BNETIN) = 00:111:21	
1409	C	*****	
1410		BNETIN=20) = BNET	
1411		ISLPHIN=20) = 1	
1412		BNETIN=20) = 00:10) / POUND	
1413	C		
1414		N = N + 20	
1415	C		
1416		05 RCIN) = 1 0	
1417	C		
1418		IF (IND) 50.00.00	
1419	C		
1410		00 IF (ABS(BNETJ) - 0000:J) - 10 0:00.00.00	
1411		00 RCIN) = (BNET - 0000:J) / (BNETJ) - 0000:J)	
1412		00 TO 00	
1413	C		
1414		00 IF (BNET) - BNETIN=33:101.01.00	
1415	C		
1416		01 SAMEIN=33) = 00:111:11	
1417	C	*****	
1418		BNETIN=11) = 00:111:21	
1419	C	*****	
1420		BNETIN=33) = BNET	
1421		ISLPHIN=33) = 1	
1422		BNETIN=33) = 00:10) / MEGHZ	
1423	C		
1424		N = N + 33	
1425		00 TO 05	
1426	C		
1427	C	00 TAIL CONDITION MIND ONLY	
1428		14 VNET = VNETJ	
1429		BNET = BNETJ	
1430	C	TEST DOWN LOADS (NEGATIVE)	
1431		IF (VNET) 13.17.17	
1432	C		
1433		13 IF (VNET - BNETIN=11) 10.17.17	
1434	C		
1435		10 SAMEIN=11) = VNET	
1436		BNETIN=11) = VNET	
1437		ISLPHIN=11) = 1	
1438		BNETIN=11) = 0 0	
1439		RCIN=11) = 1 0	
1440	C		
1441		17 IF (BNET) 10.00.00	
1442		10 IF (BNET - BNETIN=33) 10.00.00	
1443	C		
1444		10 SAMEIN=33) = BNET	
1445		BNETIN=33) = BNET	
1446		ISLPHIN=33) = 1	
1447		BNETIN=33) = 0 0	
1448		RCIN=33) = 1 0	
1449		BNETIN=11) = 100:J)	
1450	C		
1451		00 CONTINUE	
1452	C		
1453		105 IF (IND) 23:105.105.107	
1454		100 IF (IND) 05:100.100.107	
1455		100 IF (IND) 77:105.105.107	
1456	C		
1457	C	**** LOOP 100 - - HORIZONTAL TAIL LOADS ****	
1458	C		
1459		107 00 10:0.0.0.0.7.0.7.0.0.0.7.7.0.0.7.7.0.0.0.0.0.0.105:1	
1460	C		
1461		0 JCS = 0C:10)	

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07/03/74          HPPL/1 LISTING          AUTOFLOW CHART SET - SHEEP          AIRLOADS OVERLAY

CARGO NO          *****          COMMENTS          *****

1002              GO TO 10
1003              C
1004              7 HES = EC(10)
1005              GO TO 10
1006              C
1007              8 HES = EC(10)
1008              GO TO 10
1009              C
1010              9 HES = EC(10)
1011              C
1012              10 IF (HES(10)) 11,11,12
1013              11 IF (HES(10)) 100,100,12
1014              12 HES(10) = EC(107) + .057 + EC(100)
1015              C
1016              HTHZ = ES(10) + (ES(20) + (HES(10) - HES) / 200.4)
1017              C
1018              GO 100 H=1,11
1019              C
1020              111 = ES + (H-11)*
1021              C
1022              J = 12 - H
1023              C
1024              VDET = (ES(111)) + VHI(J) + HTHZ + PSH
1025              SHET = (ES(111+1)) + SHHI(J) + HTHZ + PSH
1026              C
1027              IF (VDET) 120,140,120
1028              C
1029              120 IF (VDET) - SHET (H+4) 1140,121,121
1030              C
1031              121 SAME (H+4) = ES(111)
1032              SHET (H+4) = VDET
1033              ISLPH (H+4) = 1
1034              SHZ (H+4) = HTHZ / PSHZ
1035              GO TO 140
1036              C
1037              130 IF (VDET) - SHET (H+50) 131,131,140
1038              C
1039              131 SAME (H+50) = ES(111)
1040              SHET (H+50) = VDET
1041              ISLPH (H+50) = 1
1042              SHZ (H+50) = HTHZ / PSHZ
1043              C
1044              140 IF (SHET) 100,100,150
1045              C
1046              150 IF (SHET) - SHET (H+60) 150,151,151
1047              C
1048              151 SAME (H+60) = ES(111+1)
1049              C*****CCCCC*****
1050              SAME (H+60) = ES(111+2)
1051              C*****CCCCC*****
1052              SHET (H+60) = SHET
1053              ISLPH (H+60) = 1
1054              SHZ (H+60) = HTHZ / PSHZ
1055              GO TO 160
1056              C
1057              160 IF (SHET) - SHET (H+77) 161,161,160
1058              C
1059              161 SAME (H+77) = ES(111+1)
1060              C*****CCCCC*****
1061              SAME (H+77) = ES(111+2)
1062              C*****CCCCC*****
1063              SHET (H+77) = SHET
1064              ISLPH (H+77) = 1
1065              SHZ (H+77) = HTHZ / PSHZ
1066              C
1067              160 CONTINUE
1068              C
1069              160 IF (HES(23)) 101,101,100
1070              101 IF (HES(27)) 100,100,100
1071              C
1072              C ***** LOOP END - - VERTICAL TAIL LOADS *****

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07/03/74	INPUT LISTING	AUTOLON CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
1633	C		
1634		102 RECVT = BC(130) * .007 * BC(140)	
1635		VINY = BD(110) - 100*211 * 100VY - RCO / 200 * 1	
1636		DO 200 N=1,11	
1637	C		
1638		111 = 130 * 14-1114	
1639	C		
1640		J = 12 - N	
1641	C		
1642		WET = ABS(BD(111)) * W(1,J) * VINY * RSV	
1643		BET = ABS(BD(111+1)) * B(1,J) * VINY * RSV	
1644	C		
1645		IF WET - BET IN 001240,240,201	
1646	C		
1647		201 SAVE(N+00) = ABS(BD(111))	
1648		SAVE(N+00) = -SAVE(N+00)	
1649		BET(N+00) = WET	
1650		10URN(N+00) = 1	
1651		10URN(N+00) = 10URN(N+00)	
1652		RQZ(N+00) = VINY / POSHZ	
1653		RQZ(N+00) = RQZ(N+00)	
1654	C		
1655		240 IF BET - BET IN 1101200,200,251	
1656	C		
1657		251 SAVE(N+110) = ABS(BD(111+1))	
1658		SAVE(N+121) = -SAVE(N+110)	
1659		C*****CCCCC*****	
1660		SAVE(N+44) = BD(111+2)	
1661		SAVE(N+50) = -SAVE(N+44)	
1662		C*****CCCCC*****	
1663		BET(N+110) = BET	
1664		10URN(N+110) = 1	
1665		10URN(N+121) = 10URN(N+110)	
1666		RQZ(N+110) = VINY / POSHZ	
1667		RQZ(N+121) = RQZ(N+110)	
1668	C		
1669		200 CONTINUE	
1670	C		
1671		9-10-73--SETUP BD(1-200) FOR LOADS DATA SAVE ON RCD (100-103)*	
1672		100 IF 1 - 91 200,000,200	
1673		200 BD(101) = BC(1)	
1674		BD(103) = MD(103)	
1675	C		
1676		*TEST FOR LOAD 1-23	
1677		IF 11 - 241 201,200,202	
1678		201 BD(102) = BD(1)	
1679		IF(BD(102) - BD(101)) 205,200,205	
1680		205 BD(104) = MD(104)	
1681		BD(105) = MD(105)	
1682		GO TO 207	
1683		205 BD(104) = 0.0	
1684		BD(105) = 0.0	
1685		207 BD(106) = TSK(NF(1))	
1686		BD(107) = TSK(NF(1))	
1687		BD(108) = TSK(NF(1))	
1688		GO TO 204	
1689	C		
1690		*LOAD 24--2-6 TAXI*	
1691		202 BD(102) = BC(1)	
1692		BD(104) = 0.0	
1693		BD(105) = 0.0	
1694		BD(106) = 00.0	
1695		BD(107) = 00.0	
1696		BD(108) = 00.0	
1697	C		
1698		*CLEAR LOADS PART OF BD ARRAY--BC(1-100)*	
1699		DO 203 N=1,100	
1700		BD(N) = 0.0	
1701		203 CONTINUE	
1702	C		
1703		*SAVE BD(1-200) IN RCD 100-103 BASED ON INDICATOR (1)*	

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07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	ARL 0409 07/03/74
CARD NO	CONTENTS		
1775	K=120		33000710
1776	GO TO 20		33000720
1777	10 BO(120)=BC(145)		33000730
1778	BO(120)=BC(145)		33000740
1779	IF(ND(23))40.40.20		33000750
1780	40 IF(ND(27))41.41.20		33000760
1781	20 IF(ND(15))21.22.23		33000770
1782	21 ID=5		33000780
1783	GO TO 24		33000790
1784	22 ID=4		33000800
1785	05(1)=05+BO(15)		33000810
1786	GO TO 25		33000820
1787	23 ID=6		33000830
1788	24 05(1)=BO(15)		33000840
1789	25 05(2)=BC(144)+12		33000850
1790	05(3)=BC(130)		33000860
1791	I=0		33000870
1792	DO 27 J=1,12		33000880
1793	K=I+J+120		33000890
1794	BO(K)=05(I)*BU(J+73)/COS(05(3)/57.3)		33000900
1795	27 I=I+3		33000910
1796	I=171		33000920
1797	J=120		33000930
1798	K=107		33000940
1799	20 IF(10-5)20.30.32		33000950
1800	20 05(5)=0.0		33000960
1801	GO TO 33		33000970
1802	30 05(6)=BC(104)/COS(05(3)/57.3)		33000980
1803	IF(BC(104)-BC(145))29.32.32		33000990
1804	32 05(5)=1.0		33001000
1805	33 BO(J)=BO(J)+05(1)*BU(I)		33001010
1806	BO(J)=BO(J)+05(1)*BU(I+1)+05(5)*ABS(.30*BO(11)+BO(12))		33001020
1807	BO(J)=BO(J)+05(1)*BU(I+2)		33001030
1808	DO 30 L=2.40.4		33001040
1809	M=J+L+2		33001050
1810	BO(M)=BO(M)+05(1)*BU(K)		33001060
1811	IF(10-5)37.34.37		33001070
1812	34 IF(05(6)-BO(M-1))35.36.36		33001080
1813	35 05(5)=0.0		33001090
1814	GO TO 37		33001100
1815	36 05(5)=1.0		33001110
1816	37 VI=COS(05(3)/57.3)		33001120
1817	V2=SIN(05(3)/57.3)		33001130
1818	BO(M+1)=BO(M+1)+05(1)*BU(K+3)+05(5)*ABS(.30*BO(11)+BO(12))*VI		33001140
1819	BO(M+2)=BO(M+2)+05(1)*BU(K+25)+05(5)*ABS(.30*BO(11)+BO(12))*V2		33001150
1820	30 K=K+1		33001160
1821	IF(10-2)30.15.40		33001170
1822	30 IF(05(5))14.15.14		33001180
1823	40 IF(10-4)10.41.41		33001190
1824	41 CONTINUE		33001200
1825	IF(10(5))9501.9501.9502		
1826	9501 CONTINUE		
1827	IF(ND(23))40.40.50		33001210
1828	40 IF(ND(25))51.51.50		33001220
1829	50 WRITE(6,42)BO(1),BO(10),BO(10),BO(5)		33001230
1830	42 FORMAT(1H,3X,10MIN LOADS,3X,0=COND NO,7F.0,3X,34PM,7F.3,		
1831	• 5X,44LT=,7F.0,5X,34CF=,7F.2,10X,20H=• 0PASH - (P(5)) ••		
1832	WRITE(6,43)BO(1),1-23.20		33001240
1833	43 FORMAT(1H,3X,0=COND DIST=,7F.2,3HIN,4X,10MIN AT SOB=,7F.0,34LB,4X,		
1834	• 10MIN AT SOB=,7F.2,0,3HIN-LB,4X,10MIN AT SOB=,7F.2,0,3HIN-LB///		
1835	• 13X,74STATION,10X,34MEAP,10X,0=COND NOM,21X,0HORS NOM/		
1836	• 14X,44(1H,21X,44LB),10X,34IN-LB),22X,34IN-LB//)		
1837	WRITE(6,44)BO(1),1-27.74		33001250
1838	44 FORMAT(1H 7F.2,724.0,720.0,720.0)		33001260
1839	9502 CONTINUE		
1840	51 IF(10(5))9503.9503.9504		
1841	9503 CONTINUE		
1842	IF(ND(23))52.52.53		
1843	52 IF(ND(25))54.54.53		33001270
1844	53 WRITE(6,45)BO(1),BO(10),BO(10)		33001280
1845	45 FORMAT(1H, 3X,210HORIZONTAL TAIL LOADS,3X,0=COND NO=,7F.0,3X,		

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07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLORDS OVERLAY
CARD NO	CONTENTS		
1017	11 00(11)=0C(1,0)		31000000
1018	00(13)=0C(1,0)		31000000
1019	00(14)=0C(1,0)		31000000
1020	00 TO 10		31000010
1021	12 00(11)=0C(1,3)		31000000
1022	00(13)=0C(1,3)		31000000
1023	00(14)=0C(1,3)		31000000
1024	00 TO 10		31000000
1025	13 00(11)=0C(1,6)		31000000
1026	00(13)=0C(1,6)		31000000
1027	00(14)=0C(1,6)		31000000
1028	00 TO 10		31000000
1029	14 00(11)=0.0=0C(1,3)		31000000
1030	00(13)=0C(1,3)		31000000
1031	00(14)=0C(1,3)		31000000
1032	15 IF(CM)=1.0:10.20.20		31000000
1033	16 IF(CM)=.00:10.10.17		31000000
1034	17 00(17)=.00		31000000
1035	00 TO 10		31000000
1036	18 00(17)=CM		31000000
1037	19 00(12)=ATM(17*(0.05(17)/0.7,3)-11.0-00(13)/(00(11)+1.0-00(13)))/00(13)		31000000
1038	11.0-00(17)+0.2)*0.7,3		31000000
1039	00(13)=CODE(00(17),0T(0),0T(1),10,1)		31000000
1040	00(13)=00(13)+00(13)		31000000
1041	L=0		31000000
1042	00 20 1=1,4		31000000
1043	K=14+000		31000000
1044	VA(1)=CODE(00(12),00(13),00(11),00(17),00(1),7,0,1)		31000000
1045	00 L=L+1		31000010
1046	IF(10(12)+0001,0001,0002		
1047	0001 CONTINUE		
1048	WRITE(6,21)00(11),00(12),00(13),00(13)		31000012
1049	21 FORMAT(1H,000,20H+ USPAH - (P(12) ***		
1050	1 11X,30VALUES FROM BCLAVE TABLES FOR 00,70,4,3X,0000,70,2,3X,		31000013
1051	2 3HTR=77,4,3X,0000,70,4)		31000014
1052	WRITE(6,22)0T(1),1=0,0),VA(1),1=1,0)		31000016
1053	22 FORMAT(10,7X,3HTR=,4F10,2//11X,4F10,5)		31000017
1054	0002 CONTINUE		
1055	IF(10(12)+0001,0001,0002		31000020
1056	23 1=1		31000030
1057	IF(10(12)+0001,0001,0002		
1058	0001 CONTINUE		
1059	WRITE(6,24)		31000034
1060	24 FORMAT(1H,03X,0HMIN(1)		31000035
1061	00 TO 20		31000040
1062	25 1=100		31000050
1063	IF(10(12)+0001,0001,0002		
1064	0002 CONTINUE		
1065	WRITE(6,26)		31000054
1066	26 FORMAT(1H,03X,10HMIN(1)		31000055
1067	00 TO 20		31000060
1068	27 1=107		31000070
1069	IF(10(12)+0001,0001,0002		
1070	0003 CONTINUE		
1071	WRITE(6,28)		31000074
1072	28 FORMAT(1H,03X,10HMIN(1)		31000075
1073	29 00(11)=CODE(00(13),0T(0),VA(1),0)=0.7,3/00(13)		31000080
1074	L=0		31000080
1075	00 30 1=1,10		31000080
1076	K=14+13		31000010
1077	VA(1)=CODE(00(12),00(13),00(11),00(17),00(1),7,0,1)		31000020
1078	30 L=L+1		31000030
1079	31 00 32 1=1,4		31000040
1080	L=1+14+1		31000050
1081	32 VA(1)=CODE(00(13),0T(0),VA(1),0)=0.7,3/00(13)		31000060
1082	IF(10(12)+0001,0001,0002		
1083	0003 CONTINUE		
1084	WRITE(6,33)CM		31000064
1085	33 FORMAT(10,10X,30VALUES FROM LOADING TABLES FOR 00,70,3)		31000064
1086	J=1		31000065
1087	K=4		31000067

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	****	CONTENTS	****
1000	L=1		31000000
1001	34 WRITE(10,35)YB(1),1-J,K) ,DT(L)		31000070
1002	35 FORMAT(1H0,10X,4F10.0,3X,4HSTAR,F0.3)		31000071
1003	J=J+4		31000072
1004	K=K+4		31000074
1005	L=L+1		31000075
1006	IF(L-4)34,34,46		31000076
1007	36 IF(C00-1)1137,30,30		31000080
1008	37 DS(17)=1		31000084
1009	GO TO 30		31000085
1010	38 DS(17)=C04		31000086
1011	39 DS(7)=DS(1)+SQRT(DS(17)**2-1.0)		31000087
1012	IF(DS(1))105,105,104		31000089
1013	104 DS(8) = SQRT(DS(17)**2-1.0)*COS(DS(4)/57.3)/SIN(DS(4)/57.3)		31000090
1014	IF(12.0 - DS(8))105,107,107		31000091
1015	105 DS(8) = 12.0		31000092
1016	107 L = 0		31000700
1017	DO 40 I=1,10		31000710
1018	K=K+14		31000720
1019	YB(1)=CODE(DS(8),DS(7),DP(6),DP(1),DP(4),0.0,0.5)		31000730
1020	40 L=L+4		31000740
1021	IF(10.52)9505,9505,31		
1022	9505 CONTINUE		
1023	WRITE(10,41)DS(1),DS(3),DS(7),DS(8)		31000742
1024	41 FORMAT(1H1,7X,3HAR=F0.4,3X,3HTR=F7.4,3X,3HAP=F0.4,3X,3HAP=		31000744
1025	1 F0.4,3X,3H***USPAM = (P(52) **)		31000745
1026	WRITE(10,42)DT(1),1-5,0)		31000746
1027	42 FORMAT(1H0,7X,3HTR=F0.4,3X,3HAP=F0.4,3X,3HAP=		31000748
1028	IF(NT-3)43,44,45		31000749
1029	43 WRITE(10,44)		31000750
1030	GO TO 31		31000761
1031	44 WRITE(10,46)		31000763
1032	GO TO 31		31000764
1033	46 WRITE(10,48)		31000765
1034	GO TO 31		31000767
1035	48 IF(NT-4)47,47,48		31000768
1036	47 L=5		31000770
1037	DO 40 I=1,4		31000780
1038	40 ED(1)=DT(1)		31000780
1039	ED(5)=1.0		31000800
1040	YC(5)=0.0		31000810
1041	GO TO 50		31000820
1042	40 L=3		31000830
1043	ED(1)=0.0		31000840
1044	ED(2)= 942		31000850
1045	ED(3)=1.0		31000860
1046	50 DO 55 I=1,13		31000870
1047	IF(NT-3)51,52,53		31000880
1048	51 K=1+0		31000890
1049	GO TO 54		31000900
1050	52 K=1+14		31000910
1051	GO TO 54		31000920
1052	53 K=1+175		31000930
1053	54 ES(1)=DS(K)		31000940
1054	55 YS(1)=CODE(ES(1),ED,YC,L,0)		31000950
1055	IF(10.52)9507,9507,9508		
1056	9507 CONTINUE		
1057	WRITE(10,56)CHN		31000952
1058	56 FORMAT(1H1,10X,7HSTATION,10X,7HLOADING AT DATA STATIONS,5X,3H***		31000953
1059	1 F0.3,3X,3H***USPAM = (P(52) **)		31000954
1060	WRITE(10,57)ED(1),YC(1),1-1,1,1)		31000955
1061	57 FORMAT(1H0,17X)		31000956
1062	WRITE(10,58)		31000957
1063	58 FORMAT(1H1,10X,7HSTATION,10X,7HLOADING-ANALYSIS STATIONS)		31000958
1064	IF(NT-3)59,60,61		31000959
1065	59 WRITE(10,62)		31000960
1066	GO TO 62		31000961
1067	60 WRITE(10,64)		31000962
1068	GO TO 62		31000963
1069	61 WRITE(10,66)		31000964
1070	62 WRITE(10,57)ES(1),YS(1),1-1,1,1)		31000965

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLADS OVERLAY
CARD NO	****	CONTENTS	****
0000	0000 CONT MAX		
0000	05(01)=0.0		31000000
0001	00 03 1-0.10		31000070
0002	03 05(01)=05(01)+ 0*(05(11)-11-05(11))+(05(11)-11)*05(11)		31000000
0003	00(00) = 0 0*(05(12) - (05(13))+(05(12) + 05(13))		31000001
0004	05(01) + 05(01) + 00(00)+00(00)		31000002
0005	00(00) + 05(01) + 11 0 - 00(00)+00(00)		31000003
0006	IF(C04 - 1 0) 100.04,04		31000000
0007	04 IF(07)-3100.00,07		31001000
0008	00 04(11)=05(11)+05(01)		31001010
0009	00 70 00		31001000
0070	00 04(100)=05(11)+05(01)		31001030
0071	00 70 00		31001040
0072	07 04(107)=05(11)+05(01)		31001000
0073	00 70 00		31001051
0074	100 IF(07) - 31 101.102,103		31001062
0075	101 04(11) + 04(11)+05(01)+00(00)		31001063
0076	00 70 00		31001004
0077	102 04(100) + 04(100)+05(01)+00(00)		31001000
0078	00 70 00		31001000
0079	103 04(107) + 04(107)+05(01)+00(00)		31001007
0080	00 IF(07)-3100.70,70		31001000
0081	00 1-33		31001070
0082	J=02		31001000
0083	K=35		31001000
0084	L=00		31001100
0085	05(13)+00(37)		31001110
0086	IF(107)71,70,71		31001100
0087	70 05(11)+00(70)		31001120
0088	05(101)+00(70)+0.0		31001140
0089	00 70 70		31001100
0090	71 05(11)+00(141)		31001100
0091	05(101)+00(140)+0.0		31001170
0092	00 70 70		31001100
0093	70 1-45		31001100
0094	IF(07)-173,74,74		31001200
0095	73 J=100		31001210
0096	K=141		31001200
0097	L=104		31001230
0098	05(111)+00(100)		31001240
0099	05(101)+00(112)+0.0		31001250
0100	05(131)+00(113)		31001200
0101	00 70 70		31001270
0102	74 05(111)+00(140)		31001200
0103	05(101)+00(144)+12.0		31001200
0104	05(131)+00(145)		31001300
0105	J=107		31001310
0106	K=200		31001300
0107	L=013		31001330
0108	75 05(141)+05(131)+05(12)		31001340
0109	05(101)+00(100)+00(01)+01(11)+12.51		31001300
0110	76 04(11)=0.0		31001350
0111	04(12)=0.0		31001370
0112	04(13)=0.0		31001300
0113	00(11)=0.0		31001300
0114	IF(07)-3177.00,01		31001300
0115	77 IF(107)70,70,70		31001300
0116	78 1-70		31001304
0117	J=00		31001300
0118	00 70 00		31001300
0119	79 1-30		31001400
0120	J=07		31001402
0121	00 70 00		31001404
0122	00 1-100		31001400
0123	J=105		31001400
0124	1-0 70 00		31001410
0125	01 1-130		31001412
0126	J=107		31001414
0127	00 05(101)+0*(11-0-05(11)+05(100))/07 31+005(00(11)/07 31)+05(11)+11		31001410
0128	1+00(11)		31001410
0129	00(101)+05(101)+11 0-05(101)+11 0-05(101)		31001400

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
0001	0011 CONTINUE		
0002	WRITE(6,100)CYN,0010		3100174
0003	100 FORMAT(1H1,00H,20H** UPRN = (P102) **//		
0004	11H,7H1ATION,10H,0HOUT LEND,10H,0HIND LEND,10H,0HAP INCR,		3100176
0005	2 5H,30H**70 3,5H,30H**70 21		3100177
0006	0012 CONTINUE		
0007	00 107 1-1,13		3100178
0008	J=10		3100179
0009	VS(1)=CODE(0U(1),05(10),07(20),07(10),07(37),10,11,5)		3100180
0010	VB(1)=CODE(0U(1),05(10),07(20),07(10),07(37),10,11,5)		3100180
0011	IF(10:02)0013,0013,0014		
0012	0013 CONTINUE		
0013	WRITE(6,100)ES(1),VS(1),VB(1)		3100179
0014	100 FORMAT(1H17,0,710,0,710,0)		3100175
0015	0014 CONTINUE		
0016	VS(1)=VS(1)-VB(1)		3100180
0017	IF(10:02)0015,0015,0016		
0018	0015 CONTINUE		
0019	WRITE(6,100)VS(1)		3100180
0020	100 FORMAT(1H34H710,0)		3100180
0021	0016 CONTINUE		
0022	107 CONTINUE		3100182
0023	00(0)=0		3100185
0024	00 100 1-2,12		3100188
0025	100 00(0)=05(0)+0*(ES(1)-1)-ES(1)+VS(1)-1+VB(1)		3100189
0026	00(00)=0.5*(ES(12)+ES(13))+VS(12)+VS(13)		3100191
0027	00(0)=00(0)+00(00)+00(00)		3100192
0028	00(00)=00(0)+(1.0-00(00))+00(00)		3100193
0029	0U(0)=00(0)		3100199
0030	05(0)=CODE(0C(102),07(11),07(0),0,5)		3100193
0031	J=00		3100194
0032	K=00		3100199
0033	L=00		3100199
0034	00 TO 70		3100197
0035	100 1-2		3100199
0036	J=0		3100199
0037	K=00		3100199
0038	L=0		3100199
0039	00 TO 114		3100199
0040	110 1-02		3100199
0041	J=0		3100194
0042	K=00		3100199
0043	L=00		3100199
0044	00 TO 114		3100197
0045	111 IF(10:02)112,113,113		3100199
0046	112 1-100		3100199
0047	J=115		3100199
0048	K=120		3100199
0049	L=112		3100199
0050	00 TO 114		3100199
0051	113 1-100		3100194
0052	J=170		3100199
0053	K=107		3100199
0054	L=171		3100197
0055	114 0U(1)=CODE(05(14),0U(1),0U(1),13,5)		3100199
0056	0U(1)=CODE(05(14),0U(1),0U(1),13,5)		3100199
0057	0U(1)=CODE(05(14),0U(1),0U(1),20,13,5)		3100199
0058	0U(1)=0U(1)*COS(BC(1)/57.3)+0U(2)*SIN(BC(1)/57.3)		3100200
0059	0U(2)=0U(1)*COS(BC(1)/57.3)-0U(1)*SIN(BC(1)/57.3)		3100200
0060	0U(1)=0U(1)		3100210
0061	IF(10:02)115,116,121		3100210
0062	115 1-3		3100210
0063	K=7		3100210
0064	00 TO 117		3100210
0065	116 1-03		3100210
0066	K=04		3100210
0067	117 IF(10:02)118,118,118		3100210
0068	118 J=70		3100210
0069	00 TO 120		3100210
0070	119 J=30		3100210
0071	120 0U(1)=05(13)+0U(1)+0U(1)+3)		3100210

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07-03/74	INPUT LISTING	AUTOFLEX CHART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	CONTENTS		
0743	IF (IND(23)=133,133,146		31002700
0744	146 WRITE(6,147) (BU(1),1=157,173),CMH		31002710
0745	147 FORMAT(1M1,5X,30AERT TAIL PARAMETERS,5X,20H** UPON - IP(52) **		31002720
0746	1 //2,WEIVB,FB 5,4X,SHVIB=,F7 5,4X,WDIV, F7 2,4X,WDIV,FB 2//		31002730
0747	7X,10TOP OF BODY UNITS//7X,SHVIB=,FB 5,4X,SHVIB=,FB 3,4X,		31002730
0748	3 SHVIB=,FB 3//7X,27SPANNISE UNIT DISTRIBUTIONS//5X,50TA,7X,		31002740
0749	4 SHVIB=,4X,SHVIB=,4X,SHVIB=,5X,50TA,FB 3//5X,SHVIB=,10X,31002750		
0750	5 SHVIB=,5X,SHVIB=//		31002750
0751	10 146 1=174,100		31002750
0752	1=13		31002770
0753	1=13		31002780
0754	1=13		31002780
0755	146 WRITE(6,137) (BU(1),BU(2),BU(3),BU(4))		31002800
0756	9910 CONTINUE		
0757	146 RETURN		31002810
0758	END		31002820
0759	C		
0760	C		
0761	C SUBROUTINE WPAET		
0762	C		
0763	C		
0764	C SUBROUTINE WPAET		
0765	C		
0766	C SUBROUTINE TO ARRANGE SURFACE AIRLOADS DATA AND		
0767	C NORMALIZING FACTORS		
0768	C		
0769	COMMON TCON(400),ALD(200),TDUP(132),SANE(132),MET(132),		
0770	• RUC(132),RL(44),SW(100),STEPH(23),STEPH(23),STEPH(23),		
0771	• DUP(100),SANE(100)		
0772	C		
0773	COMMON /PRINT/ IP(60)		
0774	COMMON /MISC/ MISC(100)		
0775	C		
0776	DIMENSION RATIO(204),RS(132)		
0777	DIMENSION MD(200)		
0778	DIMENSION TSH(123)		
0779	C		
0780	EQUIVALENCE (MD(1),TCON(400))		
0781	EQUIVALENCE (SW(110),TSH(1)), (SW(152),SANE)		
0782	C		
0783	C **** READ RECORD 30, THE WIND, HORIZONTAL AND VERTICAL LOADS ****		
0784	C****CCCCC****		
0785	CALL READ4(1,DUP(1),100,32)		
0786	C****CCCCC****		
0787	C		
0788	C **** READ RECORD 17, THE DEAD WEIGHT RATIOS ****		
0789	C **** (INITIALIZED AT 1 0 IN READ) ****		
0790	C****CCCCC****		
0791	CALL READ4(1,RATIO(1),204,17)		
0792	C****CCCCC****		
0793	C		
0794	C CLEAR LOADS WHICH ARE TO BE COMPUTED		
0795	C		
0796	IF (IND(23)=1400,1400,1401		
0797	C		
0798	1401 L1 = 1		
0799	C		
0800	C****CCCCC****		
0801	L2 = 100		
0802	C****CCCCC****		
0803	IND = 1		
0804	GO TO 14010		
0805	C		
0806	1400 IF (IND(23)=1400,1400,1402		
0807	C		
0808	1402 L1 = 1		
0809	L2 = 14		
0810	C****CCCCC****		
0811	GO 1402 1401,22		
0812	1402 DUP(1)=132 = 0 0		
0813	C****CCCCC****		

07/03/74	INPUT LISTING	AUTOFLEX CART SET - SHEEP	AIRLOADS OVERLAY
CARD NO	*****	CONTENTS	*****
014		IND = 2	
015		00 TO 4010	
016		C	
017		4005 IF (IND(20)) 4007, 4007, 4003	
018		C	
019		4003 L1 = 45	
020		L2 = 00	
021		C*****CCCCC*****	
022		00 4013 N=1.00	
023		4013 (SUM(N*10)) = 0.0	
024		C*****CCCCC*****	
025		IND = 3	
026		00 TO 4010	
027		C	
028		4007 IF (IND(27)) 4009, 4009, 4004	
029		C	
030		4004 L1 = 00	
031		L2 = 100	
032		C*****CCCCC*****	
033		00 4014 N=1.00	
034		4014 (SUM(N*10)) = 0.0	
035		C*****CCCCC*****	
036		IND = 4	
037		C	
038		4010 00 4011 N=1.12	
039		4011 (SUM(N)) = 0.0	
040		C	
041		00 TO 4005, 4005, 4007, 4009, 1.00	
042		C	
043		4000 CONTINUE	
044		C	
045		C IF THE NEGATIVE BENDING MOMENT AT THE ROOT OF THE HORIZONTAL	
046		C TAIL IS GREATER THAN THE POSITIVE BENDING MOMENT, REVERSE THE	
047		C HORIZONTAL TAIL POSITIVE AND NEGATIVE AIRLOADS (ARRAY DUM IN	
048		C RECORD 32) AND THE RATIOS (ARRAY RATIO IN RECORD 17)	
049		C	
050		IND(140) = 0.0	
051		C	
052		IF (IND(20)) 4007, 4007, 4003	
053		C	
054		4003 IF (IND(27)) = IND(20)) 401, 4007, 4007	
055		C	
056		401 IND(140) = 1.0	
057		C	
058		00 403 N=45.00	
059		C	
060		RESCALE = RES(N)	
061		RES(N) = RES(N+1)	
062		RES(N+1) = RESCALE	
063		C	
064		RESCALE = RES(N+20)	
065		RES(N+20) = RES(N+33)	
066		RES(N+33) = RESCALE	
067		C	
068		RESCALE = RES(N)	
069		RES(N) = RES(N+1)	
070		RES(N+1) = RESCALE	
071		C	
072		RESCALE = RES(N+20)	
073		RES(N+20) = RES(N+33)	
074		RES(N+33) = RESCALE	
075		C	
076		RATIO0 = RATIO(N)	
077		RATIO(N) = RATIO(N+1)	
078		RATIO(N+1) = RATIO0	
079		C	
080		RATIO0 = RATIO(N+20)	
081		RATIO(N+20) = RATIO(N+33)	
082		RATIO(N+33) = RATIO0	
083		C	
084		SAVE'V = SAVE(N)	

07/03/74	INPUT LISTING	AUTOFLOW CHART SET - SHEEP	AIRBORNE OVERLAY
CARD NO	****	CONTENTS	****
0405		SAVE IN1 = -SAVE IN111	
0406		SAVE IN111 = -SAVE SV	
0407	C		
0408		SAVE SV = SAVE IN221	
0409		SAVE IN221 = -SAVE IN331	
0410		SAVE IN331 = -SAVE SV	
0411	C		
0412		SAVE IN = BLP IN1	
0413		BLP IN1 = -BLP IN111	
0414		BLP IN111 = -SAVE IN	
0415	C		
0416		SAVE IN = BLP IN221	
0417		BLP IN221 = -BLP IN331	
0418		BLP IN331 = -SAVE IN	
0419	C		
0420		ISDN = (BLP IN1)	
0421		ISDN IN1 = (BLP IN111)	
0422		ISDN IN111 = ISDN	
0423	C		
0424		ISDN = (BLP IN221)	
0425		ISDN IN221 = (BLP IN331)	
0426		YES (BLP IN331) = (ISDN)	
0427	C	*****CCCCC*****	
0428	C		
0429	C	*** T. MEN ***	
0430		DO YES N = 1,11	
0431		SAVE IN = SAVE IN221	
0432		SAVE IN221 = SAVE IN331	
0433		YES SAVE IN331 = SAVE IN	
0434	C		
0435	C	*****CCCCC*****	
0436	C		
0437		YES CONTINUE	
0438	C		
0439	C	**** DETERMINE THE DESIGN LOAD CONDITIONS FOR THE WIND. ****	
0440	C	**** HORIZONTAL TAIL AND VERTICAL TAIL (IN, IN, IV) ****	
0441	C	**** THE DESIGN CONDITION IS THE CONDITION WHICH GIVES ****	
0442	C	**** THE MAXIMUM POSITIVE BENDING MOMENT AT THE ROOT ****	
0443	C		
0444	C	**** DETERMINE THE STRESS LEVEL AT THE DESIGN CONDITION ****	
0445	C	**** (STRE,STREH,STRV) AND THE TEMPERATURE AT THE DESIGN ****	
0446	C	**** CONDITION (TWIND,THT,TVT) FOR THE WIND, HORIZONTAL ****	
0447	C	**** AND VERTICAL ****	
0448	C		
0449		IF INDI2011405,406,402	
0450	C		
0451		YES IN = (BLP IN331)	
0452		STRE = STEPM IN1	
0453		TWIND = TWINF IN1	
0454		TWINDC101 = TWIND	
0455	C		
0456		DO YES N=1,40	
0457		II = (BLP IN1)	
0458		IF III = 241200,207,207	
0459		207 IF 10001210,210,200	
0460	C		
0461		200 RS IN1 = STRE / 1000	
0462		GO TO 213	
0463		200 IF 111210,210,214	
0464		214 IF 101010101210,210,211	
0465		211 IF 101010101210,210,212	
0466		210 RS IN1 = 1.0	
0467		GO TO 213	
0468		210 RS IN1 = STRE / STEPM III1	
0469		213 BLP IN1 = SAVE IN1 + RS IN1	
0470		RATIO IN1 = RS IN1 + RS IN1	
0471		YES RATIO IN1221 = RATIO IN1 + RC IN1	
0472	C	*****CCCCC*****	
0473	C		
0474		DO YES N=1,22	
0475		BLP IN1221 = SAVE IN1 + RS IN1221	

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```

07/03/74      INPUT LISTING      AUTOFLON CHART SET - SHEEP      AIRLOADS OVERLAY
CARD NO      ****      CONTENTS      ****
0007      WRITE(6,301)
0008      301 FORMAT(//2X,01HSTATION LC      +V      RS      RQZ      RC
0009      *      RS*(+V)      RS+RQZ      RS+RQZ+RC/)
0010      C
0011      WRITE(6,300)(N,IDLPH(N),SAVE(N),RS(N),RQZ(N),RC(N),IDLPH(N),RATIO(N),
0012      *      RATIO(N+100),N=1,11)
0013      C
0014      300 FORMAT(2X,12,5X,12,F13.0,F0.3,F0.3,F13.0,F0.3)
0015      C
0016      WRITE(6,302)
0017      302 FORMAT(//2X,01HSTATION LC      -V      RS      RQZ      RC
0018      *      RS*(+V)      RS+RQZ      RS+RQZ+RC/)
0019      C
0020      WRITE(6,300)(N,IDLPH(N+1),SAVE(N+1),RS(N+1),RQZ(N+1),RC(N+1),
0021      *      IDLPH(N+1),RATIO(N+1),RATIO(N+100),N=1,11)
0022      C
0023      WRITE(6,303)
0024      303 FORMAT(//2X,01HSTATION LC      +B      RS      RQZ      RC
0025      *      RS*(+B)      RS+RQZ      RS+RQZ+RC/)
0026      C
0027      WRITE(6,300)(N,IDLPH(N+2),SAVE(N+2),RS(N+2),RQZ(N+2),RC(N+2),
0028      *      IDLPH(N+2),RATIO(N+2),RATIO(N+100),N=1,11)
0029      C
0030      WRITE(6,304)
0031      304 FORMAT(//2X,01HSTATION LC      -B      RS      RQZ      RC
0032      *      RS*(+B)      RS+RQZ      RS+RQZ+RC/)
0033      C
0034      WRITE(6,300)(N,IDLPH(N+3),SAVE(N+3),RS(N+3),RQZ(N+3),RC(N+3),
0035      *      IDLPH(N+3),RATIO(N+3),RATIO(N+100),N=1,11)
0036      C
0037      400 IF (N00100) 405,405,402
0038      C
0039      402 WRITE(6,304)
0040      C
0041      WRITE(6,302)
0042      302 FORMAT(3X,23H*** HORIZONTAL TAIL ***)
0043      C
0044      WRITE(6,301)
0045      301 FORMAT(//2X,01HSTATION LC      +V      RS      RQZ      RS*(+
0046      *      V)      RS+RQZ/)
0047      C
0048      WRITE(6,300)(N,IDLPH(N+4),SAVE(N+4),RS(N+4),RQZ(N+4),
0049      *      IDLPH(N+4),RATIO(N+4),N=1,11)
0050      C
0051      300 FORMAT(2X,12,5X,12,F13.0,F0.3,F0.3,F13.0,F0.3)
0052      C
0053      WRITE(6,302)
0054      302 FORMAT(//2X,01HSTATION LC      -V      RS      RQZ      RS*(+
0055      *      V)      RS+RQZ/)
0056      C
0057      WRITE(6,300)(N,IDLPH(N+5),SAVE(N+5),RS(N+5),RQZ(N+5),
0058      *      IDLPH(N+5),RATIO(N+5),N=1,11)
0059      C
0060      WRITE(6,303)
0061      303 FORMAT(//2X,01HSTATION LC      +B      RS      RQZ      RS*(+
0062      *      B)      RS+RQZ/)
0063      C
0064      WRITE(6,300)(N,IDLPH(N+6),SAVE(N+6),RS(N+6),RQZ(N+6),
0065      *      IDLPH(N+6),RATIO(N+6),N=1,11)
0066      C
0067      WRITE(6,304)
0068      304 FORMAT(//2X,01HSTATION LC      -B      RS      RQZ      RS*(+
0069      *      B)      RS+RQZ/)
0070      C
0071      WRITE(6,300)(N,IDLPH(N+7),SAVE(N+7),RS(N+7),RQZ(N+7),
0072      *      IDLPH(N+7),RATIO(N+7),N=1,11)
0073      C
0074      404 IF (N0100) 405,405,405
0075      C
0076      405 WRITE(6,407)
0077      407 FORMAT(//7X,72HTHE POSITIVE AND NEGATIVE LOADS ON THE HORIZONTAL T

```

07/03/74

INPUT LISTING

AUTOFLOW CHART SET - SHEEP AIRLOADS OVERLAY

```

CAND NO      ****      CONTENTS      ****

2699          *ALL HAVE BEEN REVERSED/INX, BECAUSE THE NEGATIVE ON AT THE ROPT
2700          *WAS GREATER THAN THE POSITIVE ON AT THE ROPT)
2701          C
2702          *MS IF (ND=27) 3002, 3002, 472
2703          C
2704          *72 WRITE(6, 3044)
2705          C
2706          WRITE(6, 3043)
2707          3043 FORMAT(30N, 21N*** VERTICAL TAIL ***)
2708          C
2709          WRITE(6, 3041)
2710          C
2711          WRITE(6, 3001) (N, 10LPHN=00), SAME(N=00), RS(N=00), RZ(N=00),
2712          * 0LPHN=00), RATIO(N=00), N=1, 11)
2713          C
2714          WRITE(6, 3002)
2715          C
2716          WRITE(6, 3001) (N, 10LPHN=00), SAME(N=00), RS(N=00), RZ(N=00),
2717          * 0LPHN=00), RATIO(N=00), N=1, 11)
2718          C
2719          WRITE(6, 3043)
2720          C
2721          WRITE(6, 3001) (N, 10LPHN=110), SAME(N=110), RS(N=110), RZ(N=110),
2722          * 0LPHN=110), RATIO(N=110), N=1, 11)
2723          C
2724          WRITE(6, 3044)
2725          C
2726          WRITE(6, 3001) (N, 10LPHN=121), SAME(N=121), RS(N=121), RZ(N=121),
2727          * 0LPHN=121), RATIO(N=121), N=1, 11)
2728          C
2729          3002 CONTINUE
2730          C
2731          C      **** WRITE THE MIND, MY AND VY LOADS IN RECORD 32 ****
2732          C      **** AND WRITE THE HEAD HEIGHT RATIOS ON RECORD 17 ****
2733          C      **** IF ANY OF THE LOADS HAVE BEEN COMPUTED ****
2734          C
2735          IF (ND=25) 3001, 3001, 3004
2736          3001 IF (ND=26) 3002, 3002, 3004
2737          3002 IF (ND=26) 3003, 3003, 3004
2738          3003 IF (ND=27) 3005, 3005, 3004
2739          C
2740          C*****CCCC*****
2741          3004 CALL WRITE(1, 0LPH(1), 100, 32)
2742          CALL WRITE(1, RATIO(1), 204, 17)
2743          C*****CCCC*****
2744          3005 CONTINUE
2745          RETURN
2746          END

```

Appendix B

SAMPLE OUTPUT

Sample airload module output are shown in Figures B-1 through B-25. Print control indicators and originating routine name appear in the output samples.

** USPAN - IP(52) **
 VALUES FROM RCLAK TABLES FOR AR= 9.6000 SWB= 12.94 TR= 0.3500 MA/K= 10.0036
 TR= 0.0 0.25 0.50 1.00
 0.09438 0.09651 0.09526 0.09119 (WING)
 VALUES FROM LOADING TABLES FOR MN= 0.318
 1.53218 1.32994 1.21470 1.04169 STA= 0.0
 1.26499 1.17613 1.14092 1.11271 STA= 0.393
 0.73471 0.86268 0.93744 1.04151 STA= 0.707
 0.24414 0.50330 0.60127 0.69093 STA= 0.924

Figure B-1. Sample output from USPAN of wing lift curve slope parameter and spanwise loading parameter.

```

STATION      0.0      0.38300      0.70700      0.92400      1.09000
LOADING AT DATA STATIONS      MN= 0.318
1.27001
1.15001
0.89577
0.55048
0.0

LOADING-ANALYSIS STATIONS (WING)
0.0
0.17758
0.60454
0.74000
0.86827
0.97262
1.05040
1.11372
1.16937
1.20887
1.23799
1.26030
1.27901

NY SWEEP
-9.852
-10.266
-11.508
-13.164
-14.820
-16.476
-18.132
-19.788
-21.444
-23.100
-24.756
-26.412
-4.870

STATION      0.0      0.38300      0.70700      0.92400      1.09000
LOADING AT DATA STATIONS      MN= 0.318
1.27001
1.15001
0.89577
0.55048
0.0

LOADING-ANALYSIS STATIONS (WING)
0.0
0.17758
0.60454
0.74000
0.86827
0.97262
1.05040
1.11372
1.16937
1.20887
1.23799
1.26030
1.27901

NY SWEEP
-9.852
-10.266
-11.508
-13.164
-14.820
-16.476
-18.132
-19.788
-21.444
-23.100
-24.756
-26.412
-4.870

```

Figure B-2. Sample output from USPAN of wing spanwise loading parameter and local center of pressure at load evaluation stations.

USPAN - IP(52) **

STATION	QUTR LDNG	INBD LDNG	FLAP INCR	MN= 0.310	DF= 30.00
1.02000	0.0	0.0	0.0		
0.97737	0.06278	0.01386	0.04892		
0.90949	0.24822	0.05030	0.19791		
0.81897	0.48070	0.07124	0.40946		
0.72846	0.71170	0.09669	0.62071		
0.63795	0.88693	0.13111	0.75582		
0.54744	0.99040	0.17987	0.81952		
0.45693	1.08007	0.23407	0.84600		
0.36642	1.13946	0.29832	0.84114		
0.27590	1.18983	0.41185	0.77798		
0.18539	1.22320	0.61159	0.61161		
0.09488	1.23994	0.81019	0.42975		
0.0	1.23953	0.81140	0.42813		

STATION	OX SWEFF
1.02000	6.560
0.97737	6.836
0.90949	7.663
0.81897	8.766
0.72846	9.869
0.63795	10.971
0.54744	12.074
0.45693	13.177
0.36642	14.280
0.27590	15.383
0.18539	16.485
0.09488	17.588
0.0	19.171

Figure B-3. Sample output from USPAN of deflected flap spanwise loading parameter and local center of pressure at load evaluation stations.

```

** USPAN - IP(52) **

VALUES FROM BCLA/K TABLES FOR AR= 2.00000 SWB= 37.73 TR= 0.5000 RA/K= 2.0041
TR= 0.0 0.25 0.50 1.00
0.04268 0.04315 0.04270 0.03980 (HDR TAIL)

VALUES FROM LOADING TABLES FOR MN= 0.318
1.32855 1.26743 1.23203 1.16170 STA= 0.0
1.20663 1.18044 1.17246 1.16334 STA= 0.383
0.86047 0.90957 0.92676 0.96722 STA= 0.707
0.40948 0.49930 0.51510 0.54820 STA= 0.924

```

Figure B-4. Sample output from USPAN of horizontal tail lift curve slope parameter and spanwise loading parameter.

```

STATION      0.0      0.34300      0.70700      0.92400      1.00000
LOADING AT DATA STATIONS      MN= 0.318
1.23203
1.17246
0.92676
0.51510
0.0

LOADING--ANALYSIS STATIONS (FOR TAIL)
0.0
0.18215
0.60436
0.78914
0.93660
1.04091
1.11003
1.16416
1.20282
1.22204
1.23035
1.23203
1.23203

DX SWEEP
-18.648
-19.155
-20.557
-22.475
-24.294
-26.163
-28.032
-29.900
-31.769
-33.638
-35.507
-37.376
-37.376

STATION      0.0      0.34300      0.70700      0.92400      1.00000
LOADING AT DATA STATIONS      MN= 0.318
1.23203
1.17246
0.92676
0.51510
0.0

LOADING--ANALYSIS STATIONS (FOR TAIL)
0.0
0.18215
0.60436
0.78914
0.93660
1.04091
1.11003
1.16416
1.20282
1.22204
1.23035
1.23203
1.23203

DX SWEEP
-18.648
-19.155
-20.557
-22.475
-24.294
-26.163
-28.032
-29.900
-31.769
-33.638
-35.507
-37.376
-37.376

```

Figure B-5. Sample output from USPAN of horizontal tail spanwise loading parameter and local center of pressure at load evaluation stations.

```

** USPAN - IP(52) **

VALUES FROM RCLA/K TABLES FOR AR= 1.7761  SWB= 57.59  TR= 0.4056  BA/K= 1.8507

TR= 0.0  0.25  0.50  1.00
0.03591  0.03528  0.03400  0.03152  (VERT TAIL)

VALUES FROM LOADING TABLES FOR MN= 0.316

1.30547  1.22059  1.16836  1.06486  STA= 0.0
1.20517  1.17735  1.16165  1.15456  STA= 0.383
0.86897  0.93583  0.96552  1.02152  STA= 0.707
0.43151  0.52829  0.54977  0.58320  STA= 0.924

```

Figure B-6. Sample output from USPAN of vertical tail lift curve slope parameter and spanwise loading parameter.

USPAN - IP(52) **

MN= 0.31A

LOADING AT DATA STATIONS

STATION
0.0
0.3A300
0.70700
0.92400
1.03000

1.1A002
1.16640
0.95452
0.541A
0.0

LOADING-ANALYSIS STATIONS (VERT TAIL)

STATION
1.00000
1.07000
0.94336
0.88672
0.83000
0.77345
0.71681
0.66017
0.60353
0.54690
0.49026
0.43301
0.0

0.0
0.0
0.41943
0.66816
0.77313
0.85962
0.94074
1.00911
1.05612
1.09102
1.12119
1.14662
1.1A002

ON SWEEP

STATION
1.03000
1.09000
0.94336
0.88672
0.83000
0.77345
0.71681
0.66017
0.60353
0.54690
0.49026
0.43301
0.0

-26.81A
-26.81A
-20.043
-31.760
-33.495
-35.721
-37.947
-40.173
-42.300
-44.624
-46.850
-49.100
52.500

Figure B-7. Sample output from USPAN of vertical tail spanwise loading parameter and local center of pressure at load evaluation stations.

```

WING PARAMETERS
CLA= 4.73664    KW(R)A=0.87844    VM(R)A= 592.13    DXW(R)A= 243.58    DXB(W)A= 147.04
                                ** USPAN - IP(52) **

SIDE OF BODY UNITS
USZWR= 0.87844    UMXWR= 417.379    UMYWR= -60.949

SPANWISE UNIT DISTRIBUTIONS
STA      USZW(R)    UMXW(R)    UMYW(R)    MN= 0.318
SWEEP    SWEEP      SWEEP      SWEEP

1.00000    0.0      0.0      0.0
0.0774    0.00703    0.070    0.021
0.0005    0.07002    1.342    0.322
0.0100    0.00067    0.127    1.000
0.7705    0.16457    22.620    2.120
0.6300    0.24964    46.003    3.440
0.5474    0.34102    70.565    5.040
0.4560    0.43505    121.904    6.926
0.3664    0.54406    170.772    9.076
0.2750    0.65262    247.721    11.495
0.1854    0.76436    328.170    14.170
0.0949    0.87844    421.450    17.090
0.0      1.00000    513.267    19.970

```

Figure B-8. Sample output from USPAN of wing lift curve slope, center of pressure, and unit spanwise loads.

```

FLAPS DEFLECTED PARAMETERS                                ** USPAN - 1P(52) **
KRF= 0.59234      KW(R)F=0.93120      YW(R)F= 594.52      DXW(R)F= 276.91      DXB(W)F= 191.09
SIDE OF BODY UNITS
UISZFR= 0.93129      UMXFR= 444.712      UMYFR= -95.654
SPANWISE UNIT DISTRIBUTIONS
STA      USZF(R)      UMXF(R)      UMYF(R)      MN= 0.31R
SWEEP
1.0000   0.0      0.0      0.0      0.0
0.9774   0.00003   0.013   -0.006
0.9095   0.01508   0.695   -0.112
0.9190   0.06148   5.042   -0.503
0.7285   0.14019   16.494   -1.245
0.6380   0.24536   38.385   -2.346
0.5474   0.36572   73.083   -3.736
0.4569   0.40296   121.840   -5.344
0.3664   0.62186   185.142   -7.113
0.2759   0.74557   262.786   -8.945
0.1854   0.85173   353.493   -10.630
0.0949   0.93129   454.726   -11.977
7.0      1.00000   569.679   -13.926

```

Figure B-9. Sample output from USPAN of deflected flap lift curve clope, center of pressure, and unit spanwise loads.

USPAN - 1P(52) **

HOR TAIL PARAMETERS
 CLA= 2.32056 KM(R)=1.00000 YH= 100.84 DXH= 135.20
 SIDE OF BODY UNITS
 USZHR= 1.00000 IMXHR= 100.845 UMYHR= -29.067

SPANWISE UNIT DISTRIBUTIONS

STA SWEEP	USZH(R)	IMXH(R) SWEEP	UMYH(R) SWEEP	MN= 0.318
1.0000	0.0	0.0	0.0	
0.9750	0.00229	0.007	0.044	
0.9000	0.03199	0.310	0.645	
0.8000	0.10214	1.890	2.161	
0.7000	0.18902	5.319	4.197	
0.6000	0.28857	10.945	6.714	
0.5000	0.39685	19.018	9.651	
0.4000	0.51134	29.717	12.970	
0.3000	0.63050	43.167	16.646	
0.2000	0.75257	59.458	20.639	
0.1000	0.87604	78.642	24.908	
0.0	1.00000	100.741	29.426	
0.0	1.00000	100.741	29.426	

Fig B-10. Sample output from USPAN of horizontal tail lift curve slope, center of pressure, and unit spanwise loads.

USPAN - IP(52)

VERT TAIL PARAMETERS
 CYR= 1.89612 KV(R)=0.62286 NZV= 196.37 DXV= 501.46
 TOP OF BODY UNITS
 USYVR= 0.52296 UMXVR= 94.472 UMZVR= -89.641

SPANWISE INIT DISTRIBUTIONS

STA SWEEP	USYV(R) SWEEP	UMXV(R) SWEEP	UMZV(R) SWEEP	MY= 0.318
1.0000	0.0	0.0	0.0	
1.0000	0.0	0.0	0.0	
0.9434	0.01202	0.238	0.349	
0.8967	0.04218	1.321	1.297	
0.8301	0.08448	3.860	2.638	
0.7734	0.13127	8.134	4.259	
0.7168	0.18284	14.355	6.162	
0.6602	0.23873	22.706	8.347	
0.6035	0.29790	33.335	10.791	
0.5469	0.35946	46.356	13.471	
0.4903	0.42287	61.851	16.372	
0.4330	0.48856	80.110	19.524	
0.0	1.00000	305.512	18.168	

Figure B-11. Sample output from USPAN of vertical tail lift curve slope, center of pressure, and unit spanwise loads.

```

CONDITION NO= 60R25.      MN= 0.318      ALT= 0.      NF= 30.00      ** BNLD5 - IP(50) **

BODY LOADS
PZN= 28637.      PYN= 0.      XRN= 652.82

WING PANEL LOAD
PZW(R)/2= 801953.      YRW(B)= 592.98      XRW(B)= 2163.54

WING CARRY-OVER LOAD
PZA(W)= 185100.      XRA(W)= 2077.75

HORIZONTAL TAIL LOADS
PZH/?= -68R26.      YRH=100.84      XRH= 3225.20      DMXH= 2082216.

VERTICAL TAIL LOAD
PVV= 0.      ZRV= 0.0      XRV= 2866.75

AIRPLANE INERTIA FACTORS
NZ= 2.00      NY= 0.0      ODOT= 0.0      RDOT= 0.0

COMPONENT SPANWISE FACTORS
PZW(R)A= 103373A.      PZW(R)F= 570167.      PZW(B)=-137651.

```

Figure B-12. Sample output from BNLD5 of vehicle loads, center of pressure, and inertia factors.

WING LOADS COND NO= 6025. MN= 0.318 ALT= 0. DF= 30.00 ** SPARM - IP(51) **

SNR DIST= 117.00IN SW AT SOB= 801953.LB BM AT SOB=381715200.IN-LB TM AT SOB=-65142720.IN-LB

STATION (IN)	SHEAR (LB)	BEND MCM (IN-LB)	TORS MOM (IN-LB)
1254.67	0.	0.	0.
1226.24	1479.	20994.	10291.
1141.11	21571.	1002612.	154764.
1027.55	72172.	6325479.	487559.
913.98	139747.	19358560.	871371.
800.42	221405.	34865296.	1310488.
686.86	312606.	69187200.	1827445.
573.29	409707.	110201152.	2439199.
459.73	510483.	162450896.	3162884.
346.17	612225.	226199728.	4025576.
232.61	710470.	301304064.	5083688.
119.04	901953.	397181568.	6388989.

Figure B-13. Sample output from SPABM of wing airload shears, bending moments, and torques.

HORIZONTAL TAIL LOADS				COND NO= 60825.	MN= 0.318	ALT= 0.	** SPARM - IP(51) **	
SOR DIST=	0.0 IN	SM AT SOR=	-79149.LR	BM AT SOB=	-7981819.IN-LB	TM AT SOR=	2300645.IN-LR	
STATION (IN)		SHEAR (LR)		BEND MOM (IN-LR)		TORS MOM (IN-LB)		
235.59		0.		0.		0.		
229.70		-191.		-534.		-3476.		
212.03		-2532.		-24506.		-51030.		
188.47		-8084.		-149560.		-171048.		
164.91		-14961.		-421019.		-322228.		
141.35		-22840.		-866290.		-531398.		
117.79		-31410.		-1505326.		-763888.		
94.23		-40472.		-2352056.		-1026575.		
70.68		-49904.		-3416626.		-1317536.		
47.12		-59566.		-4706107.		-1633594.		
23.56		-69338.		-6274507.		-1971462.		
0.0		-79149.		-7973588.		-2329016.		

Figure B-14. Sample output from SPARM of horizontal tail airload shears, bending moments, and torques.

VERTICAL TAIL LOADS COND NO= 60825. MN= 0.318 ALT= 0. ** SPABM - IP(51) **
 SOR DIST= 132.17IN SH AT SOB= 0.18 BM AT SOB= 2082216.IN-LB TM AT SOB= 0.IN-LB

STATION (IN)	SHEAR (LB)	BEND MOM (IN-LB)	TORS MOM (IN-LB)
699.44	0.	0.	0.
699.44	0.	0.	0.
659.82	0.	1238860.	1673573.
620.21	0.	1238860.	1673573.
580.50	0.	1238860.	1673573.
540.09	0.	1238860.	1673573.
501.36	0.	1238860.	1673573.
461.75	0.	1238860.	1673573.
422.13	0.	1238860.	1673573.
382.52	0.	1238860.	1673573.
342.90	0.	1238860.	1673573.
302.86	0.	1238860.	1673573.

Figure B-15. Sample output from SPABM of vertical tail airload shears, bending moments, and torques.

*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD 17) *** ** WINGNET - IP(53) **

*** W I N G ***

STATION	LC	+V	RS	RNZ	RC	RS*(-V)	RSRNZ	RSRNZ*RC
1	3	2131.	1.000	1.000	1.000	2131.	1.000	1.000
2	3	32043.	1.000	1.000	1.000	32043.	1.000	1.000
3	3	101635.	1.000	1.000	1.000	101635.	1.000	1.000
4	3	190435.	1.000	1.000	1.000	190435.	1.000	1.000
5	3	263650.	1.000	1.000	1.000	263650.	1.000	1.000
6	3	350229.	1.000	1.000	1.000	350229.	1.000	1.000
7	3	438337.	1.000	1.000	1.000	438337.	1.000	1.000
8	3	526503.	1.000	1.000	1.000	526503.	1.000	1.000
9	3	614028.	1.000	1.000	1.000	614028.	1.000	1.000
10	5	834107.	0.768	1.000	1.000	834107.	0.768	0.768
11	5	957934.	0.768	1.000	1.000	957934.	0.768	0.768

STATION	LC	-V	RS	RNZ	RC	RS*(-V)	RSRNZ	RSRNZ*RC
1	7	-275.	0.768	1.000	1.000	-275.	0.768	0.768
2	7	-12456.	0.768	1.000	1.000	-12456.	0.768	0.768
3	7	-39316.	0.768	1.000	1.000	-39316.	0.768	0.768
4	7	-71552.	0.768	1.000	1.000	-71552.	0.768	0.768
5	7	-108303.	0.768	1.000	1.000	-108303.	0.768	0.768
6	7	-148739.	0.768	1.000	1.000	-148739.	0.768	0.768
7	7	-192000.	0.768	1.000	1.000	-192000.	0.768	0.768
8	7	-237581.	0.768	1.000	1.000	-237581.	0.768	0.768
9	7	-284980.	0.768	1.000	1.000	-284980.	0.768	0.768
10	7	-333643.	0.768	1.000	1.000	-333643.	0.768	0.768
11	7	-383174.	0.768	1.000	1.000	-383174.	0.768	0.768

Figure B-16. Sample output from WINGNET of wing design airload shear envelope and normalizing factors.

STATION	LC	+BM	RS	RNZ	RC	RS*(+BM)	RSRNZ	RSRNZ*RC
1	3	30250.	1.000	1.000	1.000	30250.	1.000	1.000
2	3	1485321.	1.000	1.000	1.001	1485321.	1.000	1.001
3	3	9074139.	1.000	1.000	1.000	9074139.	1.000	1.000
4	3	25087216.	1.000	1.000	1.000	25087216.	1.000	1.000
5	3	50297696.	1.000	1.000	1.000	50297696.	1.000	1.000
6	3	85147456.	1.000	1.000	1.000	85147456.	1.000	1.000
7	3	129913898.	1.000	1.000	1.000	129913898.	1.000	1.000
8	3	184687216.	1.000	1.000	1.000	184687216.	1.000	1.000
9	3	249434528.	1.000	1.000	1.000	249434528.	1.000	1.000
10	3	324058624.	1.000	1.000	1.000	324058624.	1.000	1.000
11	3	408404224.	1.000	1.000	0.992	408404224.	1.000	0.992

STATION	LC	-BM	RS	RNZ	RC	RS*(-BM)	RSRNZ	RSRNZ*RC
1	7	-12414.	0.768	1.000	1.000	-9532.	0.768	0.768
2	7	-580099.	0.768	1.000	1.000	-445429.	0.768	0.768
3	7	-3519777.	0.768	1.000	1.000	-2702661.	0.768	0.768
4	7	-9815010.	0.768	1.000	1.000	-7536458.	0.768	0.768
5	7	-20027408.	0.768	1.000	1.000	-15379052.	0.768	0.768
6	7	-34622624.	0.768	1.000	1.000	-26584992.	0.768	0.768
7	7	-53970304.	0.768	1.000	1.000	-41441104.	0.768	0.768
8	7	-78362560.	0.768	1.000	1.000	-60170704.	0.768	0.768
9	7	-108034288.	0.768	1.000	1.000	-82954160.	0.768	0.768
10	7	-143160544.	0.768	1.000	1.000	-109975872.	0.768	0.768
11	7	-183962400.	0.768	1.000	1.000	-141178800.	0.768	0.768

Figure B-17. Sample output from WINGNET of wing design airload bending moment envelope and normalizing factors.

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*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD 17) ***
** WHVNET - IP(53) **

*** HORIZONTAL TAIL ***

STATION LC      +V      RS      RNZ      RS*(+V)  RS*RNZ
1         3      303.    1.000   -2.500    303.    -2.500
2         3     4142.    1.000   -2.500    4142.    -2.500
3         3    12863.    1.000   -2.500    12863.    -2.500
4         3    23237.    1.000   -2.500    23237.    -2.500
5         3    34889.    1.000   -2.500    34889.    -2.500
6         3    47419.    1.000   -2.500    47419.    -2.500
7         3    60552.    1.000   -2.500    60552.    -2.500
8         3    74108.    1.000   -2.500    74108.    -2.500
9         3    87884.    1.000   -2.500    87884.    -2.500
10        3   101709.    1.000   -2.500   101709.    -2.500
11        3   115483.    1.000   -2.500   115483.    -2.500

STATION LC      -V      RS      RNZ      RS*(-V)  RS*RNZ
1         7     -67.    0.768   -0.400    -52.    -0.307
2         7    -930.    0.768   -0.400   -721.    -0.307
3         7   -3000.    0.768   -0.400  -2303.    -0.307
4         7   -5555.    0.768   -0.400  -4265.    -0.307
5         7   -8484.    0.768   -0.400  -6514.    -0.307
6         7  -11671.    0.768   -0.400  -8962.    -0.307
7         7  -15042.    0.768   -0.400 -11550.    -0.307
8         7  -18551.    0.768   -0.400 -14244.    -0.307
9         7  -22146.    0.768   -0.400 -17005.    -0.307
10        7  -25783.    0.768   -0.400 -19797.    -0.307
11        7  -29434.    0.768   -0.400 -22601.    -0.307

```

Figure B-18. Sample output from WHVNET of horizontal tail design airload shear envelope and normalizing factors.

STATION	LC	+RM	RS	RNZ	RS*(+RM)	RS*RNZ
1	3	893.	1.000	-2.500	893.	-2.500
2	3	40168.	1.000	-2.500	40168.	-2.500
3	3	240479.	1.000	-2.500	240479.	-2.500
4	3	665714.	1.000	-2.500	665714.	-2.500
5	3	1350397.	1.000	-2.500	1350397.	-2.500
6	3	2319931.	1.000	-2.500	2319931.	-2.500
7	3	3591762.	1.000	-2.500	3591762.	-2.500
8	3	5177965.	1.000	-2.500	5177965.	-2.500
9	3	7096125.	1.000	-2.500	7096125.	-2.500
10	3	9319415.	1.000	-2.500	9319415.	-2.500
11	3	11877795.	1.000	-2.500	11877795.	-2.500

STATION	LC	-RM	RS	RNZ	RS*(-RM)	RS*RNZ
1	7	-108.	0.768	-0.400	-152.	-0.307
2	7	-9087.	0.768	-0.400	-6978.	-0.307
3	7	-55484.	0.768	-0.400	-42603.	-0.307
4	7	-156252.	0.768	-0.400	-119978.	-0.307
5	7	-321615.	0.768	-0.400	-246953.	-0.307
6	7	-559075.	0.768	-0.400	-429247.	-0.307
7	7	-873685.	0.768	-0.400	-670859.	-0.307
8	7	-1269386.	0.768	-0.400	-974698.	-0.307
9	7	-1749773.	0.763	-0.400	-1347795.	-0.307
10	7	-2313342.	0.768	-0.400	-1776300.	-0.307
11	7	-2963753.	0.768	-0.400	-2275718.	-0.307

THE POSITIVE AND NEGATIVE LOADS ON THE HORIZONTAL TAIL HAVE BEEN REVERSED
BECAUSE THE NEGATIVE RM AT THE ROOT WAS GREATER THAN THE POSITIVE RM AT THE ROOT

Figure B-19. Sample output from MHWNET of horizontal tail design airload bending moment envelope and normalizing factors.

*** DESIGN LOADS (RECORD 32) AND RATIOS (RECORD 17) ***

*** VERTICAL TAIL ***

*** WHVNET - IP(53) ***

STATION	LC	+V	RS	RNZ	RS*(+V)	RS*RNZ
1	0	0.	1.000	1.000	0.	1.000
2	19	1880.	1.000	0.141	1880.	0.141
3	19	6663.	1.000	0.141	6663.	0.141
4	19	12830.	1.000	0.141	12830.	0.141
5	19	19659.	1.000	0.141	19659.	0.141
6	19	27074.	1.000	0.141	27074.	0.141
7	19	35015.	1.000	0.141	35015.	0.141
8	19	43361.	1.000	0.141	43361.	0.141
9	19	51991.	1.000	0.141	51991.	0.141
10	19	60831.	1.000	0.141	60831.	0.141
11	19	69934.	1.000	0.141	69934.	0.141

STATION	LC	-V	RS	RNZ	RS*(-V)	RS*RNZ
1	0	0.	1.000	1.000	0.	1.000
2	19	-1880.	1.000	0.141	-1880.	0.141
3	19	-6663.	1.000	0.141	-6663.	0.141
4	19	-12830.	1.000	0.141	-12830.	0.141
5	19	-19659.	1.000	0.141	-19659.	0.141
6	19	-27074.	1.000	0.141	-27074.	0.141
7	19	-35015.	1.000	0.141	-35015.	0.141
8	19	-43361.	1.000	0.141	-43361.	0.141
9	19	-51991.	1.000	0.141	-51991.	0.141
10	19	-60831.	1.000	0.141	-60831.	0.141
11	19	-69934.	1.000	0.141	-69934.	0.141

Figure B-20. Sample output from WHVNET of vertical tail design airload shear envelope and normalizing factors.

STATION	LC	+RM	RS	RN7	RS*(+RM)	RS*RN7
1	0	0.	1.000	1.000	0.	1.000
2	3	146969A.	1.311	0.0	1926856.	0.0
3	3	146969A.	1.311	0.0	1926856.	0.0
4	3	146969A.	1.311	0.0	1926856.	0.0
5	3	146969A.	1.311	0.0	1926856.	0.0
6	19	2706599.	1.000	0.141	2706599.	0.141
7	19	3936427.	1.000	0.141	3936427.	0.141
8	19	5488839.	1.000	0.141	5488839.	0.141
9	19	737751A.	1.000	0.141	737751A.	0.141
10	19	9612231.	1.000	0.141	9612231.	0.141
11	19	12230438.	1.000	0.141	12230438.	0.141
STATION	LC	-RM	RS	RN7	RS*(-RM)	RS*RN7
1	0	0.	1.000	1.000	0.	1.000
2	3	-146969A.	1.311	0.0	-1926856.	0.0
3	3	-146969A.	1.311	0.0	-1926856.	0.0
4	3	-146969A.	1.311	0.0	-1926856.	0.0
5	3	-146969A.	1.311	0.0	-1926856.	0.0
6	19	-2706599.	1.000	0.141	-2706599.	0.141
7	19	-3936427.	1.000	0.141	-3936427.	0.141
8	19	-5488839.	1.000	0.141	-5488839.	0.141
9	19	-737751A.	1.000	0.141	-737751A.	0.141
10	19	-9612231.	1.000	0.141	-9612231.	0.141
11	19	-12230438.	1.000	0.141	-12230438.	0.141

Figure B-21. Sample output from WAMNET of vertical tail design airload bending moment envelope and normalizing factors.

** BLCNTL - IP(54) **										
LOAD CONDITION	ALTITUDE FEET	PACH NUMBER	PRESSURE PSI	LOCAL TEMP DEC F	TOTAL TEMP DEC F	SUN FLUX BTU/HR/FT2	SKIN TEMP DEC F	STRESS WING	STRESS HORIZONTAL	STRESS VERTICAL
1 +NZ BALANCED	0	0.6100	14.70	518.7	556.0	356.6	555.0	66000.	66000.	66000.
2 +NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
3 +NZ BALANCED	22500	0.6700	6.06	438.4	504.6	425.1	505.5	66000.	66000.	66000.
4 +NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
5 +NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
6 -NZ BALANCED	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
7 -NZ BALANCED	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
8 FLAP DOWN MANEUVER	0	0.5734	14.70	518.7	530.2	356.6	534.7	66000.	66000.	66000.
9 FLAP DOWN 1 G TRIM	0	0.1809	14.70	518.7	522.2	356.6	531.4	66000.	66000.	66000.
10 + VERTICAL GUST	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
11 + VERTICAL GUST	20000	0.6100	6.77	447.3	406.6	425.0	507.6	66000.	66000.	66000.
12 + VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
13 + VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
14 - VERTICAL GUST	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
15 - VERTICAL GUST	20000	0.6100	6.75	447.3	506.6	425.0	507.6	66000.	66000.	66000.
16 - VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
17 - VERTICAL GUST	0	0.0	0.0	0.0	0.0	0.0	0.0	0.	0.	0.
18 LATERAL GUST	0	0.5740	14.70	518.7	552.6	356.6	552.4	66000.	66000.	66000.
19 LATERAL GUST	20000	0.6100	6.75	447.3	406.6	425.0	507.6	66000.	66000.	66000.
20 PITCH ACCELERATION	0	0.6000	14.70	518.7	556.0	356.6	555.0	66000.	66000.	66000.
21 PITCH ACCELERATION	22500	0.6700	6.06	438.4	504.6	425.1	505.5	66000.	66000.	66000.
22 YAW ACCELERATION	0	0.6100	14.70	518.7	556.0	356.6	555.0	66000.	66000.	66000.
23 YAW ACCELERATION	22500	0.6700	6.06	438.4	504.6	425.1	505.5	66000.	66000.	66000.

DESIGN TEMPERATURE CONDITION				
WING	HORIZONTAL	VERTICAL	3	23
	80.0	80.0		

*** MAXIMUM NET BENDING MOMENTS FOR FATIGUE ***				
FM AT SOP	CONDITION	N2	INERTIA FM/N2	NET FM AT SOP
140137616.	3	2.50	-28266704.	69470864.

FM AT STA 2	CONDITION	N2	INERTIA FM/N2	NET FM AT STA 2
124261152.	3	2.50	-22037552.	4916720.

DESIGN TEMPERATURE CONDITION

WING	CONDITION	
	3	15
HORIZONTAL	80.0	80.0
VERTICAL	80.0	23

*** MAXIMUM NET BENDING MOMENTS FOR FATIGUE ***

FM AT SOP	CONDITION	NZ	INERTIA BM/NZ	NET BM AT SOP
140127616.	3	2.50	-28266704.	69470864.
FM AT STA 2	CONDITION	NZ	INERTIA FM/NZ	NET FM AT STA 2
124261152.	3	2.50	-22037552.	49167210.

Figure B-22. Sample output from BLCNTL of ambient properties equilibrium skin temperature, structure compression yield strength, design conditions, and maximum net bending moments at wing fatigue evaluation stations.

SPECTRA SEGMENT NO 1.

SOF	RFNO	MJM	WNS	RENO	MON	EXCEEDANCES-GUST	EXCEEDANCES-MAN
208964320.	192467120.	175060920.	150472736.	142975600.	126478400.	100991216.	92484016.
76096816.	60489680.	73189072.	59891952.	39021296.	19960640.	0.	0.
16497183.	32004352.	16407183.	0.	0.	0.	0.	0.
32004352.	16407183.	0.	0.	0.	0.	0.	0.
16407183.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	0.
0.	0.	0.	0.	0.	0.	0.	

1G SOF M = 54009608. 1G WNS M = 66535504.

Figure B-23. Sample output from FATMG of flight segment wing bending moment spectra.

SPECTRA SEGMENT NC '0.

SOF BEND MOM	WJS BEND MOM	EXCEEDANCES-TAXI
-51436.16.	-39048752.	0.10000E+02
-50125256.	-33647000.	0.50000E+02
-48773544.	-37845720.	0.25000E+01
-46061072.	-35743164.	0.75000E+00
-43351000.	-33640056.	0.22500E+02
-40642128.	-31530112.	0.50000E+03
-37932640.	-29435552.	0.11250E+05
-35223168.	-27333024.	1.22500E+06
-32513596.	-25230430.	0.21500E+07
-29804224.	-23127952.	0.62500E+07
-24305264.	-19922064.	0.92500E+07
-21675792.	-16810320.	0.21500E+07
-18966320.	-14711725.	0.22500E+06
-16256651.	-12615245.	0.11250E+05
-13547375.	-10512704.	0.50000E+03
-10837900.	-8410162.	0.22500E+02
-8128425.	-6307622.	0.75000E+00
-5418950.	-4205081.	0.25000E+01
-4064212.	-3155310.	0.50000E+02
-2709475.	-2102541.	0.10000E+02

Figure B-24. Sample output from FATMG of taxi segment wing bending moment spectra.

** FATMG - IP(55) **

SPECTRA SEGMENT NO 11.

GROUND-AIR-GROUND CYCLES

SUF PERD MCM NOS BEND MCM
 80134250. 310470.
 -04502304. -50140500.

OCCURRENCES
 0.25400E+05
 0.25400E+05

Figure B-25. Sample output from FATMG of ground-air-ground segment wing bending moment spectra.